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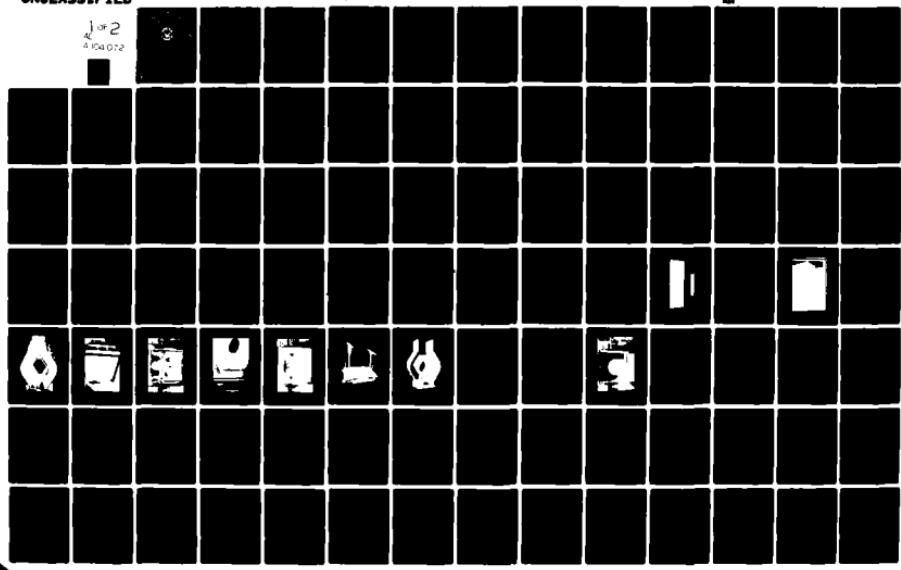
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THESIS

FLOW CHARACTERISTICS OF A MULTIPLE NOZZLE
EXHAUST GAS EDUCTOR SYSTEM

by

Dennis Leo Ryan III

March 1981

Thesis Advisor:

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Flow Characteristics of a Multiple Nozzle
Exhaust Gas Eductor System

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Submitted in partial fulfillment for the
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ABSTRACT

Cold flow tests of a four nozzle eductor system were conducted to evaluate the flow characteristics of several mixing stack configurations. A previously tested mixing stack design used a plug to shield the primary flow nozzles from view. Flow visualization was used to determine the flow pattern in the stack. The results of the visualization runs were then used to modify the mixing stack geometry. The eductor system flow characteristics were evaluated in terms of non-dimensional parameters governing the flow phenomena from a one dimensional analysis of a simple eductor system. The eductor system's pumping capacity was improved over that of the previously tested unmodified mixing stack.

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NOMENCLATURE

English Letter Symbols

A	- Area, in. ²
c	- Sonic velocity, ft/sec
C	- Coefficient of discharge
D	- Diameter, in.
F _a	- Thermal expansion factor
F _{fr}	- Wall skin-friction force, lbf
g _c	- Proportionality factor in Newton's Second Law, $g_c = 32.174 \text{ lbm-ft/lbf-sec}^2$
h	- Enthalpy, Btu/lbm
k	- Ratio of specific heats
L	- Length, in.
P	- Pressure, in. H ₂ O
P _a	- Atmospheric pressure, in. Hg
P _v	- Velocity head, in. H ₂ O
PMS	- Static pressure along length of mixing stack in. H ₂ O
R	- Gas constant for air, 53.34 ft-lbf/lbm-R
s	- Entropy, Btu/lbm-R
S	- Primary dimension of mixing stack
T	- Absolute temperature, R
u	- Internal energy, Btu/lbm
U	- Velocity, ft/sec

V	- Specific volume, ft ³ /lbm
W	- Mass flow rate, lbm/sec
D	- Distance from primary nozzle exit to mixing stack, in.
Y	- Expansion factor

Dimensionless Groupings

A*	- Secondary flow area to primary flow area ratio
AR	- Area ratio
f	- Friction factor
K	- Flow coefficient
K _e	- Kinetic energy correction factor
K _m	- Momentum correction factor at the mixing stack exit
K _p	- Momentum correction factor at the primary nozzle exit
M	- Mach number
ΔP* = p*	- Pressure coefficient
PMS*	- Mixing stack pressure coefficient
Re	- Reynolds number
D/S	- Standoff; Ratio of distance from primary nozzles to entrance of mixing stack, (D) to primary dimension of mixing stack (S)
T*	- Absolute temperature ratio of the secondary flow to primary flow
T _t * = TT*	- Absolute temperature ratio of the tertiary flow to primary flow
W* = W*	- Secondary mass flow rate to primary mass flow rate ratio

$W_t^* = WT^*$	- Tertiary mass flow rate to primary mass flow rate ratio
L/S	- Ratio of distance from entrance of mixing stack to diameter of mixing stack
α^*	- Induced flow density to primary flow density

Greek Letter Symbols

μ	- Absolute viscosity, lbf-sec/ft ²
ρ	- Density, lbm/ft ³

Subscripts

0	- Section within secondary air plenum
1	- Section at primary nozzle exit
2	- Section at mixing stack exit
f	- Film or wall cooling
m	- Mixed flow or mixing stack
or	- Orifice
p	- Primary
s	- Secondary
t	- Tertiary (Cooling)
u	- Uptake
w	- Mixing stack inside wall

Computer Tabulated Data and Illustrative Plots

UPTMACH	- Uptake Mach number
PA-PS	- Static pressure at mixing stack entrance, in. H ₂ O
PA-PT	- Static pressure in tertiary air plenum, in. H ₂ O

I. INTRODUCTION

With gas turbines becoming a more popular means of powering naval vessels, special considerations need to be given to their particular air breathing and exhausting characteristics. With air-fuel ratios of four to five times that of conventional steam plants and the requirement for a relatively large amount of combustion air, a large quantity of hot exhaust gas is generated. Due to gas turbine design, these exhaust gases are at temperatures significantly above those of conventionally powered ships. A few of the problems caused by these high temperatures are thermal damage to electronic equipment located on the masts of these ships, hot gas corrosion of masts and other superstructures located in the hot gas wake, and a significant infrared radiation signature created by the hot gas plume and hot external surfaces of the stack.

This thesis is an extension of research done by Ellin [Ref. 1], Moss [Ref. 2], Lemke and Staehli [Ref. 7], and Shaw [Ref. 8] to determine better geometric designs for the exhaust plenum and mixing stack system of gas turbine powered naval ships.

Ellin initiated the work by constructing an eductor model testing facility consisting of an uptake, primary flow nozzle, mixing stack, a means to control and measure the primary air flow, and a means to measure the secondary air flow; see

Figures 1 and 2. The primary air flow in the testing facility represents a gas turbine's hot exhaust gas. The secondary air flow is ambient air induced into the entrance of the mixing stack by the primary air flow; see Figure 3. From Ellin's study of multiple nozzle flow systems consisting of several identical round nozzles, it was determined that four primary flow nozzles were preferable to either three or five, and that nozzle length has little or no effect on the eductor system's overall performance. Ellin then verified the independence of the one-dimensional model correlation parameters used on flow rate or Mach number. He determined that for Mach numbers from 50% to 145% of the design Mach number of 0.064, the correlation parameters suggested in the one-dimensional analysis did in fact provide a good correlation of the data.

Moss' work followed, and it initially consisted of verifying the one-dimensional analysis as did Ellin. He then tested the effect of the stand-off distance (that distance between the exit plane of the primary flow nozzles and the entrance plane of the mixing stack). For the primary flow nozzles he tested, Moss determined that the optimum stand-off distance for maximum eductor pumping was a distance equal to 0.5 diameters ($0.5D_m$) of the mixing stack. An independent investigation of this, conducted by Harrel [Ref. 3], confirmed Moss' findings. Moss then investigated the effects of a conical transition placed on the entrance to the mixing stack.

He concluded that a straight mixing stack without an entrance transition provided a better system performance.

The study conducted by Lemke and Staehli [Ref. 7] investigated the effects on the eductor system's overall performance of varying the geometric configuration of the mixing stack and changing the area of the primary flow nozzles. Their work showed that a decrease in the ratio of the area of the mixing stack to primary flow nozzles from 3.0 to 2.5 decreased uptake back pressure but reduced the pumping coefficient of the eductor. Lemke and Staehli then investigated the effects of a solid diffusor, a two-ring and a three-ring diffuser. The results of these tests showed a decrease in uptake back pressure and an improvement in the eductor's pumping capacity. They then performed tests on a ported mixing stack. Their work determined that significant air flow through the ports could provide film cooling on the inside of the mixing stack. To enhance the film cooling provided by the ported mixing stack, Lemke and Staehli placed a shroud around the mixing stack. This shroud did not degrade the pumping or mixing characteristics, yet it provided thermal shielding of the mixing stack. Their final configuration was a combination of the ported mixing stack with flow through shroud and diffusor. Lemke and Staehli concluded that this geometric configuration, combined with a ratio of the area of the mixing stack to the primary flow nozzles 3.0, provided the best system overall performance.

The work done by Shaw [Ref. 8] concentrated on a method to further reduce the infrared signature of the mixing stack. Of particular concern was the concealment of the hot primary flow nozzles from an overhead view. Using the results of Lemke and Staehli, Shaw designed a rectangular mixing stack with an internal symmetric plug, Figure 4. The internal plug, which shields the primary nozzles from direct view, is cooled by two means. The plug was ported, to provide film cooling along the exhaust gas side of the device. Also, the plug was shrouded (on the non-exhaust gas side) to direct the flow of cooling air to the ports and provide a means of convectively cooling the downstream half of the plug, see Figure 5. Shaw found that the rectangular mixing stack with plug increased the system back pressure and reduced the pumping capacity of the eductor. He also determined that the design and placement of the plug improved mixing of the exhaust gases with the secondary cooling air and that the plug provided a potential means of presenting a cool surface to view.

Based upon the conclusions and recommendations of Shaw, it was decided that geometric changes in the eductor should be investigated in order to lessen the penalties involved with plug installation. These changes will be described in detail in Section IV.

Evaluation of eductor system performance was measured in four areas: the amount of secondary air flow induced by the primary air flow, the degree of mixing of primary and induced

air flows within the mixing stack system, the amount of uptake back pressure impressed upon the turbine exhaust by the eductor system, and the amount of wall cooling air available to reduce the exterior stack temperature of the eductor system. Additionally, a qualitative indication of the flow patterns in the mixing stack was obtained through flow visualization.

The key factor which allows cold flow testing to predict the effects of a hot gas eductor system is the similarity of the momentum and energy transfer mechanisms in turbulent flows.

II. EDUCTOR ANALYSIS

This investigation, being an extension of the work of Shaw [Ref. 8], and others [Ref. 1, 2, and 7]; uses the same one-dimensional analysis to model the eductor system. This one-dimensional analysis is given in Appendix A. As a result of this analysis, dimensionless parameters controlling the phenomena are obtained which are then used in the correlation of experimental data. Three such dimensionless groups used

are:

$$\Delta P^* = \frac{P_a - P_{os}}{\frac{\rho_s}{2g_c} U_p^2}$$

a pressure coefficient which compares the pumped head $P_a - P_{os}$ to the driving head U_p^2 of the primary flow

$$W^* = \frac{W_s}{W_p}$$

a flow rate ratio, secondary to primary mass flow rate

$$T^* = \frac{T_s}{T_p}$$

an absolute temperature ratio, secondary to primary

III. EXPERIMENTAL CORRELATION

For the geometries and flow rates investigated, it was confirmed by Ellin and Moss that a satisfactory correlation of the variables P^* , T^* , and W^* takes the form

$$\frac{\Delta P^*}{T^*} = \varnothing(W^*T^*^n) \quad (1)$$

where the exponent n is determined to be equal to 0.44. The details of the determination of 0.44 as the correlating exponent for the geometric parameters of the models tested is given in Reference [1]. To obtain an eductor model's pumping characteristic curve, the experimental data is correlated and analyzed using equation (1) that is P^*/T^* is plotted as a function of $W^*T^*^{0.44}$. This correlation is used to predict the open to the environment operating point. Variations in the eductor model's geometry will change the appearance of the pumping ability between models. For ease of discussion, $W^*T^*^{0.44}$ will henceforth be referred to as the pumping coefficient. Similarly, $WT^*TT^*^{0.44}$ will be referred to as the film cooling or tertiary pumping coefficient.

IV. MODEL GEOMETRIES

The multiple nozzle eductor systems studied here are designed specifically for service onboard gas turbine powered ships. The model consisted of a single primary uptake, a single cluster of four primary nozzles of constant cross section, and a mixing stack.

A. MIXING STACK WITH PORTED AND SHROUDED PLUG

The eductor mixing stack as designed by Shaw with its tertiary air flow cooling ports and associated plug shrouding provided the base design for the initiation of this study. Shaw's design utilized a symmetric plug to shield the hot primary exhaust nozzles from direct view. The initial design by Shaw is illustrated in Figure 4. He then provided two means of cooling the surfaces of the plug that were exposed to direct view. The first method provides film cooling along both sides of the downstream half of the plug. The film cooling air is induced through three sets of ports located as illustrated in Figure 5. The dimensional configuration of the ports is shown in Figure 6. The second means of cooling the plug provides an air flow to convectively cool the downstream half of the plug. This is accomplished with a shroud that directs a flow of air along the backside of the plug before it is pulled through the ports to be used

for film cooling. This flow of air is called tertiary air flow and is illustrated in Figure 5.

B. CUSP MODIFICATION OF THE PLUG

The symmetric design of the plug was modified by the addition of a cusp on the exhaust gas inlet side in order to provide a more uniform and gradual direction of the gas flow about the plug and to physically separate the flow into two symmetric parts. The cusp addition is pictured in Figures 7 and 9 and dimensionally illustrated in Figure 8. The cusp was manufactured from 0.102 cm (0.04 in) thick sheet aluminum. Each side of the cusp was formed to a concave shape with a 24.77 cm (9.75 in) radius of curvature. Axial supports were soldered on the inside of the cusp to prevent warpage. The cusp was then glued over the inlet end of the plug. The apex of the cusp was aligned on the cross-sectional centerline of the mixing stack to maintain symmetry of the flow about each half of the mixing stack. The apex of the cusp extended upstream toward the entrance of mixing stack 7.11 cm (2.8 in) along the axial centerline from the end of the plug. The lip formed where the edges of the cusp met the plug was filled in with putty to provide a smooth surface. The corners between the cusp and the sides of the mixing stack were sealed with silicone epoxy to minimize leakage.

C. MIXING STACK SIDE MODIFICATION

The second geometric modification to Shaw's design changed the shape of the curved sides of the mixing stack. In Shaw's original design, at the position where the two air flows around the plug reconverged, the sides of the mixing stack had a 137 degree angle bend into the final exit duct portion (see Figure 5). This point was 48.59 cm (19.13 in) from the entrance of the stack. This sharp corner was removed from the stack by installing a curved side beginning at a point 43.18 cm (17.0 in) from the stack entrance. The radius of curvature of the new piece was 43.18 cm (17 in). The new sides were also made of .102 cm (.04 in) sheet aluminum. Edges formed where the new section was fitted into the mixing stack were filled in with putty to maintain a smooth surface. The mixing stack with the side modification is dimensionally shown in Figure 10. The picture in Figure 11 highlights the area changed by the side modification and cusp addition.

V. EXPERIMENTAL FACILITY

Air is supplied to the primary nozzles by means of a centrifugal compressor and associated ducting schematically illustrated in Figure 1. The mixing stack configuration being tested is placed inside an air plenum containing an airtight partition so that two separate air flows, secondary and tertiary, may be measured. The air plenum facilitates the accurate measurement of secondary and tertiary air flows by using ASME long radius flow nozzles.

A. PRIMARY AIR SYSTEM

The circled numbers found in this section refer to locations on Figure 1. The primary air ducting is constructed of 16-gage steel with 0.635 cm (0.25 in) thick steel flanges. The ducting sections were assembled using 0.635 cm (0.25 in) bolts with air drying silicone rubber seals between the flanges of adjacent sections. Entrance to the inlet ducting ① is from the exterior of the building through a 91.44 cm (3.0 ft) square to a 30.48 cm (1.0 ft) square reducer, each side of which has the curvature of a quarter ellipse. A transition section ② then changes the 30.48 cm (1.0 ft) square section to a 35.31 cm (13.90 in) diameter circular section ③. This circular section runs approximately 9.14 m (30 ft) to the centrifugal compressor inlet.

A standard ASME square edged orifice ④ is located 15 diameters downstream of the entrance reducer and 11 diameters upstream of the centrifugal compressor inlet, thus insuring stability of flow at both the orifice and compressor inlet. Piezometer rings ⑤ are located one diameter upstream and one-half diameter downstream of the orifice. The duct section also contains a thermocouple just downstream of the orifice. Primary flow is measured by means of the standard ASME square edged orifice designed to the specifications given in the ASME power test code [Ref. 5]. The 17.55 cm (6.902 in) diameter orifice used was constructed out of 304 stainless steel 0.635 cm (0.25 in) thick. The inside diameter of the duct at the orifice is 35.31 cm (13.90 in) which yields a beta ($\beta = d/D$) of 0.497. The orifice diameter was chosen to give the best performance in regard to pressure drop and pressure loss across the orifice for the primary air flow rate used (1.71 Kg/sec (3.77 lbm/sec)).

The centrifugal compressor 7 used to provide primary air to the system is a Spencer Turbo Compressor, catalogue number 25100-H, rated at 6000 cfm at 2.5 psi back pressure. The compressor is driven by a three phase, 440 volt, 100 horsepower motor.

A manually operated sliding plate variable orifice ⑥ was designed to constrict the flow symmetrically and facilitate fine control of the primary air flow. During operation, the butterfly valve ⑧, located at the compressor's

discharge, provided adequate regulation of primary air flow, eliminating the necessity of using the sliding plate valve. The sliding plate valve was positioned in the wide-open position for all data runs.

On the compressor discharge side, immediately downstream of the butterfly valve, is a round to square transition 9 followed by a 90 degree elbow 10 and a straight section of duct. All ducting to this point is considered part of the fixed primary air supply system. A transition section is fitted to this last square section which reduces the duct cross section to a circular section 29.72 cm (11.17 in) in diameter. This circular ducting tapers down to a diameter of 26.30 cm (11.5 in) to provide the primary air inlet to the eductor system being tested. The transition is located far enough upstream of the model to insure that the flow reaching the model is fully developed.

B. SECONDARY AIR PLENUM

The secondary air plenum, pictured in Figure 1 and 2, is constructed of 1.905 cm (0.75 in) plywood and measures 1.22 m by 1.22 m by 1.88 m (4 ft by 4 ft by 6.17 ft). It serves as an enclosure that can contain all or only part of the eductor model and still allow the exit plane of the mixing stack to protrude. The purpose of the secondary air plenum is to serve as a boundary through which secondary air for the eductor system must flow. Long radius ASME flow nozzles,

designed in accordance with ASME power test codes [Ref. 5] and constructed of fiberglass, penetrate the secondary air plenum, thereby providing the sole means for metering the secondary air reaching the eductor. Appendix D of Reference [1] outlines the design and construction of the secondary air flow nozzles. By measuring the temperature of the air entering and the pressure differential across the ASME flow nozzles, the mass flow rate of secondary air can be determined. Flexibility is provided in measurement of the mass flow rate of secondary air by employing flow nozzles with three different throat diameters: 20.32 cm (8 in), 10.16 cm (4 in), and 5.08 cm (2 in). By using a combination of flow nozzles, a wide variety of secondary cross sectional areas can be obtained.

A secondary air flow straightener, shown in Figures 1 and 2, consisting of a double screen is installed 1.22 m (4 ft) from the open end of the secondary air plenum, between the ASME long radius nozzles and the primary air flow nozzles. The purpose of the straightener is to reduce any swirl effect that could result when only a small secondary air flow area exists.

C. TERTIARY AIR PLENUM

The tertiary air plenum, pictured in Figures 12 and 13, is constructed of 1.90 cm (0.75 in) plywood and measure 1.22 m by 1.22 m by 1.22 m (4ft by 4 ft by 4 ft). It serves as

an enclosure that completely surrounds the mixing stack and allows the exit and entrance regions to protrude. An airtight rubber diaphragm type seal, schematically illustrated in Figure 2 and pictured in Figures 14 and 15, is located at each end of the enclosure. This allows measurement of a tertiary air flow independent of the secondary air flow. Tertiary air flow is measured with the use of long radius ASME flow nozzles designed in accordance with ASME test codes [Ref. 5] and constructed of fiberglass. These nozzles are located so that they penetrate the airtight tertiary air plenum, thereby providing the sole means for metering the tertiary air reaching the eductor. By measuring the temperature of the air entering and the pressure differential across the ASME flow nozzles, the mass flow rate of tertiary air can easily be obtained. Flexibility in measuring the tertiary flow is provided by employing different size flow nozzles: two of 20.32 cm (8 in) throat diameter, three of 10.16 cm (4 in) throat diameter, and two of 5.08 cm (2 in) throat diameter. By using various combinations of these flow nozzles, a wide variety of tertiary cross section flow areas can be obtained.

The interior of the tertiary air plenum is pictured in Figure 13. The stand which holds the mixing stack can be seen mounted inside the plenum. Figure 15 shows the diaphragm air seal at the exit plane of the mixing stack. As can be seen, removable sections were located in the exit plane

door to allow for adjustments to the mixing stack and instrumentation without removing the diaphragms.

D. ALIGNMENT

The alignment of the mixing stack with the primary air flow nozzles was accomplished using a level, a 30.78 (12.0 in) rule graduated in 0.25 mm (0.01 in) and 45.72 x 30.48 cm (18.0 x 12.0 in) square. The graduated rule and square were to establish the stand-off distance (D/S) and to center the primary flow nozzles within the entrance area of the mixing stack. The geometric alignment was checked for accuracy using pressure readings at symmetric points on the model. Additional verification was obtained by subsequent symmetric exit velocity profiles measurements. The three axis mounting stand, pictured in Figure 16 allowed alignment adjustments to be performed easily.

E. INSTRUMENTATION

Pressure taps for measuring gage pressures are located inside the primary air uptakes just prior to the primary nozzles, inside the secondary air plenum, inside the tertiary air plenum, and at various points on the model. A variety of manometers, pictured in Figure 17, were used to indicate the pressure differentials. A schematic representation of the pressure measuring instrumentation is illustrated in Figures 18 and 19. Monitoring of each of the various pressures was facilitated by the use of a scanivalve and a

multiple valve manifold. The scanivalve was used to select the pressure tap to be read, while the multiple valve manifold allowed selection of the optimum manometer for the pressure being recorded. A vent was included in the multiple valve manifold which provided a means of venting the manometers between pressure readings. When taking readings of the pressure distribution in the mixing stack, it was necessary to manually change the tubing from one end of the manometer to the other in order to get the negative pressure readings. The valve manifold provided a selection of a 15.24 cm (6.0 in) inclined water manometer, a 5.08 cm (2.0 in) inclined water manometer, and a 1.27 cm (0.5 in) inclined oil manometer (specific gravity 0.827). In addition, the following dedicated manometers were used in the system: a 50.80 cm (20 in) single column water manometer connected to the primary air flow just prior to the primary nozzles, a 1.27 cm (50 in) U-tube water manometer with each leg connected to a piezometric ring on either side of the orifice plate in the air inlet duct, and a 2.54 cm (1.0 in) inclined water manometer connected to the upstream piezometric ring.

Primary air temperatures, measured at the orifice outlet and just prior to the primary nozzles, are measured with copper-constantan thermocouples. The thermocouples are in assemblies manufactured by Honeywell under the trade name Megapak. Polyvinyl covered 20 gage copper-constantan extension wire is used to connect the thermocouples to a

Newport Digital Pyrometer, model number 267, which provides a digital display of the measured temperature in degrees Fahrenheit. The Newport Digital Pyrometer failed in use and was replaced by an Omega Digital Thermometer, Model Number 2176A. Secondary/tertiary ambient air temperature is measured with a mercury-glass thermometer and recorded in degrees Fahrenheit.

Velocity profiles at the mixing stack exit plane are obtained by using a pitot tube, pictured in Figure 20. The tube is affixed to a mounting template which allows accurate determination of the major axis, minor axis, minor and diagonal positions and distances. Alignment pins allow changes in velocity traverse directions to be made fast and accurately. The pitot tube is used in conjunction with the 15.24 cm (6 in) inclined water manometer for obtaining the velocity pressure head.

An additional objective of this research was to assemble an automated data acquisition system. The data which was used in this thesis was taken "manually", i.e., pressure and temperature values visually acquired from manometer and thermometer. Each piece of data was then inputed into a computer for reduction. Although equipment limitation prevented full automation, "semi-automatic" data acquisition now exists and is described in Appendix B.

VI. EXPERIMENTAL METHOD

Evaluation of the eductor model requires the experimental determination of pressure differentials across the ASME long radius flow nozzles, temperature of primary and induced air flows, internal mixing stack pressures, and mixing stack exit velocities. These experimentally determined quantities are then reduced to obtain pumping coefficients, induced air flow rates, pressure distributions within the mixing stack, and mixing stack exit velocity profiles. In addition an indication of flow behavior in the stack was accomplished by flow visualization. The performance characteristics of the eductor model are then evaluated to determine the model's relative effectiveness.

The following discussion addresses the individual performance characteristics of the eductor model and how they were determined.

A. PUMPING COEFFICIENTS

The secondary pumping coefficient and the tertiary pumping coefficient provide the basis for analysis of the eductor model's pumping performance. Thus, changes in eductor model parameters which affect pumping can be noted by a change in pumping coefficient. The pumping coefficient(s) is desired at the operating point corresponding to that of the ship-board prototype. At this point there is no restriction of

the secondary (or tertiary) air flow. In the model, this is simulated by completely opening the air plenum(s) to the environment. However, at this condition, the secondary (or tertiary) air flow rate cannot be measured. Therefore, the eductor's characteristics are determined, plotted, and then extrapolated to find the operating point, see Figure 21.

The pumping characteristics of the eductor model are established by varying the associated induced air flow rate, either secondary or tertiary, from zero to its maximum measurable rate. This rate is determined by sequentially opening the ASME flow nozzles mounted in the appropriate plenum and recording the pressure drop across the nozzles. Values for nozzle cross sectional area, pressure drop, and induced air temperature are then used to calculate the dimensionless parameters P^*/T^* and $W^*T^{*0.44}$ or PT^*/TT^* and $WT^{*TT^{*0.44}}$ as described in Appendix A. The dimensionless parameters are then plotted as illustrated in Figure 21. Extrapolation of the pumping characteristics curve to intersect with the zero pressure/temperature coefficient abscissa locates the appropriate operating point coefficient of the model.

B. INDUCED AIR FLOWS

Two induced air flows are identified in this study: secondary and tertiary.

The secondary air flow is the amount of air induced by the primary nozzles which mixed with the primary air flow to reduce the exhaust gas temperature.

The tertiary air flow indicates the amount of air induced by the stack pressure distribution to provide cooling air through the ports in the stack plug. In addition to the method of metering tertiary air flow as described in Section III, the effect on the tertiary pumping coefficient by varying the number of rows of cooling ports open, see Figure 6, was investigated.

C. PRESSURE DISTRIBUTIONS

The mixing stack axial static pressure was obtained using a series of pressure taps fixed to the mixing stack. These taps were placed in two axial rows, the rows being along the centerline of the flow. Along each row the taps were axially spaced in increments of one quarter primary dimension. The exact location of the pressure taps is indicated on Figure 2 and 4. The non-dimensional stack pressure, PMS*, (see Appendix A) is plotted versus X/S to obtain a mixing stack pressure distribution, see Figures 26a through 261.

D. EXIT VELOCITY PROFILES

Velocity profiles at the mixing stack exit were calculated from the pressures measured using the pitot tube pictured in Figure 20. Since it was impractical to obtain a complete three-dimensional plot of velocities at the exit plane of

the mixing stack, advantage was taken of the symmetry of the velocity surface resulting from the arrangement of the primary nozzles. Only three traverses were made. The first traverse passes vertically along the centerline down the mixing stack exit, the second traverse passes horizontally across the mixing stack. This traverse, along the centerline, passes across the top of the plug. The third traverse passes diagonally through the intersection of the centerlines. Figure 22 illustrates the orientation and identification of the three velocity traverses, while Figures 27a through i, show plots of the velocity traverses.

E. FLOW VISUALIZATION

Based upon Shaw's results it was felt necessary to obtain a visual indication of the flow patterns in the mixing stack. Visualization of flow was accomplished by using string or yarn placed in the flow stream. Two procedures were used. The first utilized a string the length of the mixing stack. With this string the shape of the air stream flow throughout the entire mixing stack could be seen. The second procedure involved the insertion of a probe with a small string attached to the end of it into the mixing stack. With this procedure it was very easy to investigate specific location within the stack for flow action, particularly for transition regions from positive to negative pressure. In order to record the flow visualizations it was necessary to

use a 16 mm motion picture camera set for a speed of 60 frames per second.

VII. DISCUSSION OF EXPERIMENTAL RESULTS

Exhaust eductor systems designed for marine gas turbine applications must substantially cool exhaust gases, present an exterior stack surface temperature which will not give an easily detectable infrared signature, and effectively disburse exhaust gases. In order to quantitatively evaluate the overall eductor model performance, four areas of performance were identified: the amount of secondary air flow induced by the primary air flow, referred to here as pumping; the degree of mixing of primary and induced air flow within the mixing stack system, referred to here as mixing; the amount of uptake back pressure impressed upon the turbine exhaust by the eductor system; and the amount of cooling air available to reduce the exterior stack temperature of the eductor system. A qualitative means of assessing flow patterns in the mixing stack was accomplished by flow visualization with string placed in the gas flow regions.

The eductor model and its modified configurations in this study were designed to shield the primary flow nozzles from an overhead view and to provide cooling to those surfaces which are visible from above.

The initial tests performed were for flow visualization. The tests showed the following condition within the mixing stack. First there appeared to be large scale turbulence

within the stack. This was demonstrated by the wide range of travel of a string running the length of the stack. Secondly, there was evidence of some sort of reverse flow, probably a vortex formation on the downstream sides of the plug. Finally, there was a demonstration of flow oscillatory interaction at the outlet end of the plug. The range of motions of the strings in the flow is shown in Figure 23.

Further model testing was conducted using an eductor model with the addition of a cusp on the inlet end of the symmetric plug. The tests were conducted with a primary flow nozzle area to mixing stack area ratio of 3.0, a mixing stack length L/S of 3.0 and a standoff distance D/S of 0.5. These ratios of mixing stack length, standoff distance and primary nozzle to mixing stack area were used as a result of the previous study by Shaw [Ref. 8]. The results of this first set of eductor system tests are shown in Tables I through XIII.

These tests showed the following results. First, the uptake back pressure, over the variation in secondary and tertiary flows tested, showed a reduction of up to .5 inches of H₂O. However, at the operating point of the eductor system (secondary and tertiary areas wide open) the back pressure was the same as Shaw's 9.85 inches of H₂O. This uptake back pressure may be too high and would impact on the turbine efficiency. Secondly, the eductor model pumping coefficient was improved by approximately sixteen percent

over that of Shaw. Figure 24a demonstrates that with Shaw's design the pumping coefficient at the operating point would be less than 0.3. With the addition of the cusp alone, the pumping coefficient is increased above 0.35. Finally, the velocity profiles in all three direction; vertical, horizontal, and diagonal showed little change from Shaw's data. On the average these profiles were relatively flat. These velocity traverses are depicted in Figures 17a through 27i.

The next mixing stack modification was the replacement of a sharp corner in the sides with a curved section that smoothly transitioned from the ducting over the plug to the exit ducting. The results of the tests performed after this modification are tabulated in Tables XIII through XVIII.

Initial comparisons of data were with the back pressures obtained. Over the ranges of secondary and tertiary flow combinations the back pressures increased over those values recorded for the cusp only addition. However, at the wide open operating point the back pressure was the same as that with the cusp only and with Shaw's data, 9.85 inches of H₂O. With the addition of the side modification the pumping coefficient, over the range of flows used, was reduced from those values obtained without the side modification (see Figure 24c). The pumping coefficients at the operating point without and with the side modification were essentially the same. Finally, the velocity profiles in all three directions showed more variation with the side modification

than previously. However, in the aggregate these profiles were still relatively flat. These traverses are in Figures 27g, h and i.

The following observations were made in comparing the cooling (tertiary) pumping coefficients (WT^*TT^* ^{0.44}). Concomitant with an increase or decrease in secondary pumping coefficient was an inverse action with the tertiary pumping coefficient. The addition of the cusp decreased the tertiary pumping coefficient from .060, the value obtained by Shaw, by approximately 33% to .040 (see Figure 25c). Over the ranges of flow combinations tested the addition of the side modification improved the tertiary pumping coefficient. At the operating point the values of the tertiary pumping coefficients for with and without the side modification essentially converged. These results are displayed in Figures 25a through 25g.

It was further noted that decreasing the number of tertiary ports open had the effect of increasing the secondary pumping coefficient and consequently decreasing the tertiary pumping coefficient.

The final indicator looked at was the pressure distribution in the mixing stack. The data obtained was essentially unchanged from that of Shaw's. Positive pressures occurred along the exterior curved surface of the mixing stack. As with Shaw, a large negative pressure (PMS^*) = - 0.30, occurs at the junction of the curved surface and

the rectangular mixing stack exit plane. It was noted that with the addition of the cusp only, this large negative pressure (PMS*) increased to a value of -0.35 (Figure 26a). With the further modification of the curved side the large negative pressure was significantly reduced to -.15, a value much less than Shaw's (Figure 26i). No significant pressures were noted along the flat surface of the mixing stack.

VI. CONCLUSIONS

This investigation studied the effects on the eductor system's overall performance of a mixing stack geometry which has a rectangular cross-section and employs a plug placed over the primary flow nozzles. The plug was modified with the addition of a cusp on the inlet end. The sides of the mixing stack was further modified to remove a sharp corner.

The conclusions resulting from this investigation are given here.

1. The addition of the cusp on the plug increased the secondary pumping coefficient and increased a negative pressure at a point within the mixing stack.
2. The addition of the cusp reduced the tertiary cooling pumping coefficient.
3. The modification of the sides of the mixing stack reduced the negative pressure at the specific point within the mixing stack.
4. The modification to the curved side of the mixing stack did not improve the performance of the eductor system and should be removed.

VII. RECOMMENDATIONS FOR FURTHER STUDY

This study showed the effects on a specific eductor system with several geometric configurations. This research adds some information to the overall data base upon which further research can be conducted. Recommendations for further investigation of this eductor system are presented here.

1. Further explore the flow patterns within the stack using tufting or some other flow visualization process.
2. Test the eductor model with the following modifications: remove the side modification, add an exterior shroud to the mixing stack and add a diffusor configuration to the exit plane of the mixing stack.

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2. Moss, C.M., Effects of Several Geometric Parameters on the Performance of A Multiple Nozzle Eductor System, Master's Thesis, Naval Postgraduate School, September 1977.
3. Harrell, J.P. Jr., Experimentally Determined Effects of Eductor Geometry on the Performance of Exhaust Gas Eductors for Gas Turbine Powered Ships, Engineer's Thesis, Naval Postgraduate School, September 1977.
4. Pucci, P.F., Simple Eductor Design Parameters, Ph.D. Thesis, Stanford University, September 1954.
5. American Society of Mechanical Engineers Interim Supplement 19.5 on Instrumentation and Apparatus, Fluid Meters, Sixth edition, 1971.
6. Kline, S.J. and McClintock, F.A., "Describing Uncertainties in Single-Sample Experiments," Mechanical Engineering, p. 3-8, January 1953.
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8. Shaw, R.S., Performance of a Multiple Nozzle Exhaust Gas Eductor System for Gas Turbine Powered Ships, Master's Thesis, Naval Postgraduate School, December 1980.

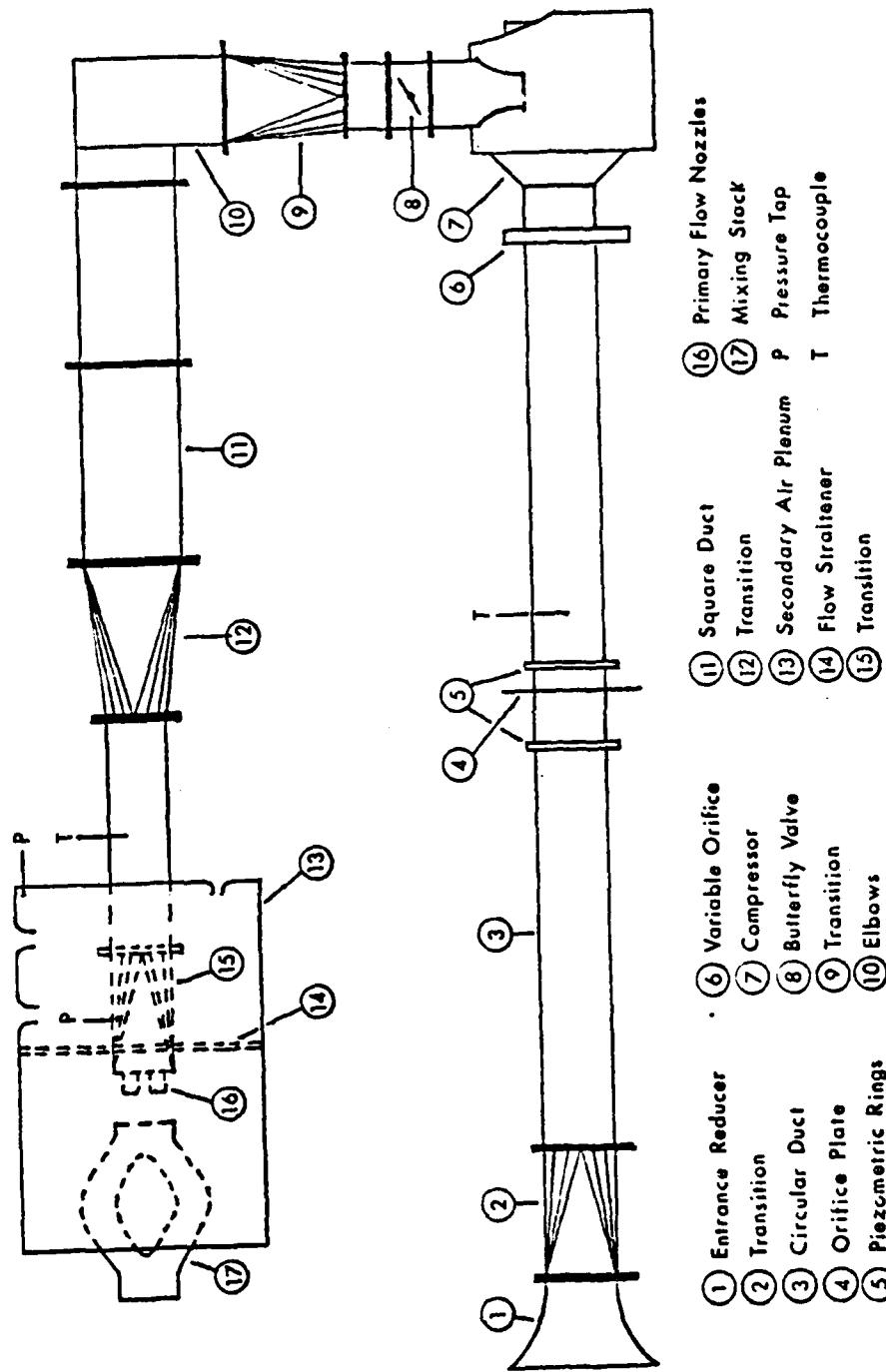


FIGURE 1. EDUCATOR MODEL TESTING FACILITY

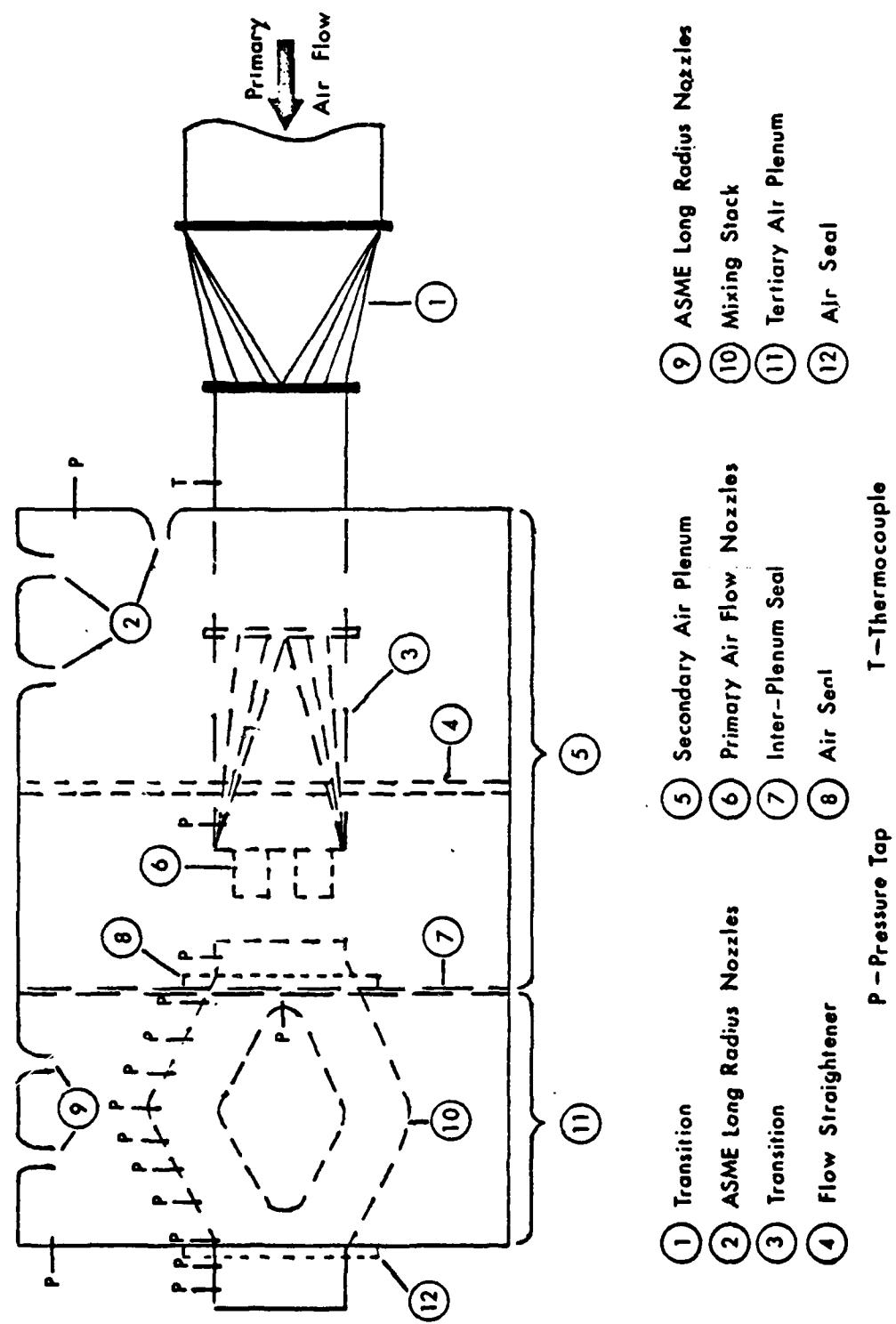


FIGURE 2. EDUCTOR MODEL WITH SECONDARY AND TERTIARY PLENUMS

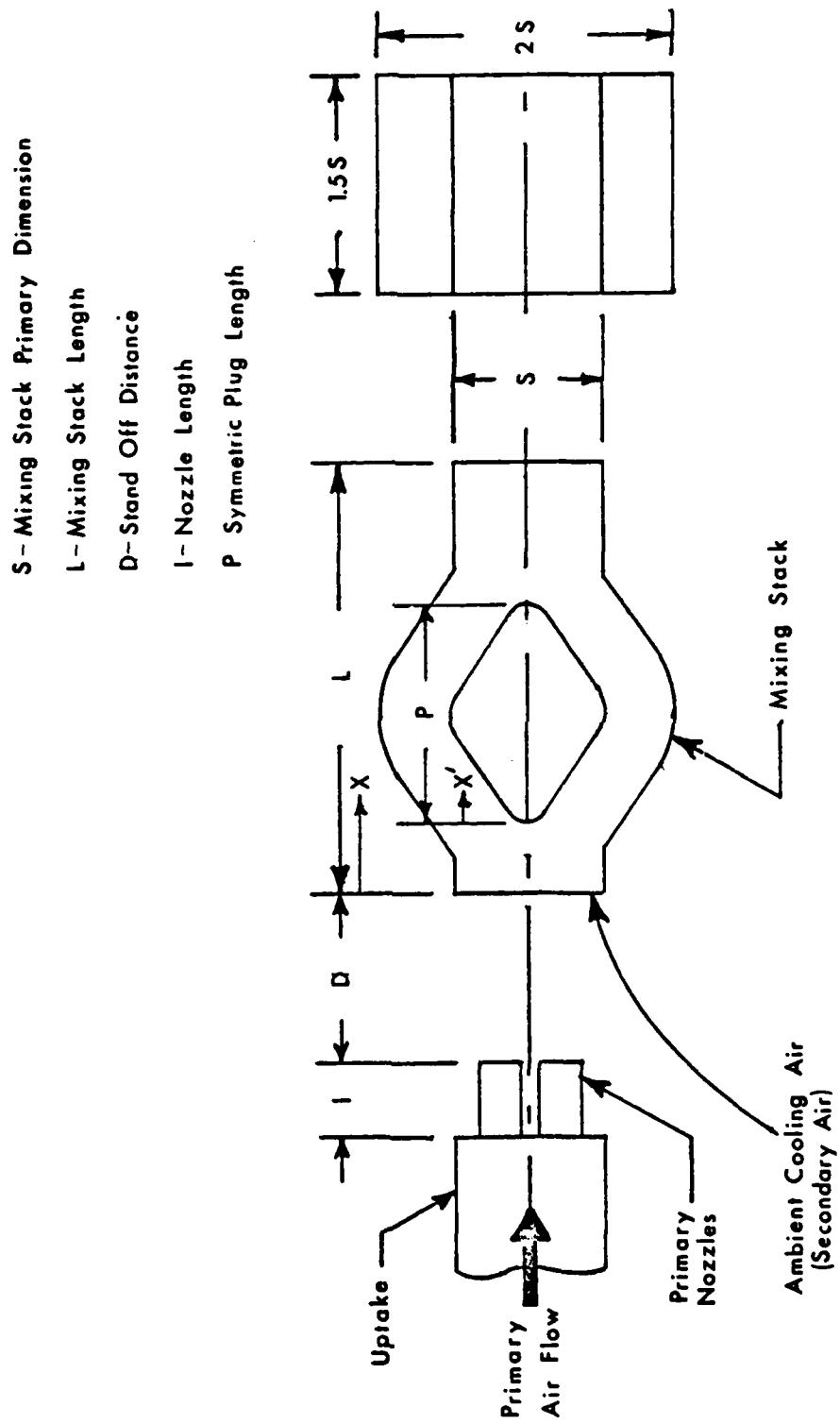


FIGURE 3. SCHEMATIC OF MIXING STACK EDUCTOR

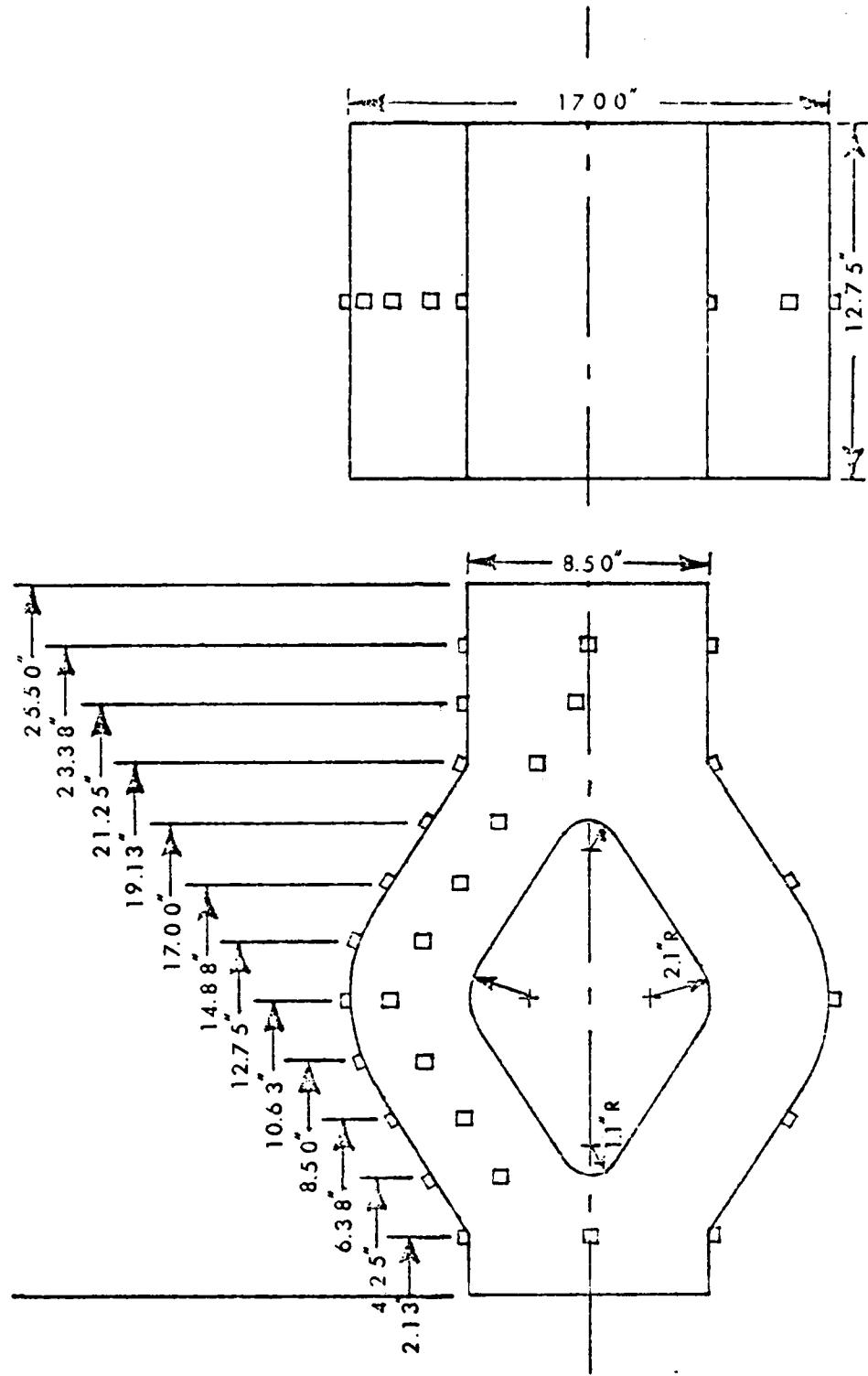


FIGURE 4. MIXING STACK WITH SYMMETRIC PLUG

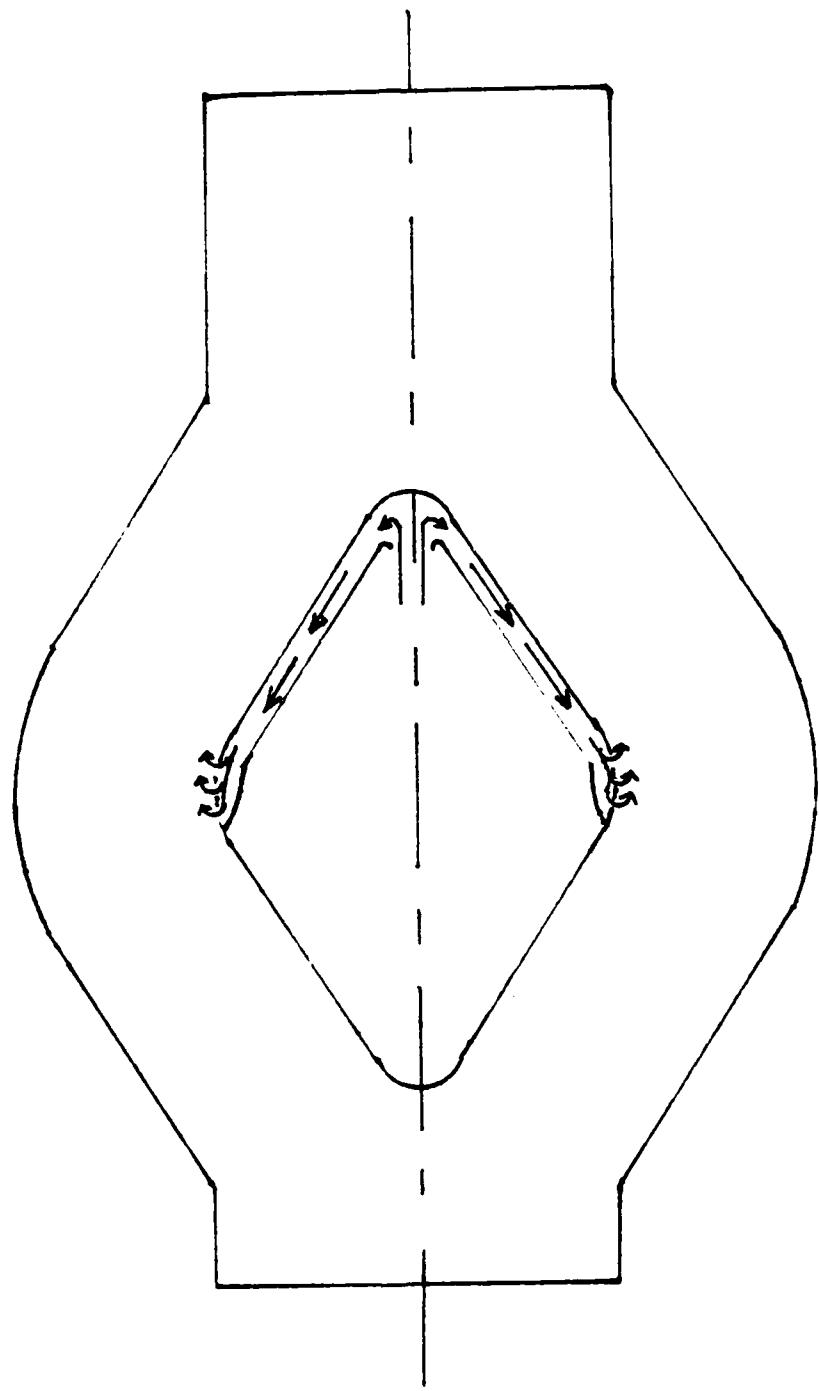


FIGURE 5. MIXING STACK WITH PORTED AND SHROUDED PLUG

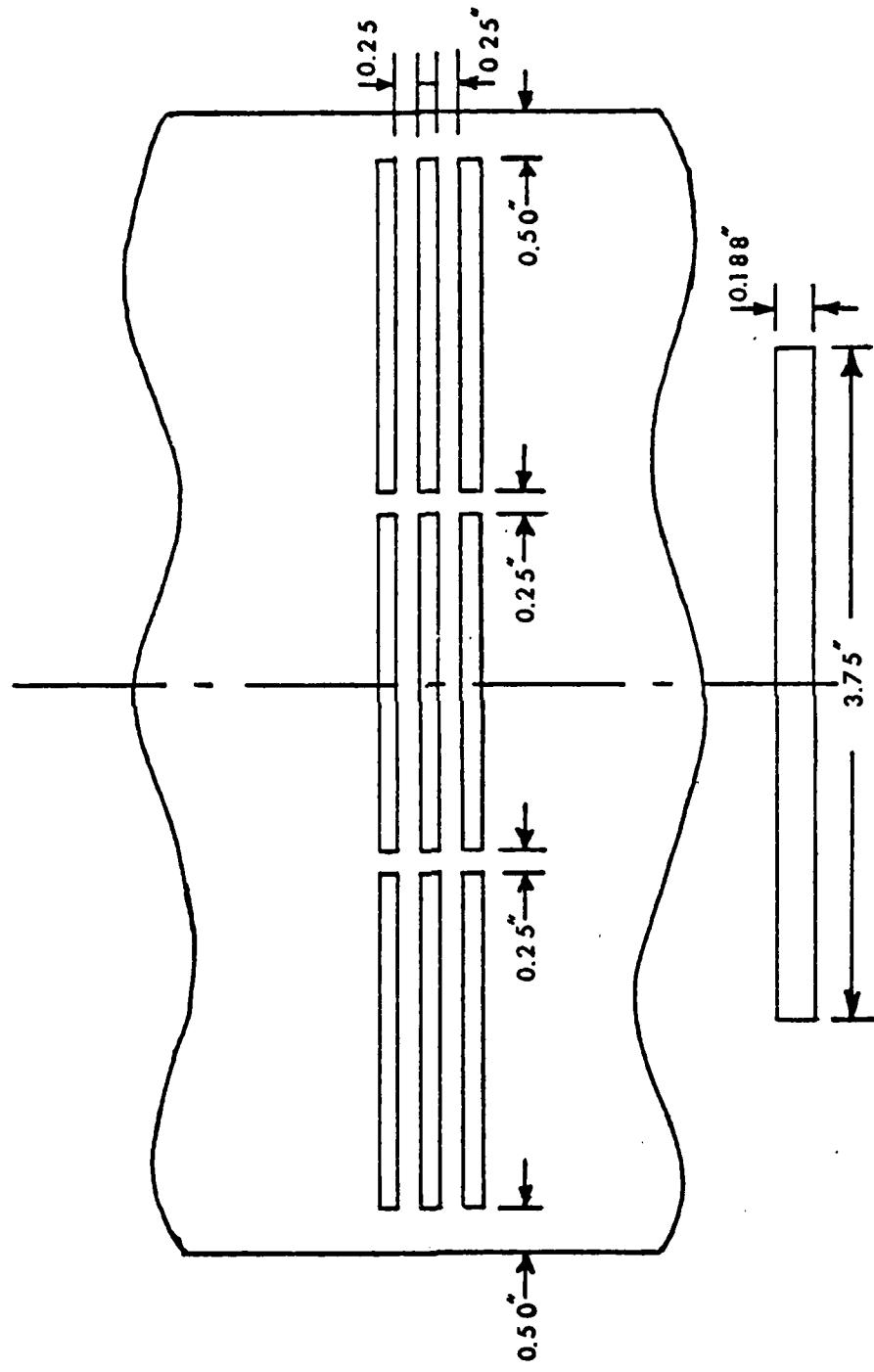


FIGURE 6. SCHEMATIC OF PLUG PORTING ARRANGEMENT

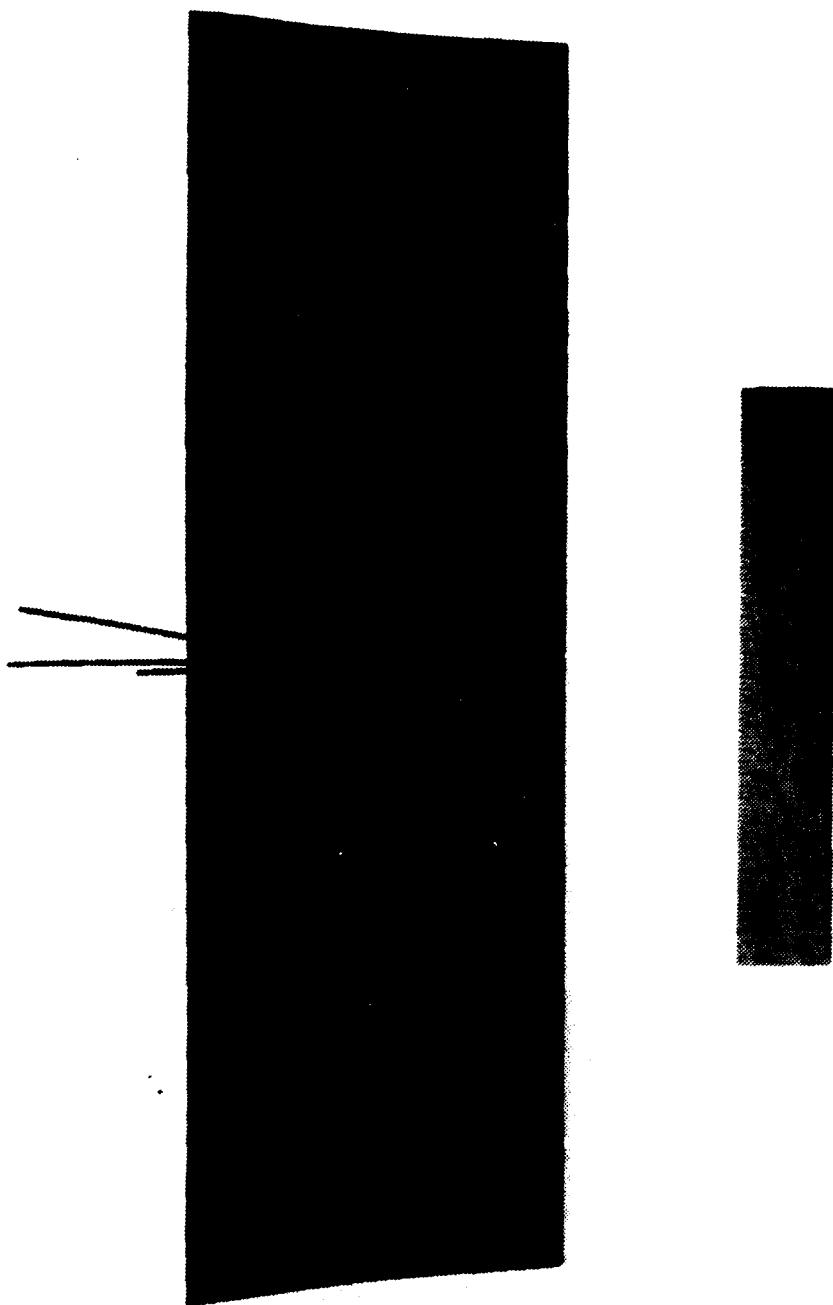


FIGURE 7. CUSP ILLUSTRATION

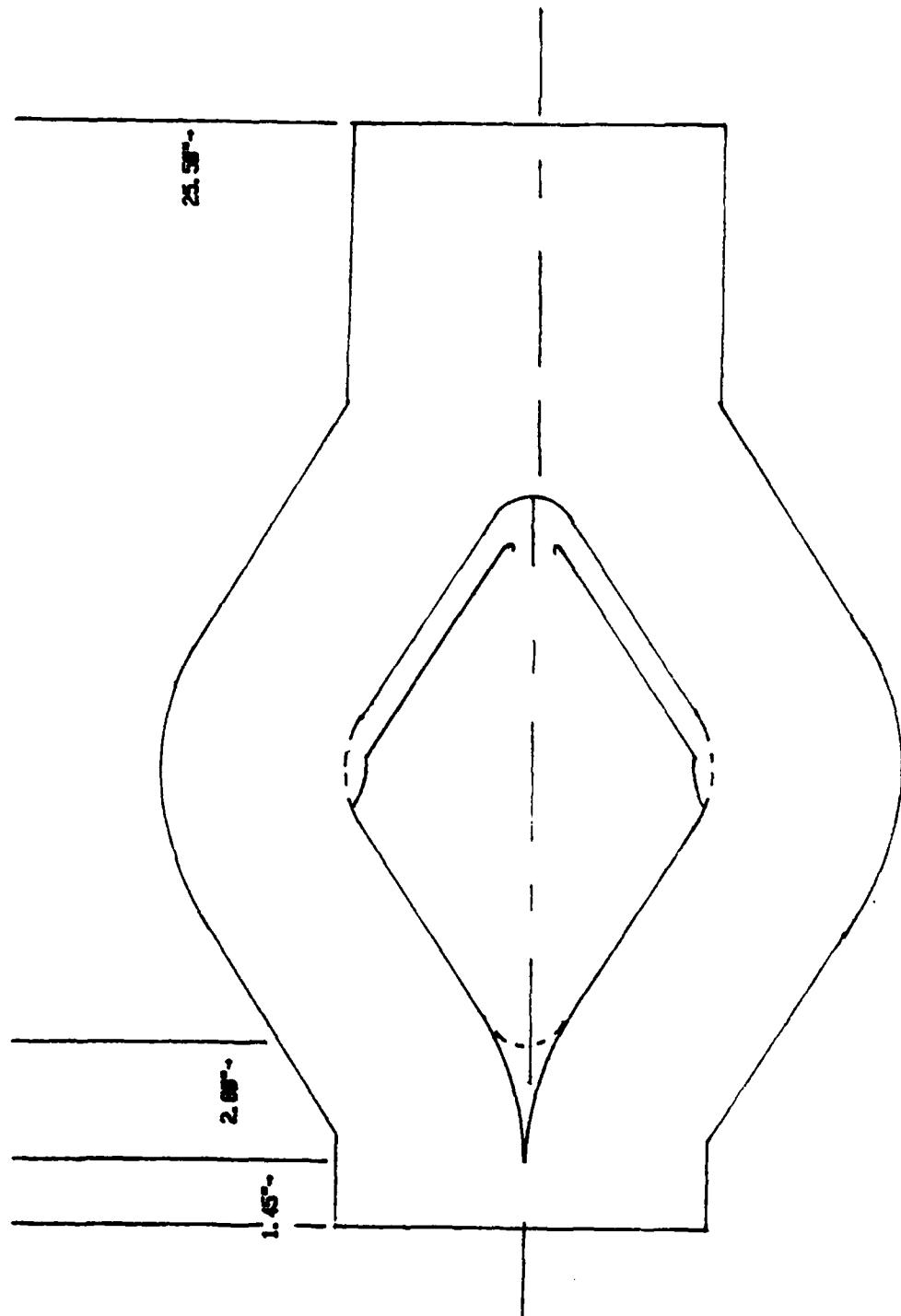


FIGURE 8. DIMENSIONAL ILLUSTRATION OF MIXING STACK WITH CUSP

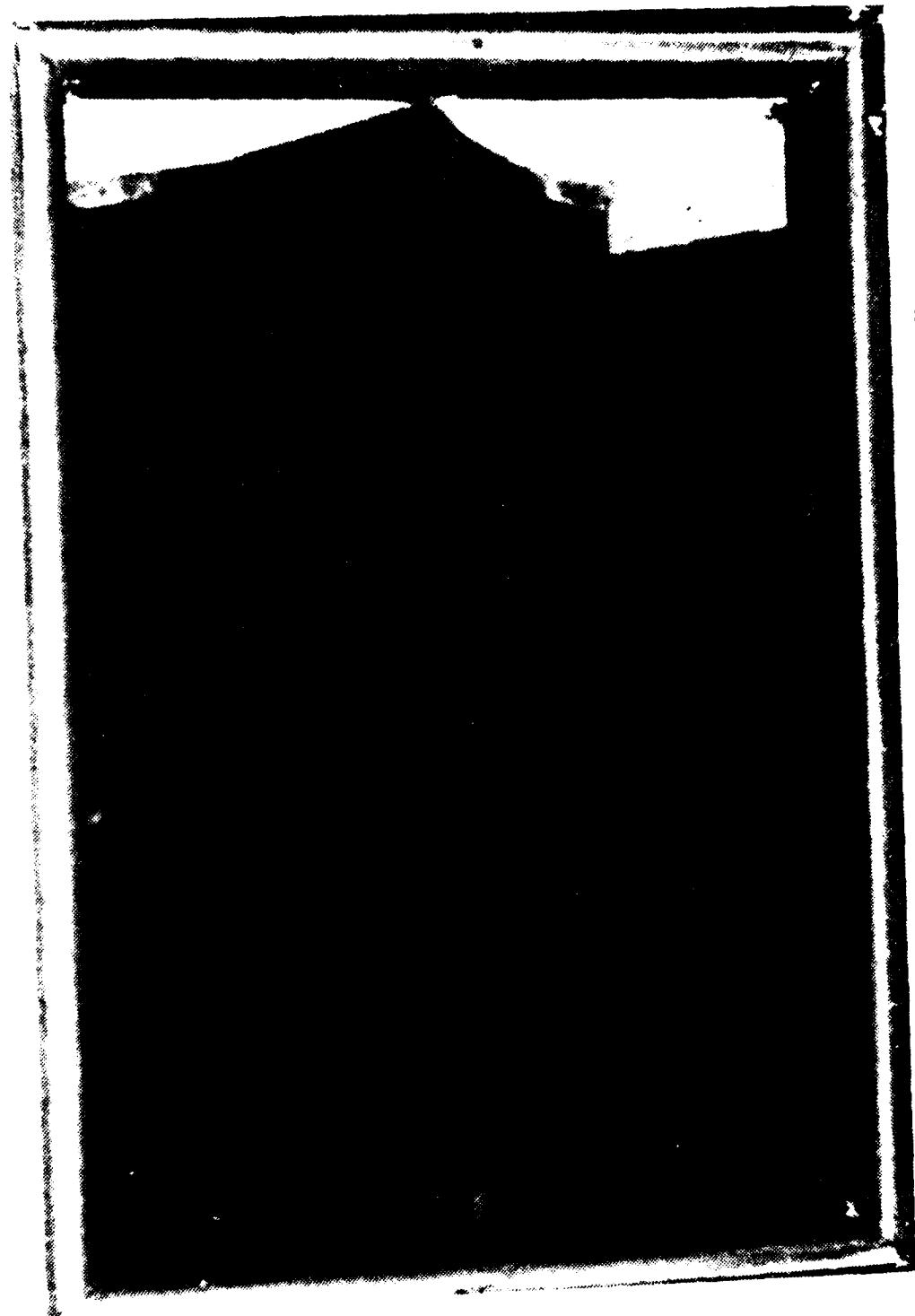


FIGURE 9. MIXING STACK WITH CUSP INSTALLED

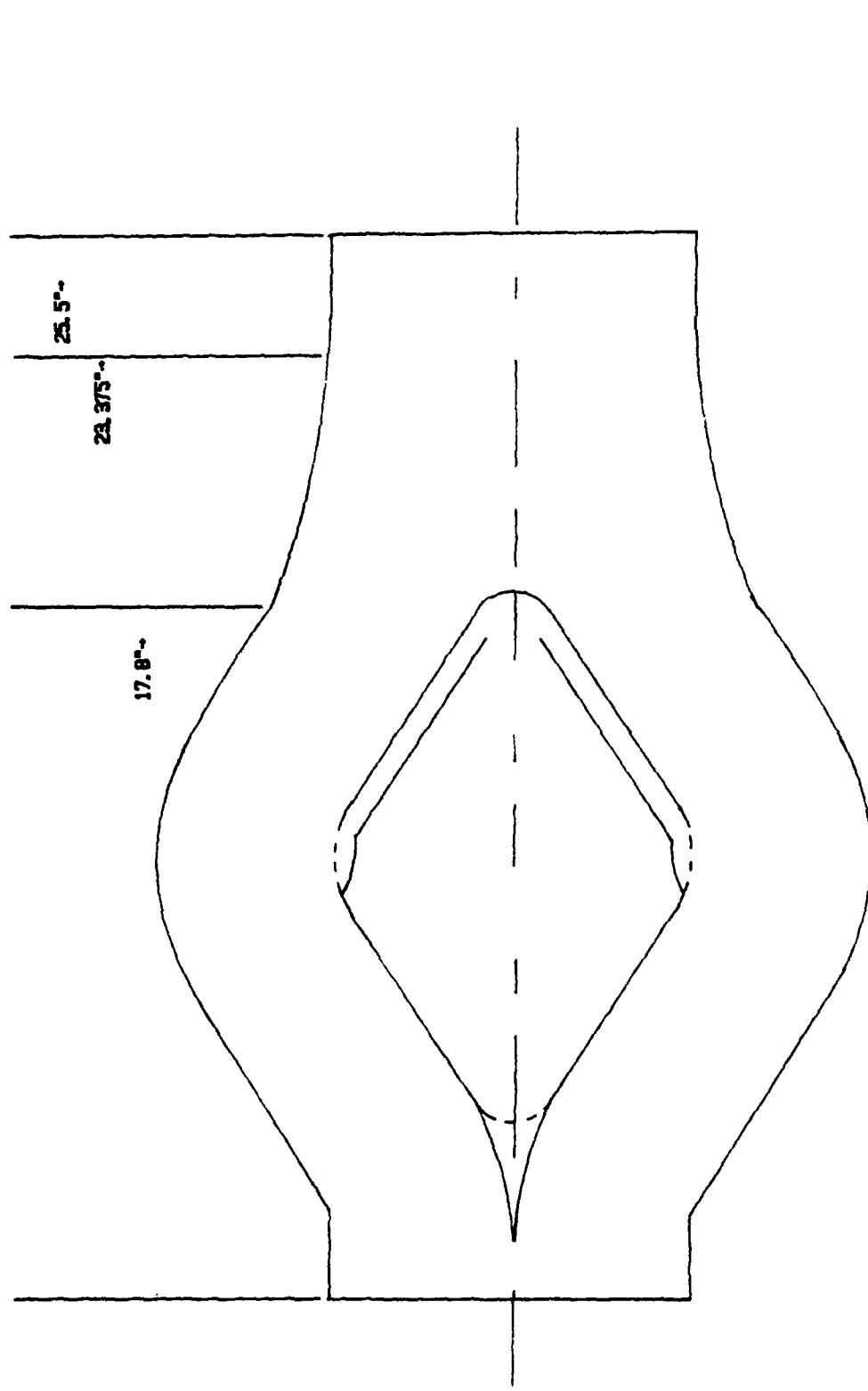


FIGURE 10. DIMENSIONAL ILLUSTRATION OF MIXING STACK WITH CUSP AND SIDE MODIFICATION

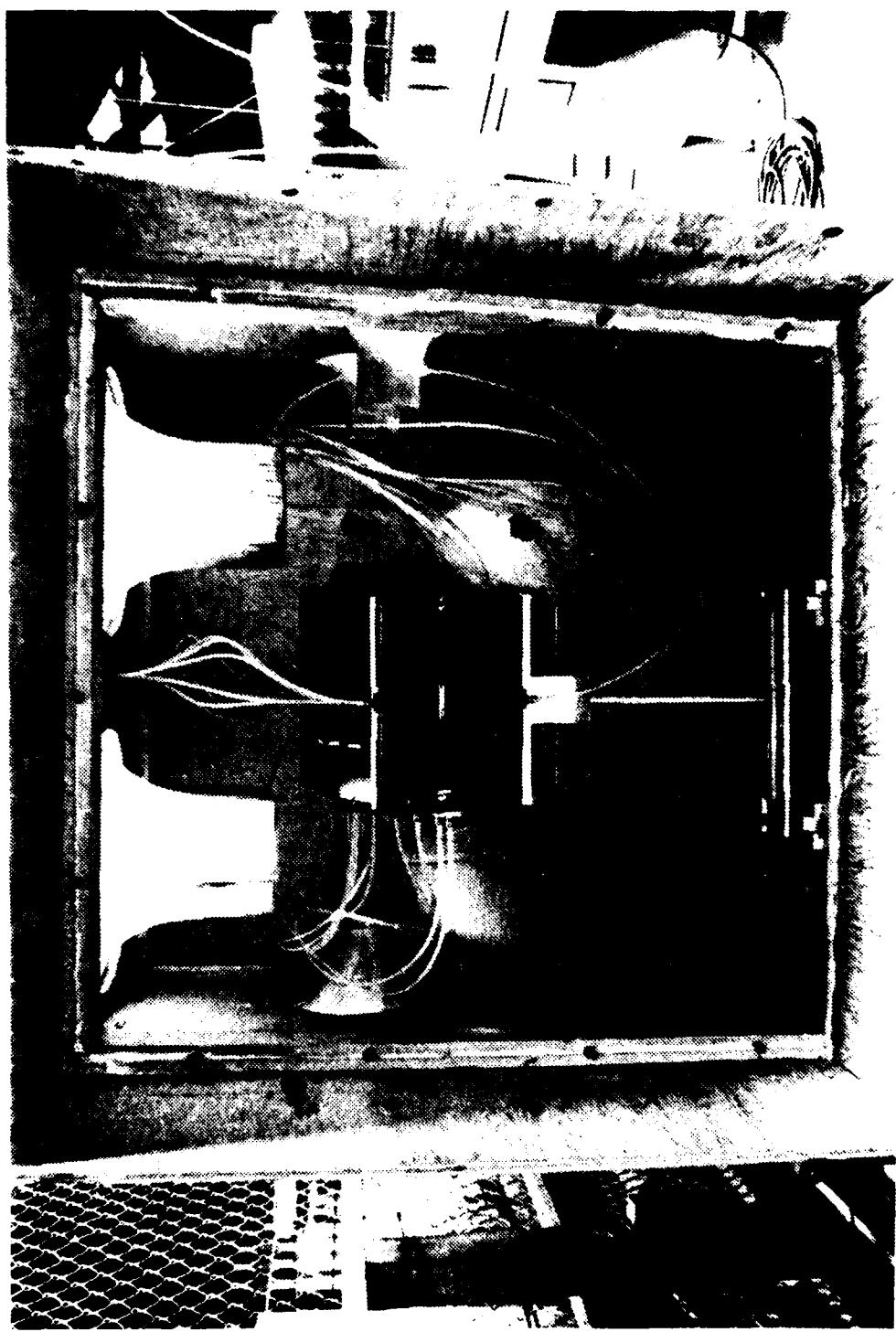


FIGURE 11. MIXING STACK WITH CUSP AND SIDE MODIFICATION



FIGURE 12. SECONDARY AND TERTIARY AIR PLENUMS

FIGURE 13. TERTIARY AIR PLENUM INTERIOR



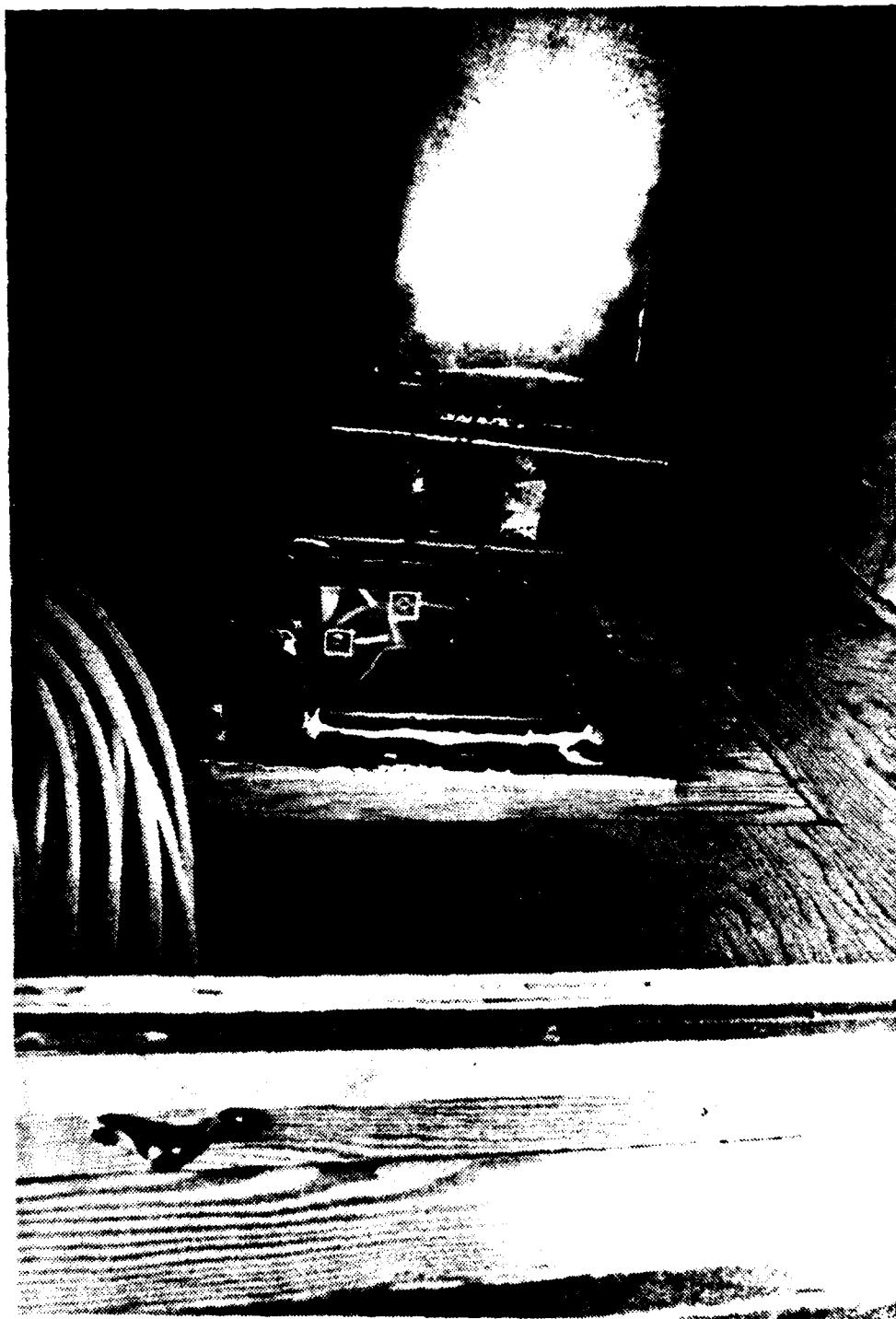


FIGURE 14. MIXING STACK ENTRANCE WITH AIR SEAL

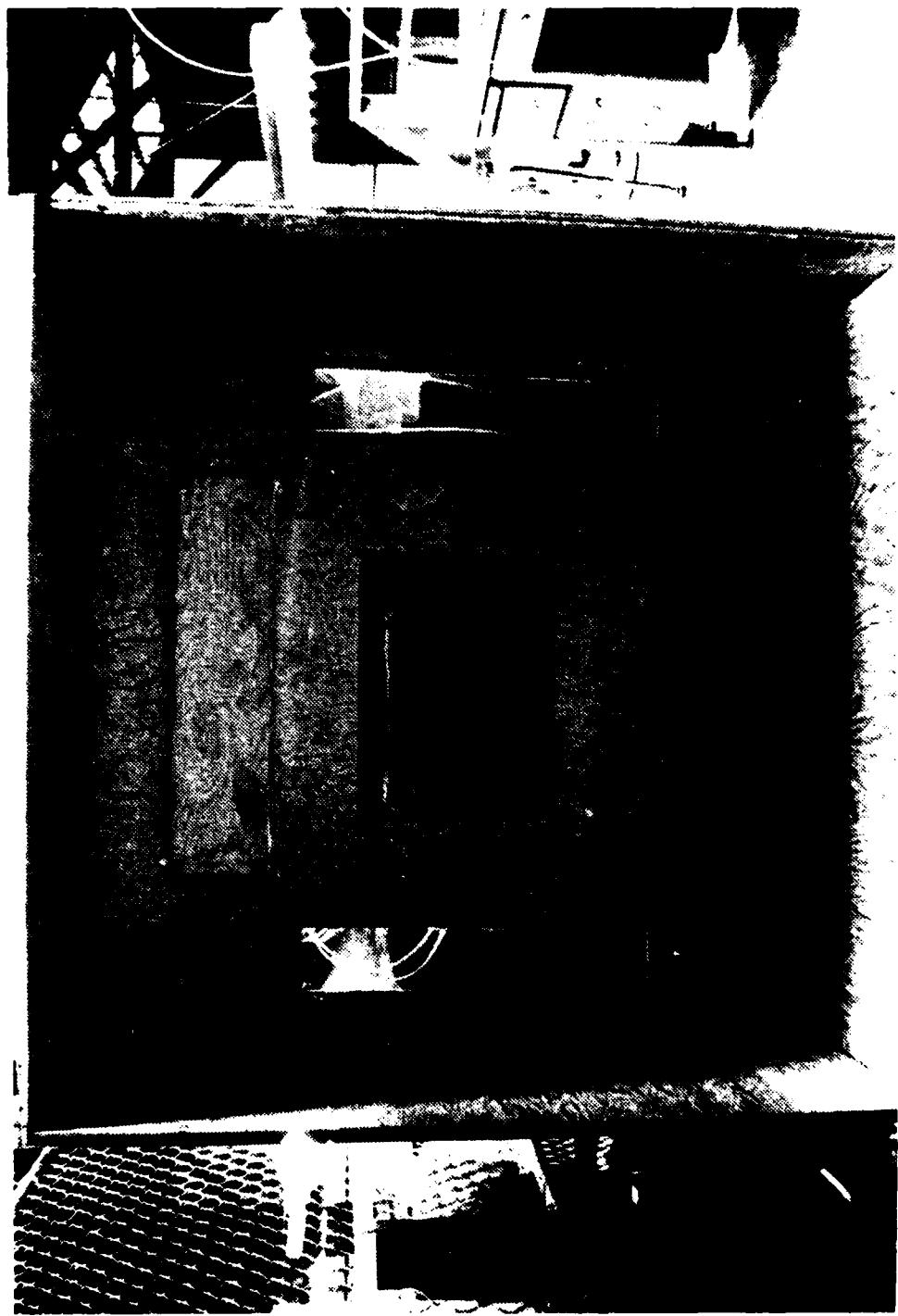


FIGURE 15. MIXING STACK EXIT WITH AIR SEAL

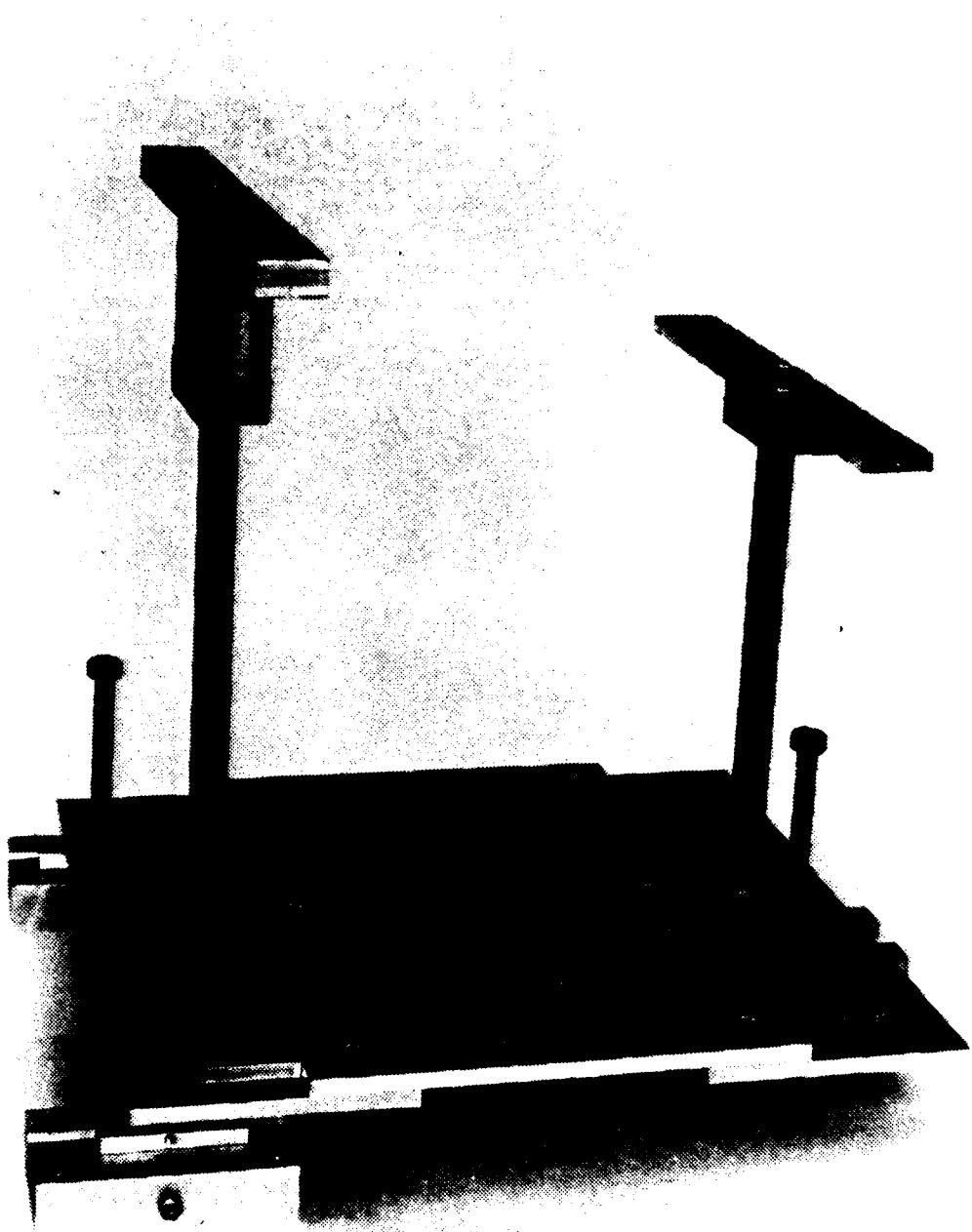


FIGURE 16. MIXING STACK MOUNTING STAND

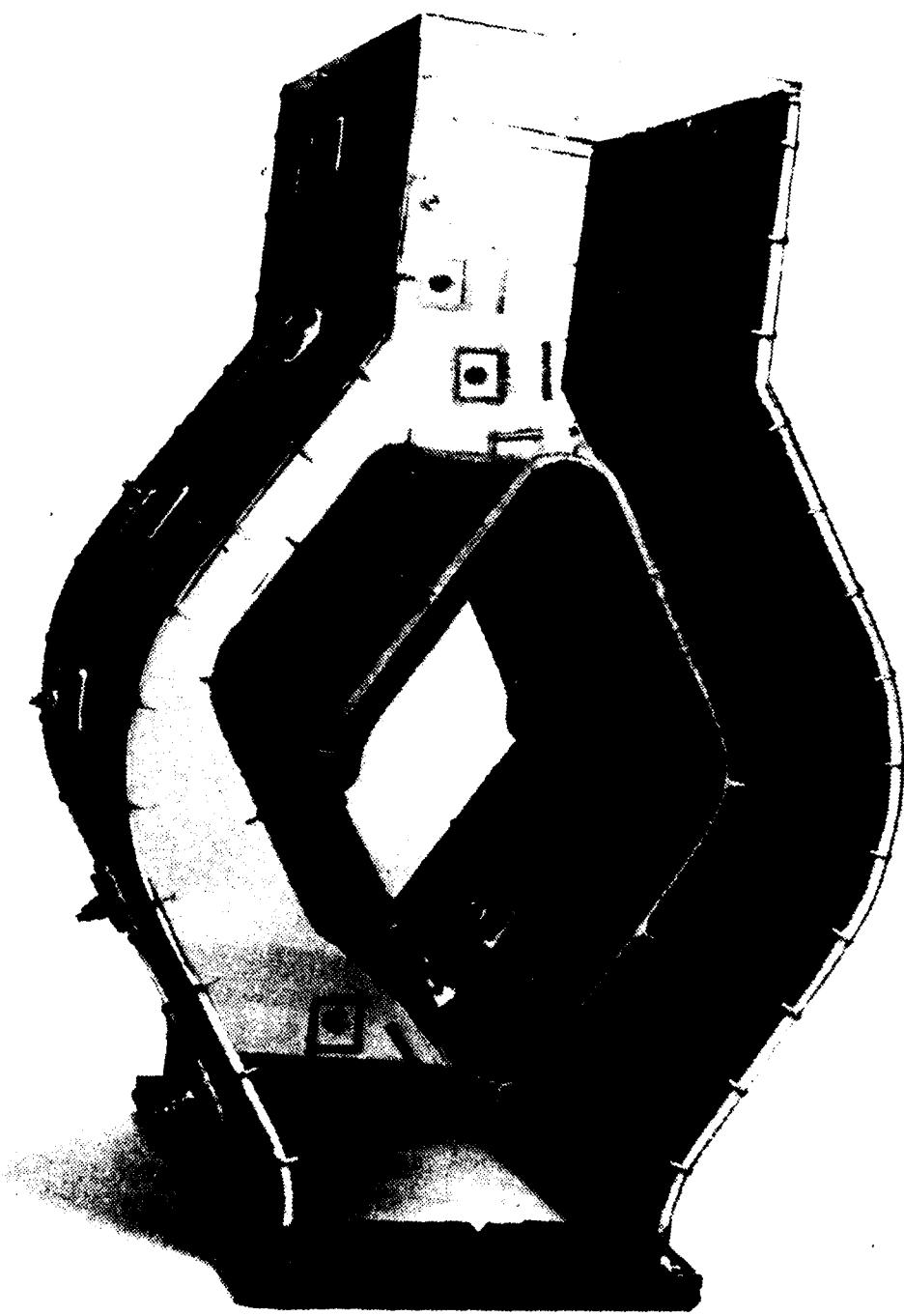


FIGURE 17. MIXING STACK WITH PORTED AND SHROUDED PLUG

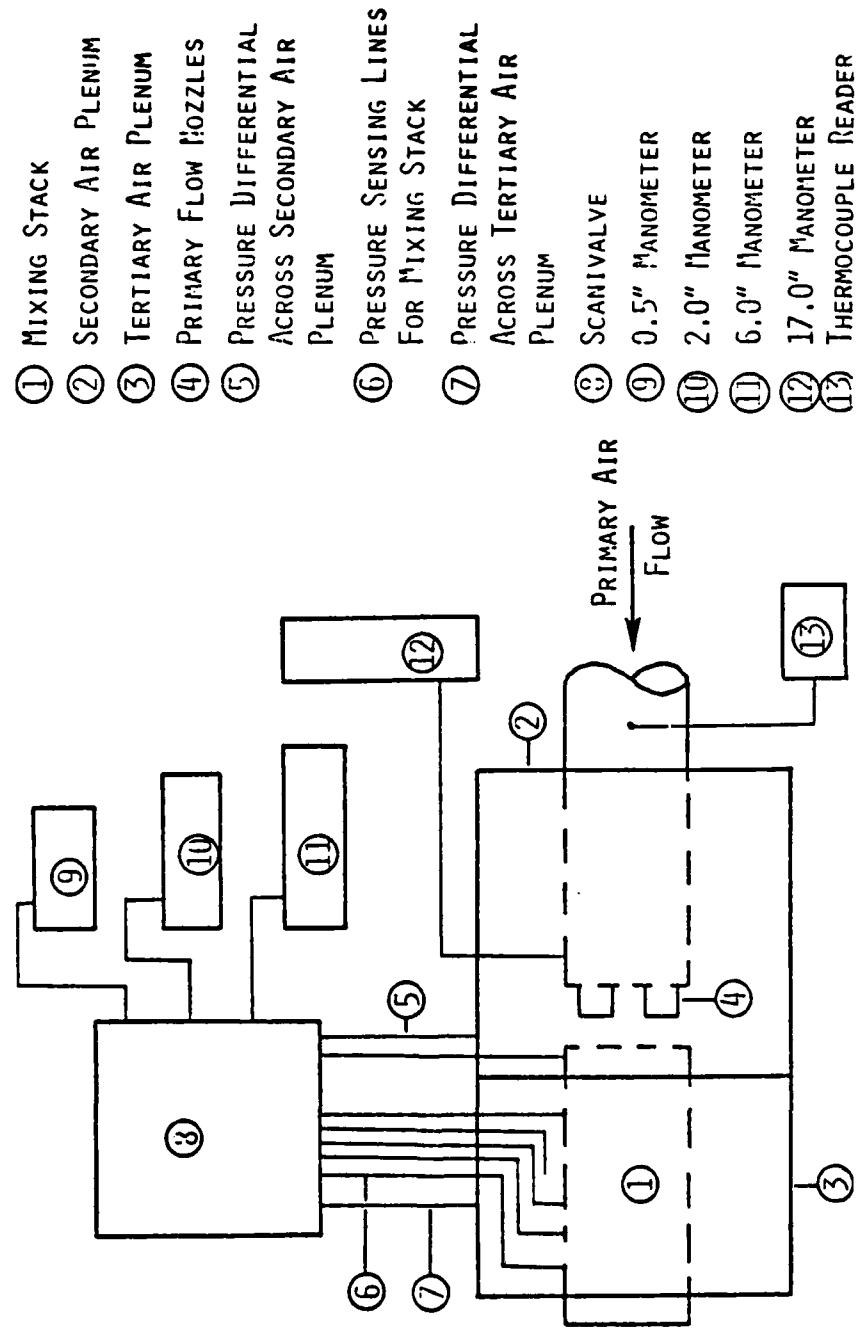


FIGURE 18. SCHEMATIC OF INSTRUMENTATION

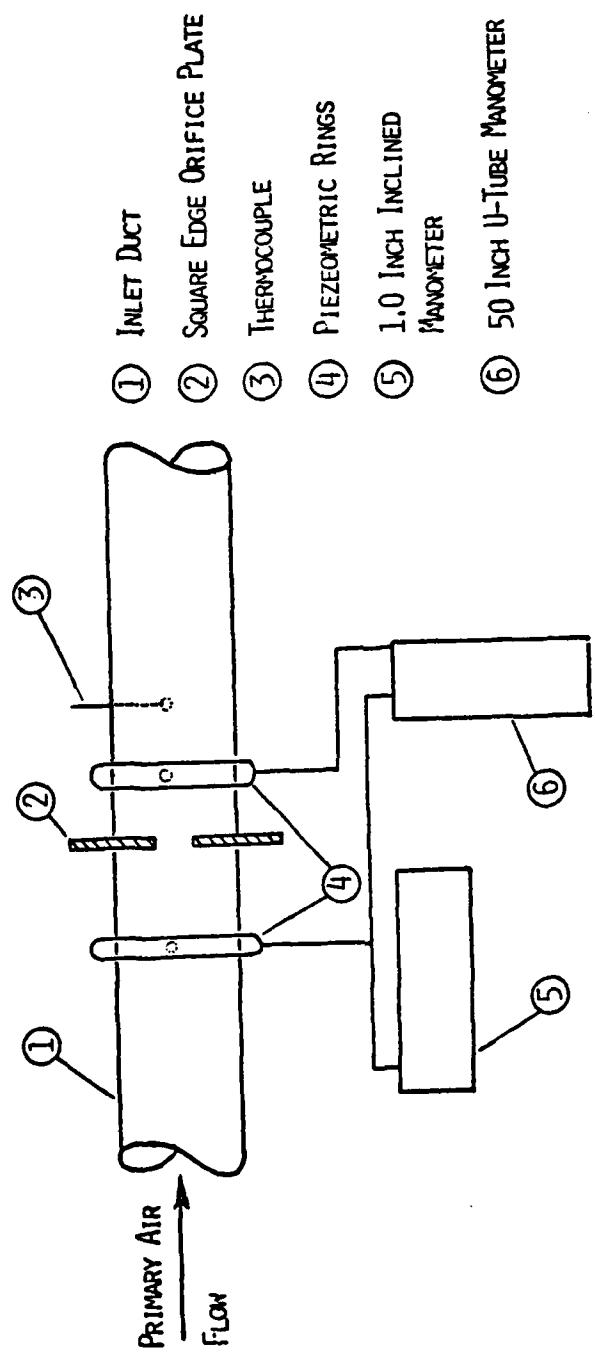


FIGURE 19. SCHEMATIC OF INSTRUMENTATION FOR PRIMARY AIR FLOW MEASUREMENT

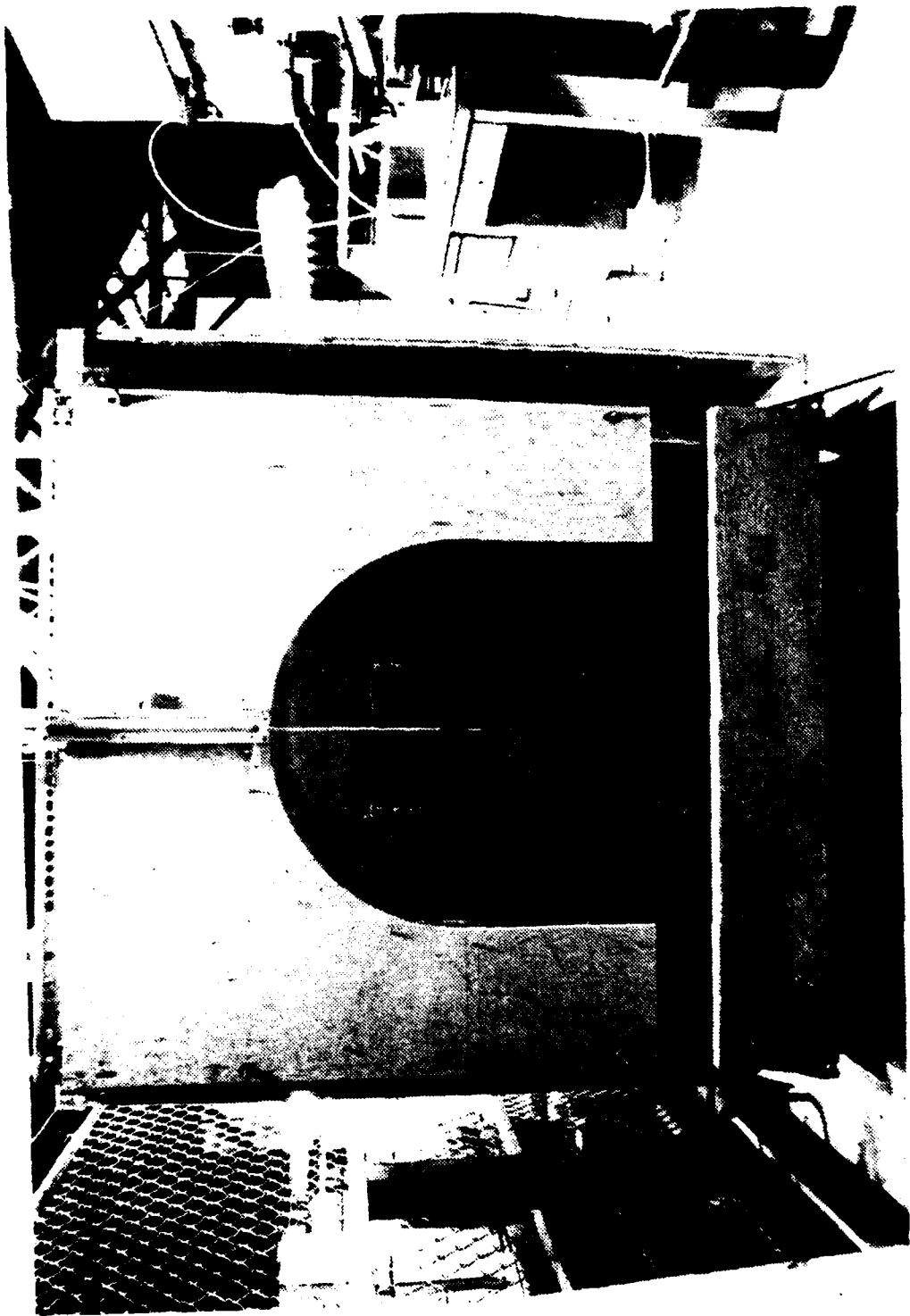


FIGURE 20. VELOCITY PROFILE APPARATUS

EXPERIMENTAL PUMPING
COEFFICIENT

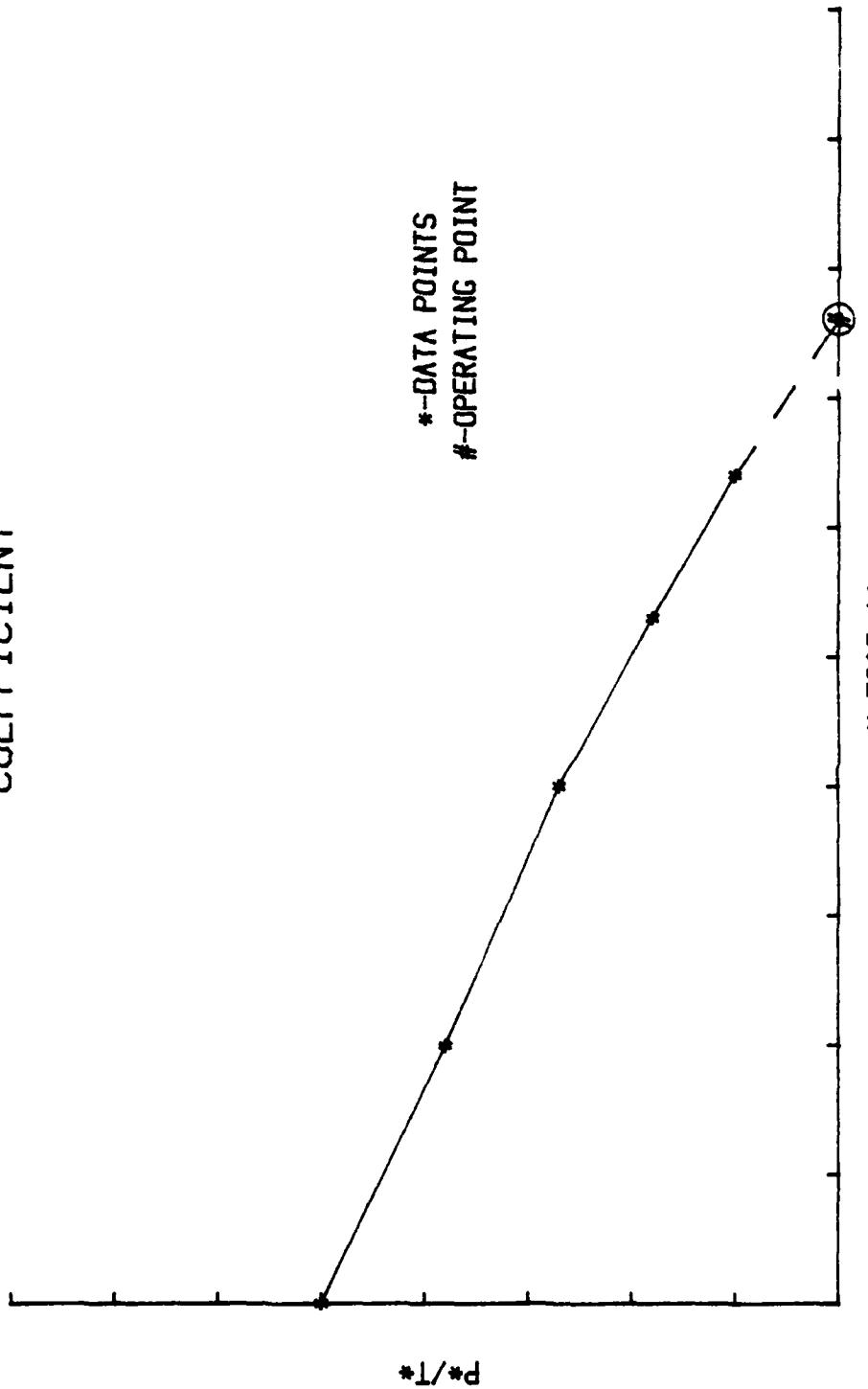


FIGURE 21. EXAMPLE PLOT OF PUMPING COEFFICIENT

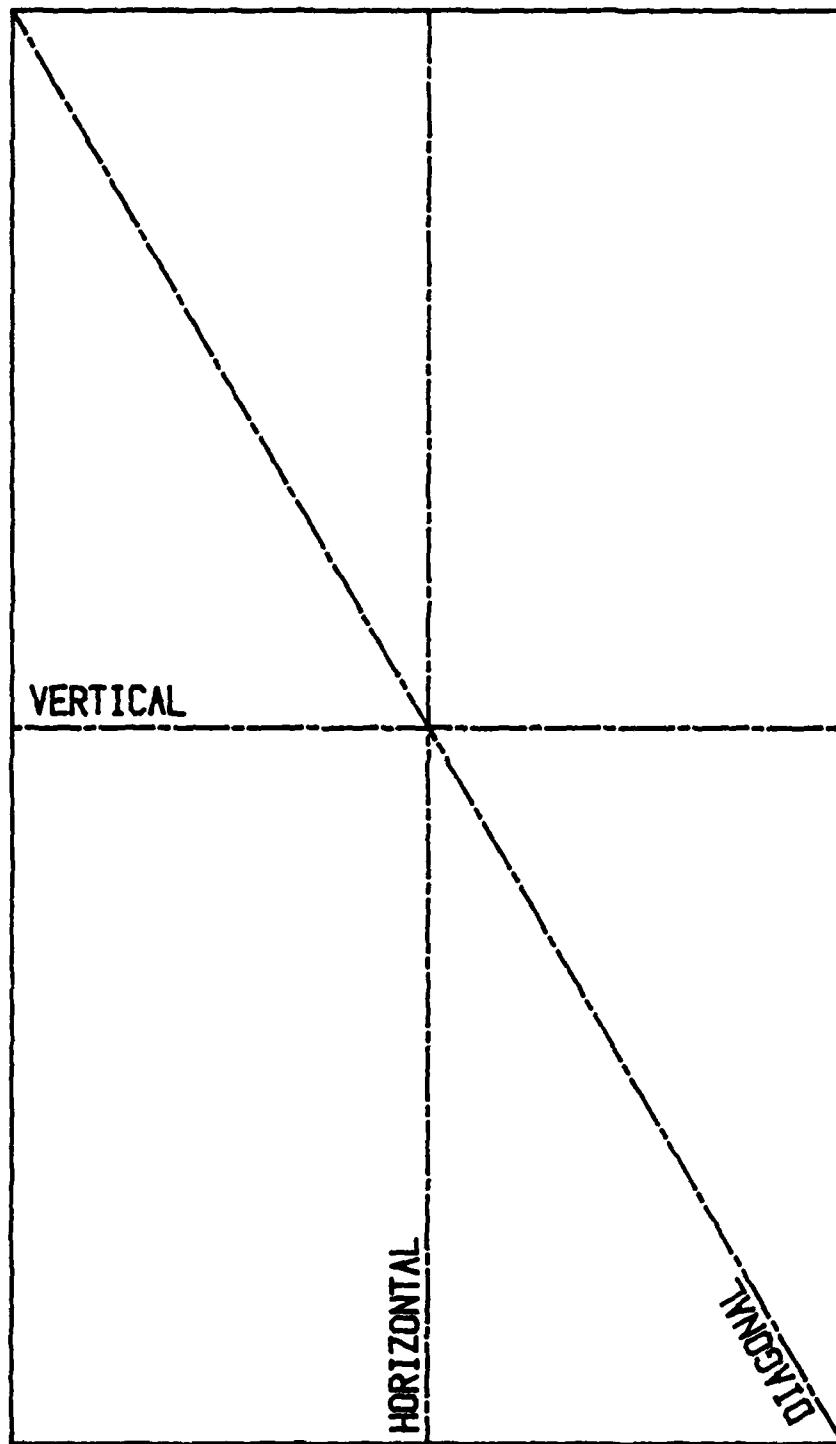


FIGURE 22. VELOCITY TRAVERSE DIRECTIONS

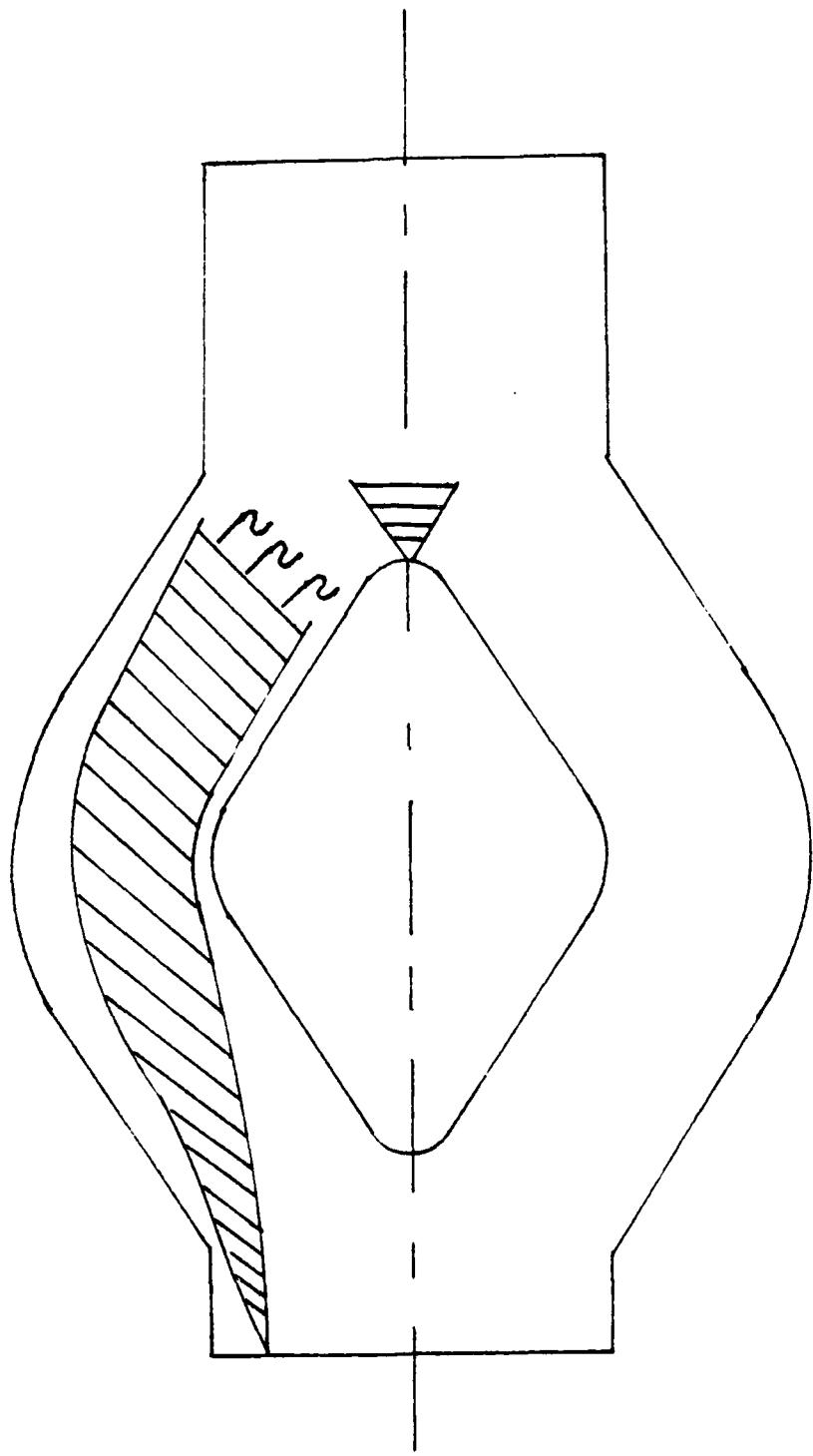


FIGURE 23. SCHEMATIC OF FLOW VISUALIZATION RESULTS

EXPERIMENTAL PUMPING
COEFFICIENT COMPARISON

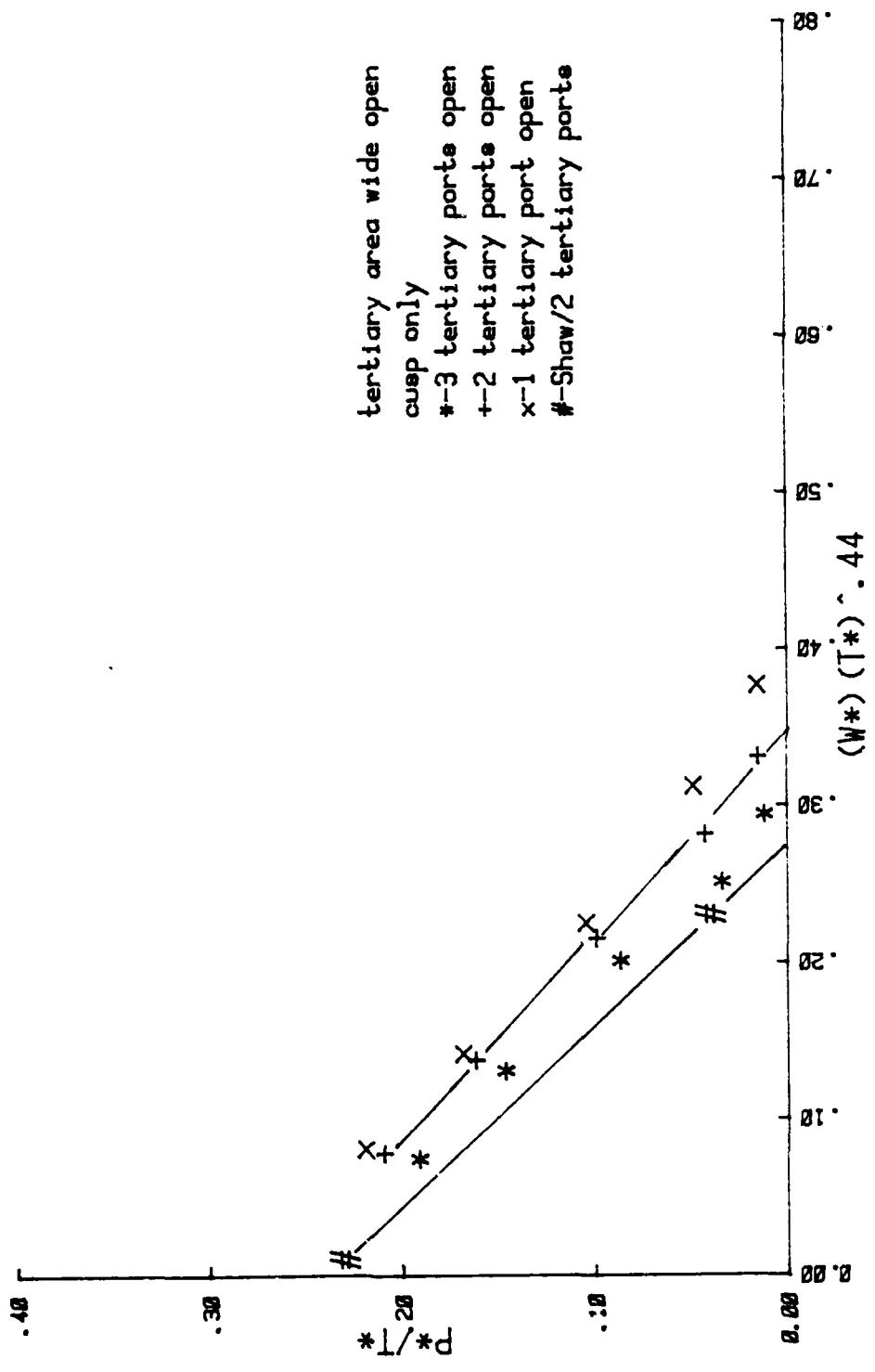


FIGURE 24a

EXPERIMENTAL PUMPING
COEFFICIENT COMPARISON

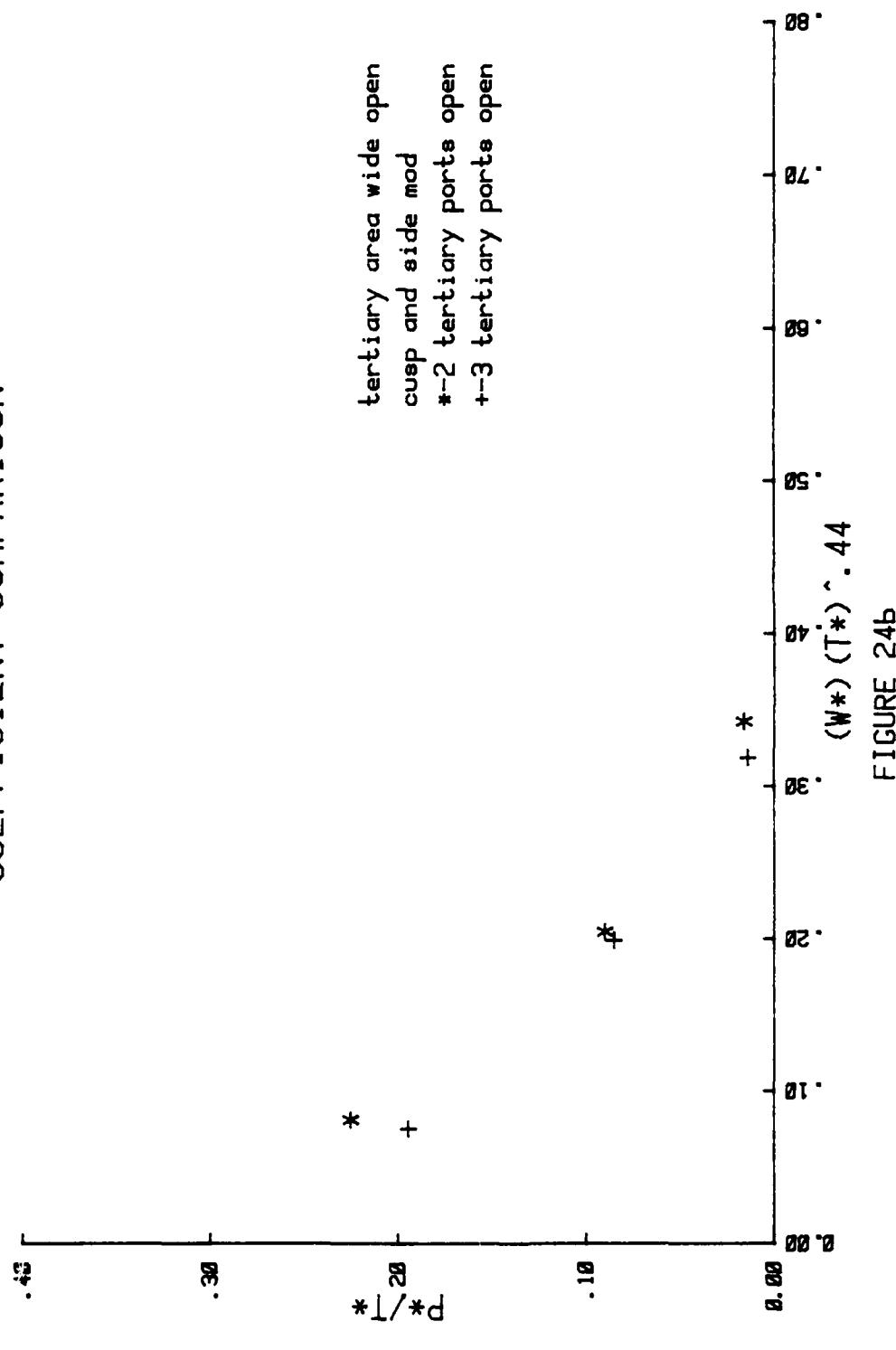


FIGURE 24b

EXPERIMENTAL PUMPING
COEFFICIENT COMPARISON

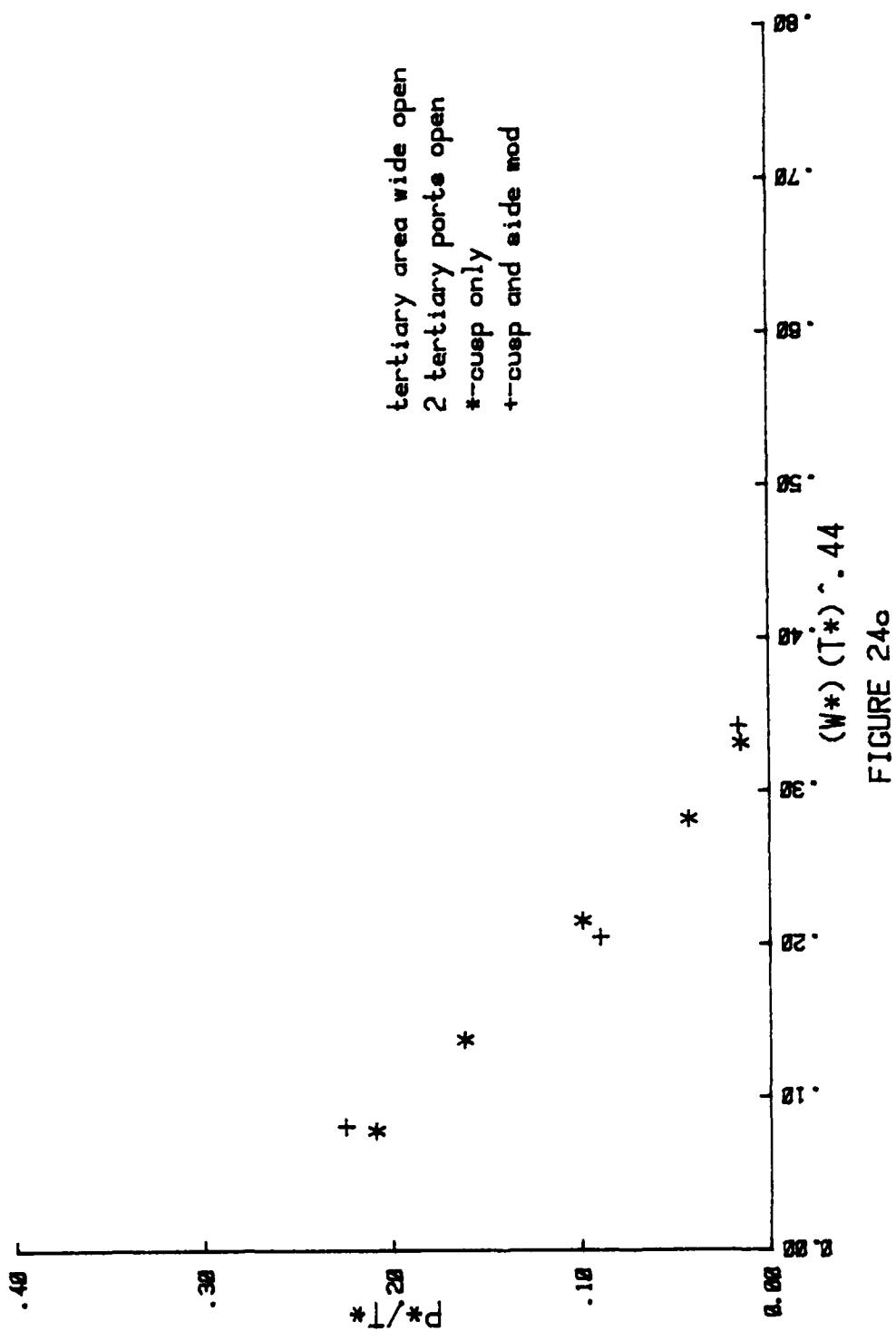


FIGURE 24a

EXPERIMENTAL FILM PUMPING
COEFFICIENT COMPARISON

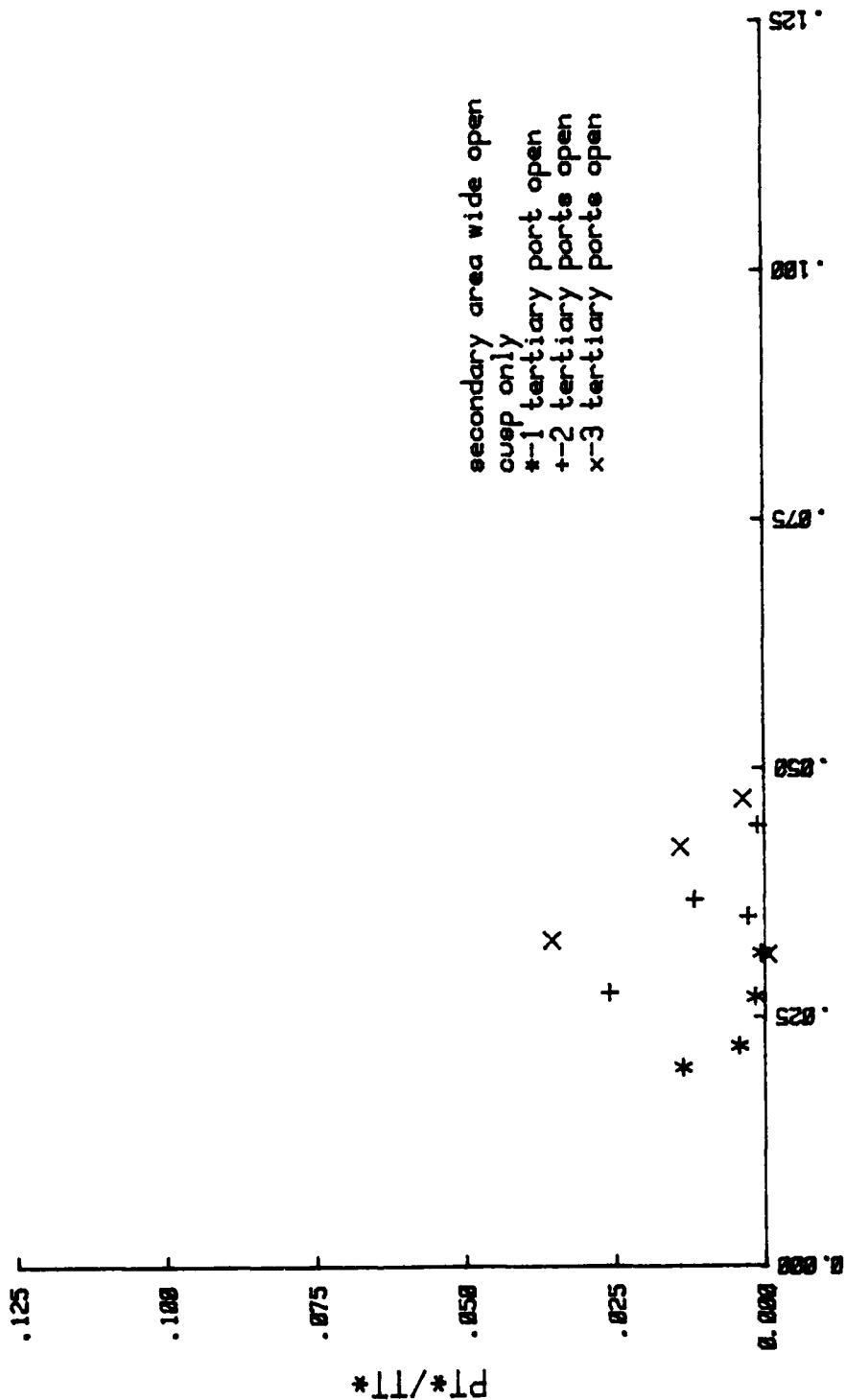


FIGURE 25a

EXPERIMENTAL FILM PUMPING
COEFFICIENT COMPARISON

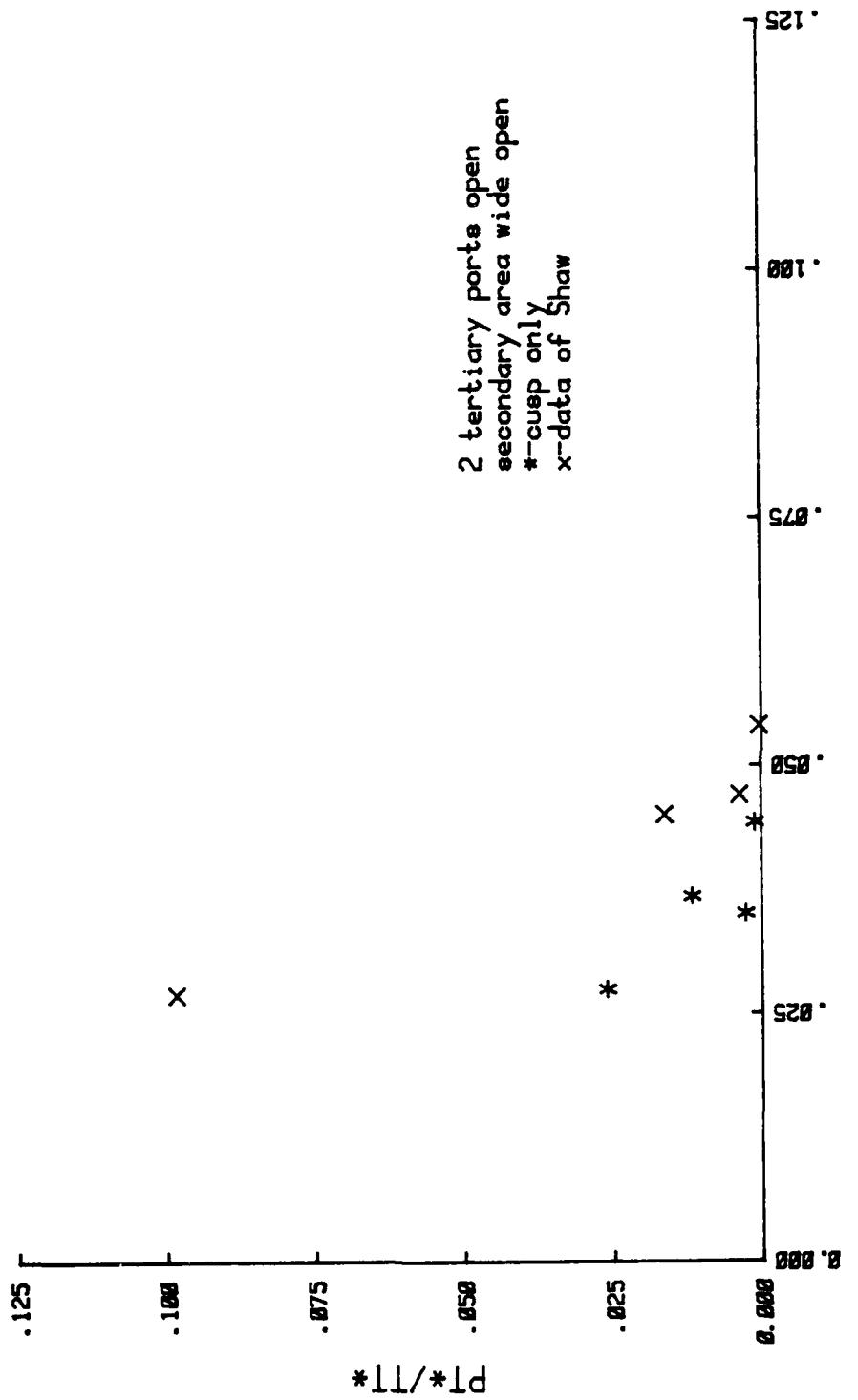
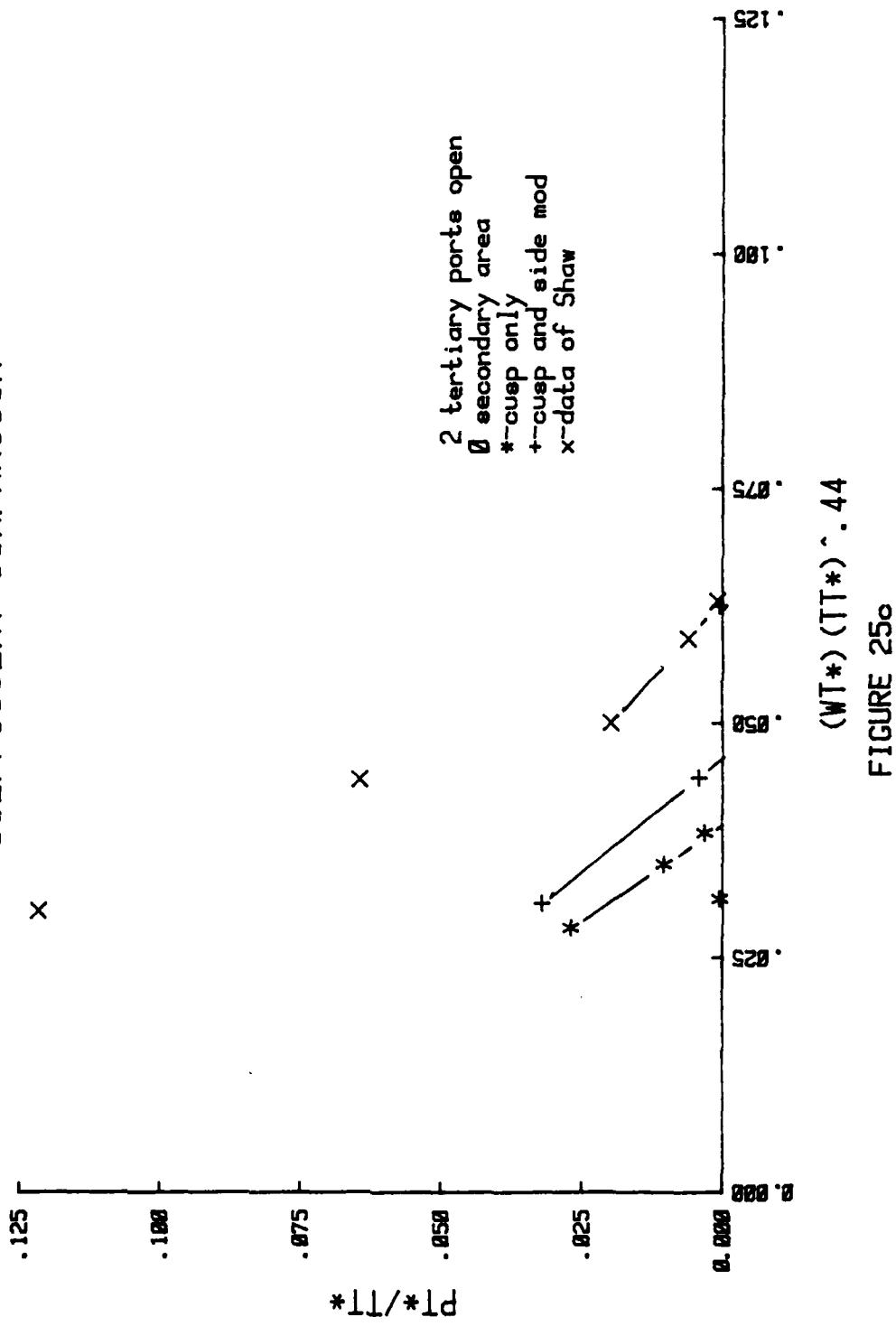


FIGURE 25b

EXPERIMENTAL FILM PUMPING
COEFFICIENT COMPARISON



EXPERIMENTAL FILM PUMPING
COEFFICIENT COMPARISON

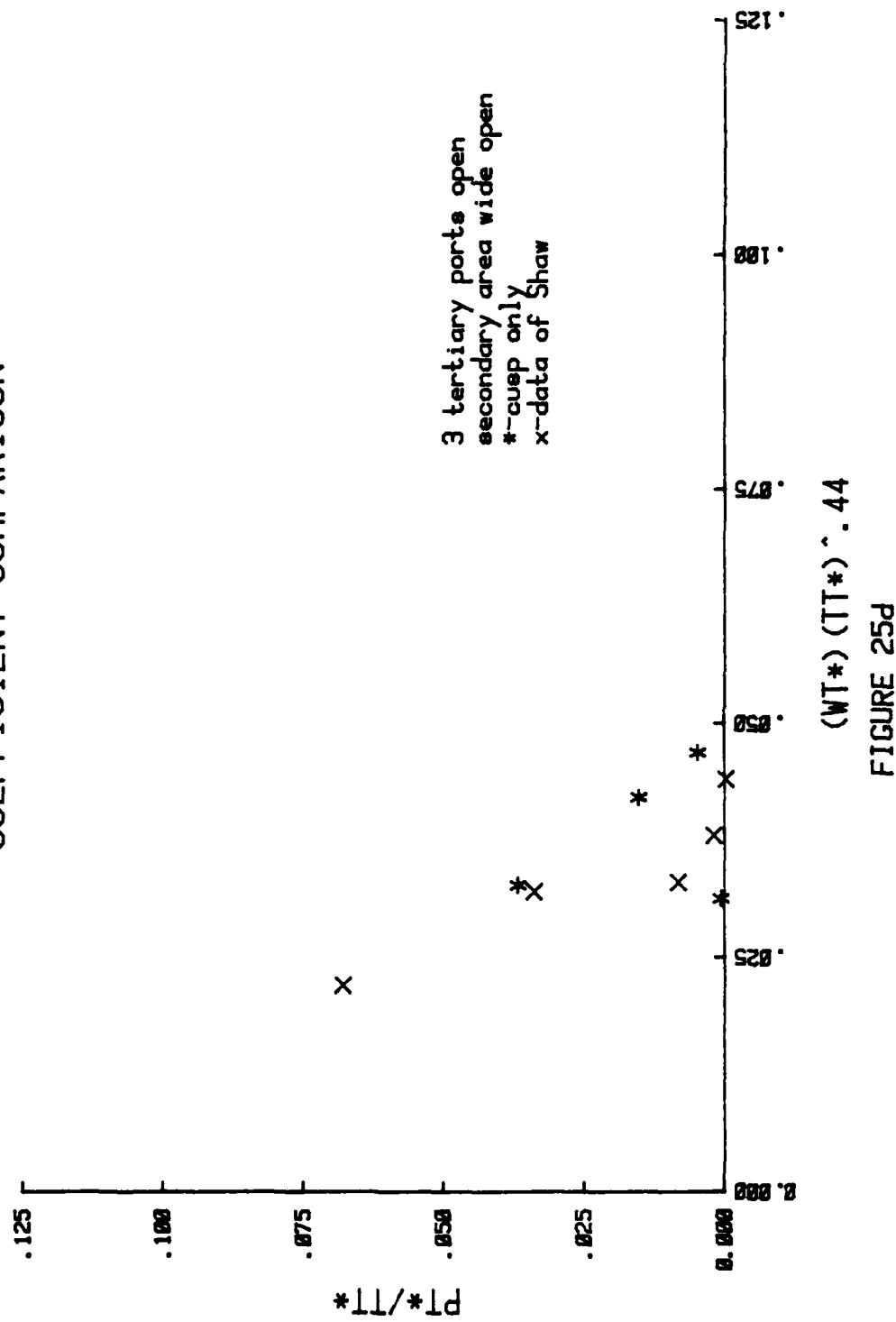


FIGURE 25d

EXPERIMENTAL FILM PUMPING
COEFFICIENT COMPARISON

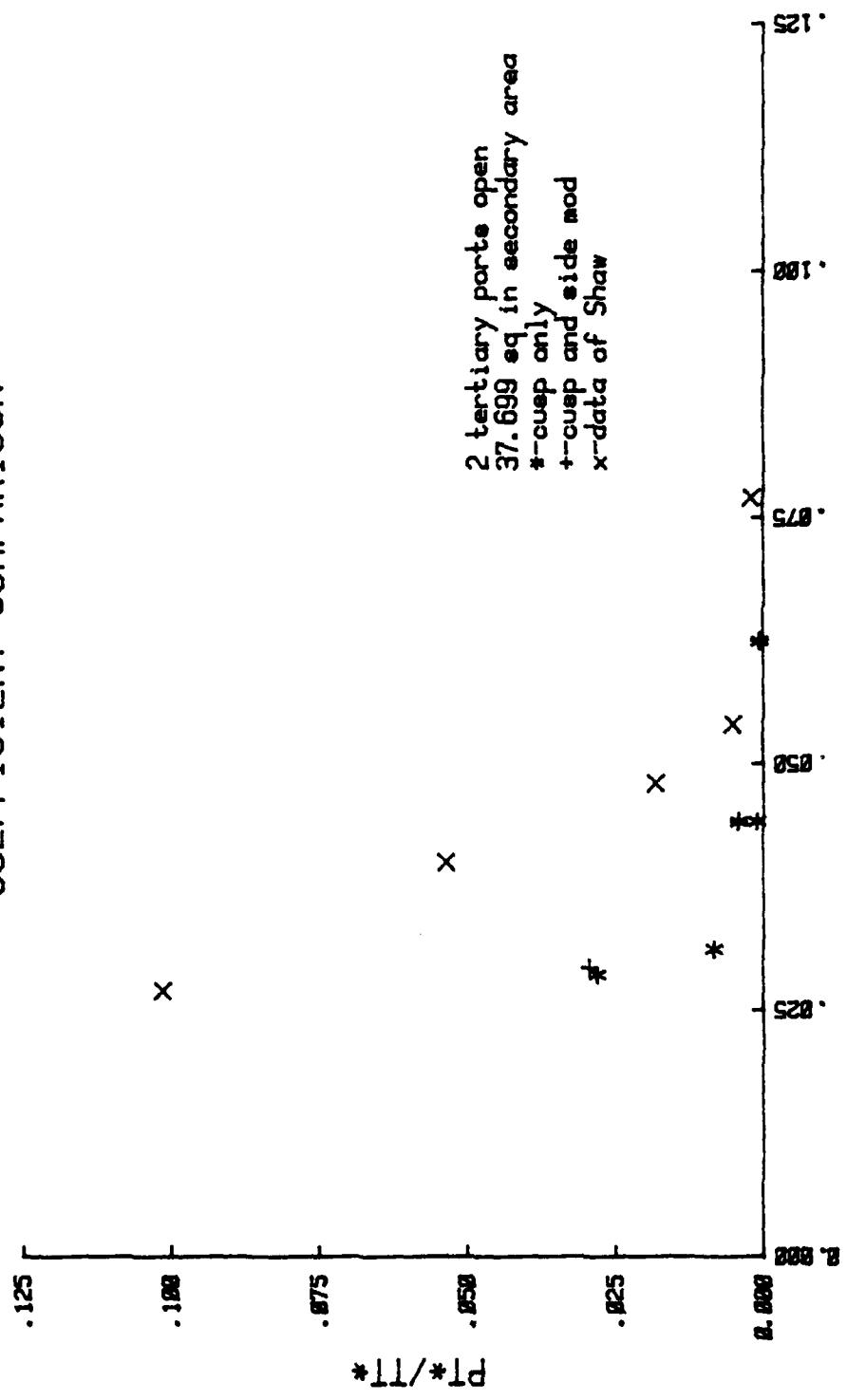
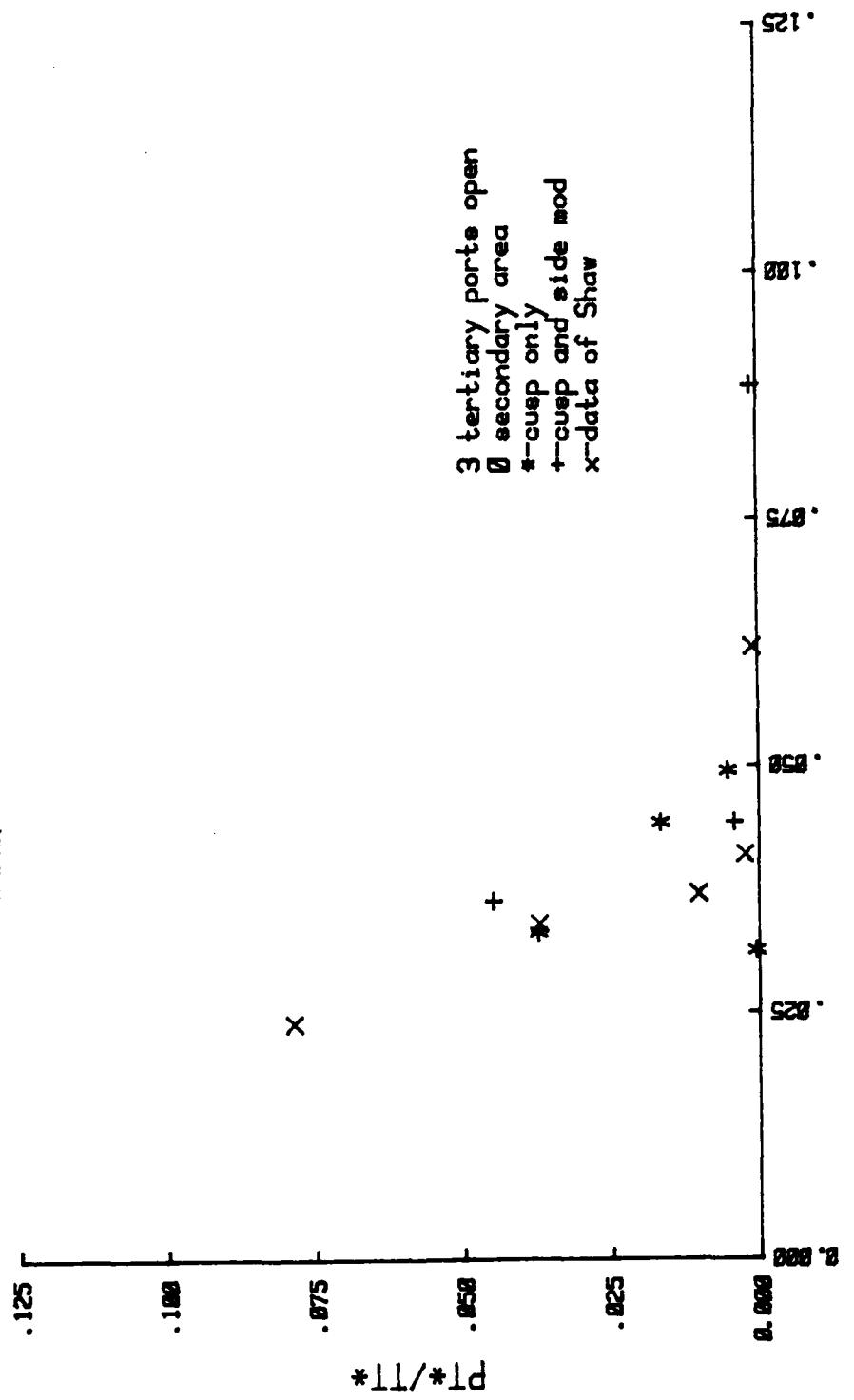


FIGURE 25e
 $(WT^*) (TT^*) = .44$

EXPERIMENTAL FILM PUMPING
COEFFICIENT COMPARISON



$(WT^*) (TT^*)^{.44}$
FIGURE 25f

EXPERIMENTAL FILM PUMPING
COEFFICIENT COMPARISON

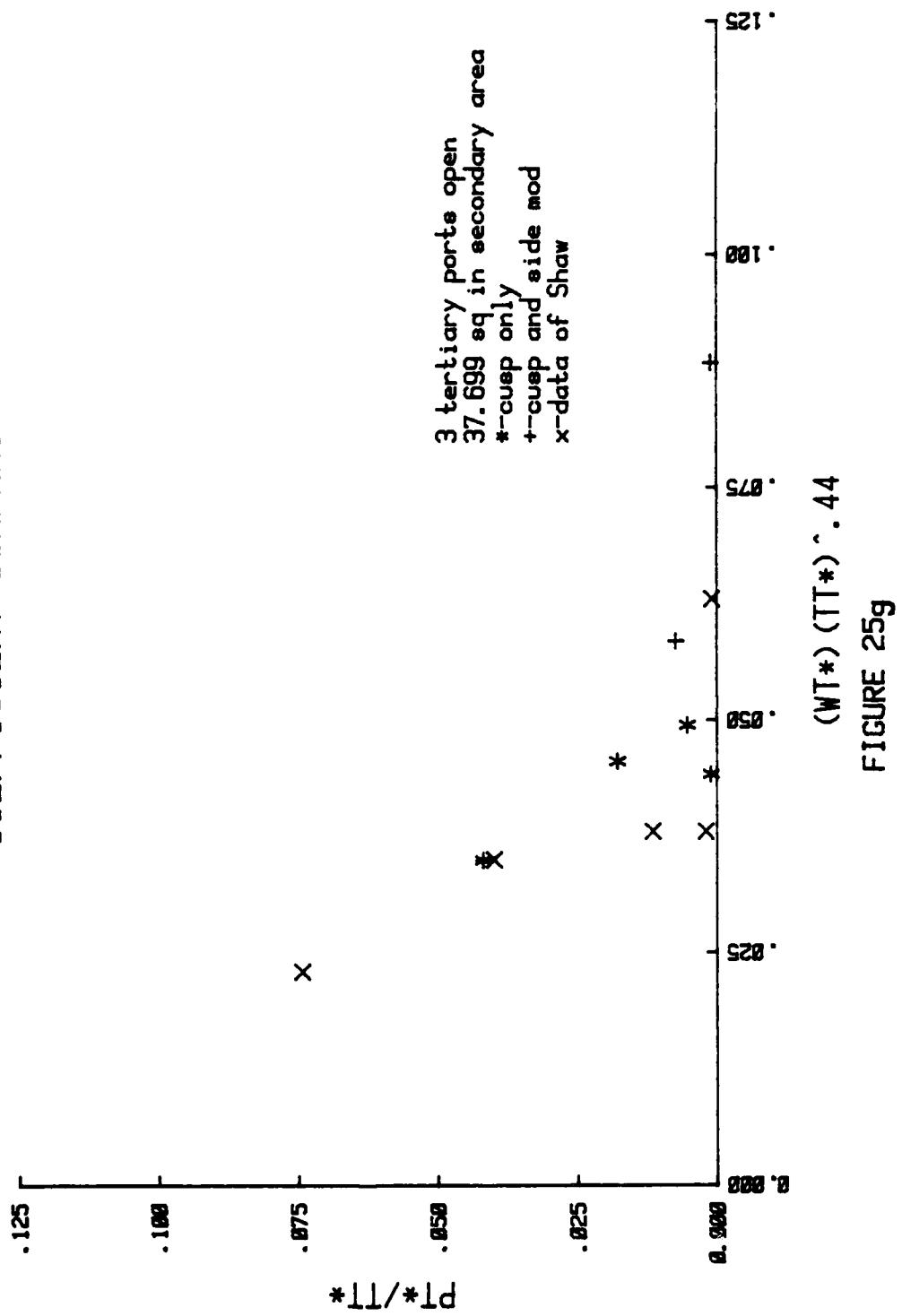
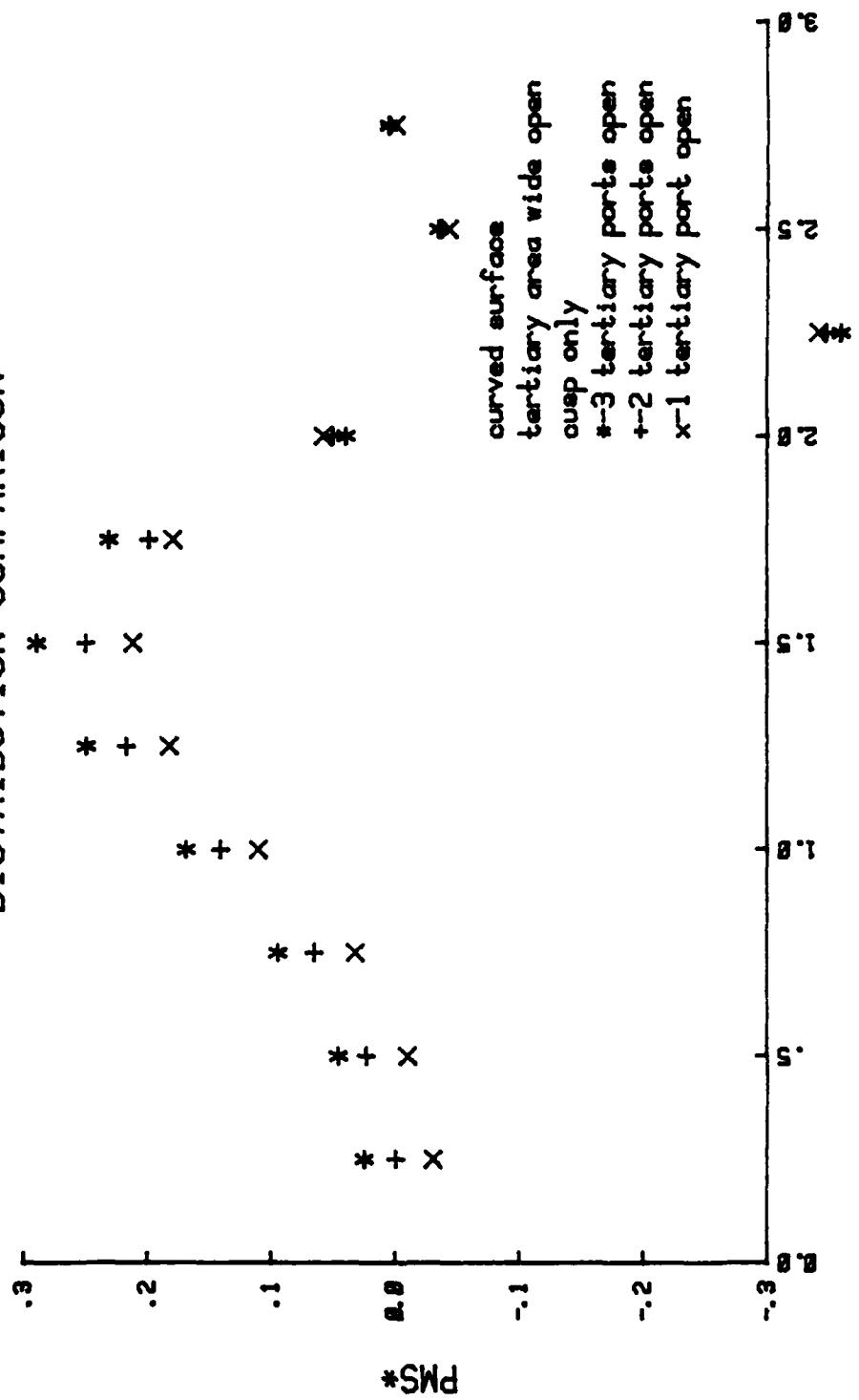


FIGURE 25g

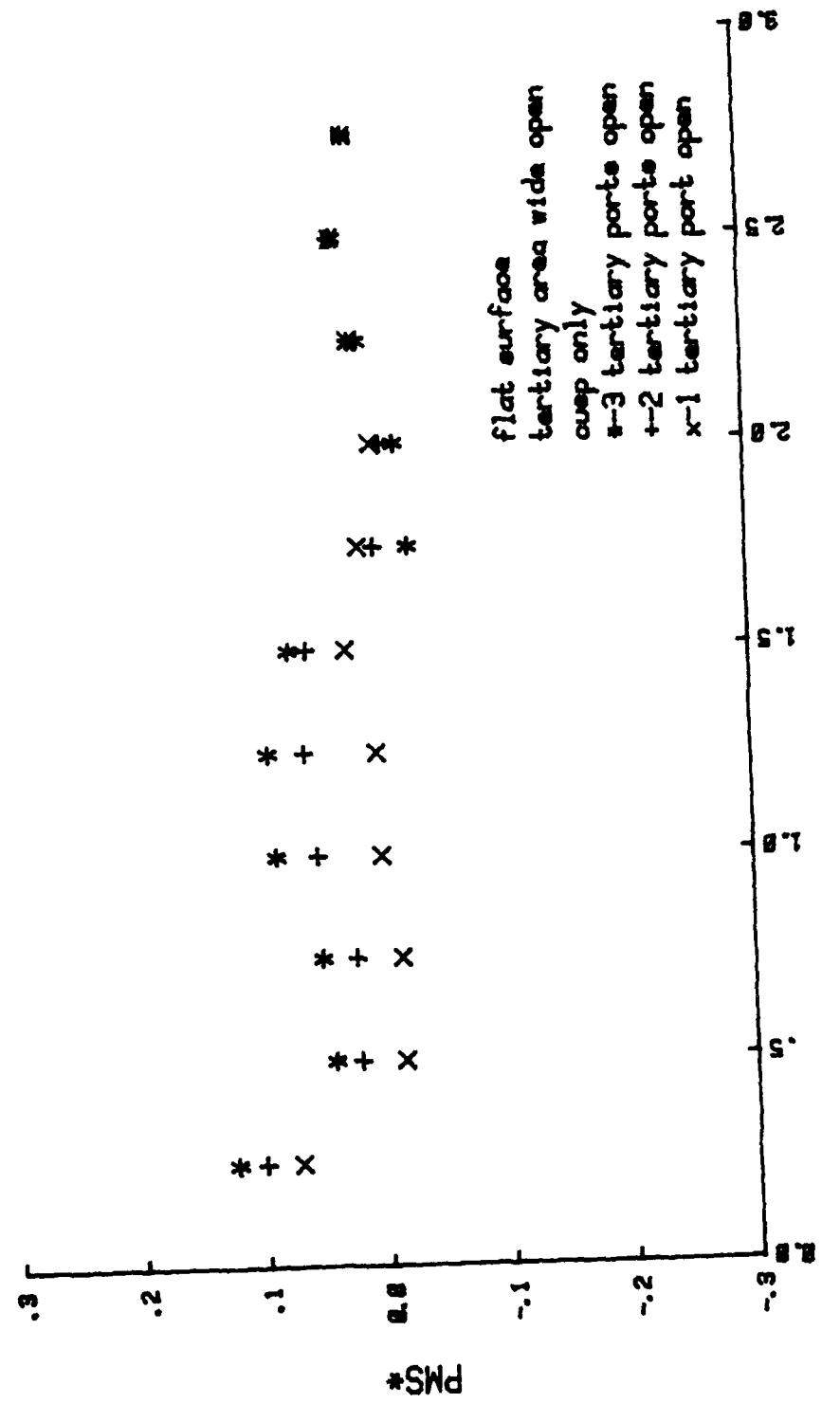
AXIAL PRESSURE DISTRIBUTION COMPARISON



AXIAL DISTANCE (X/S)

FIGURE 26a

AXIAL PRESSURE
DISTRIBUTION COMPARISON



AXIAL DISTANCE (X/S)

FIGURE 26b

AXIAL PRESSURE DISTRIBUTION COMPARISON

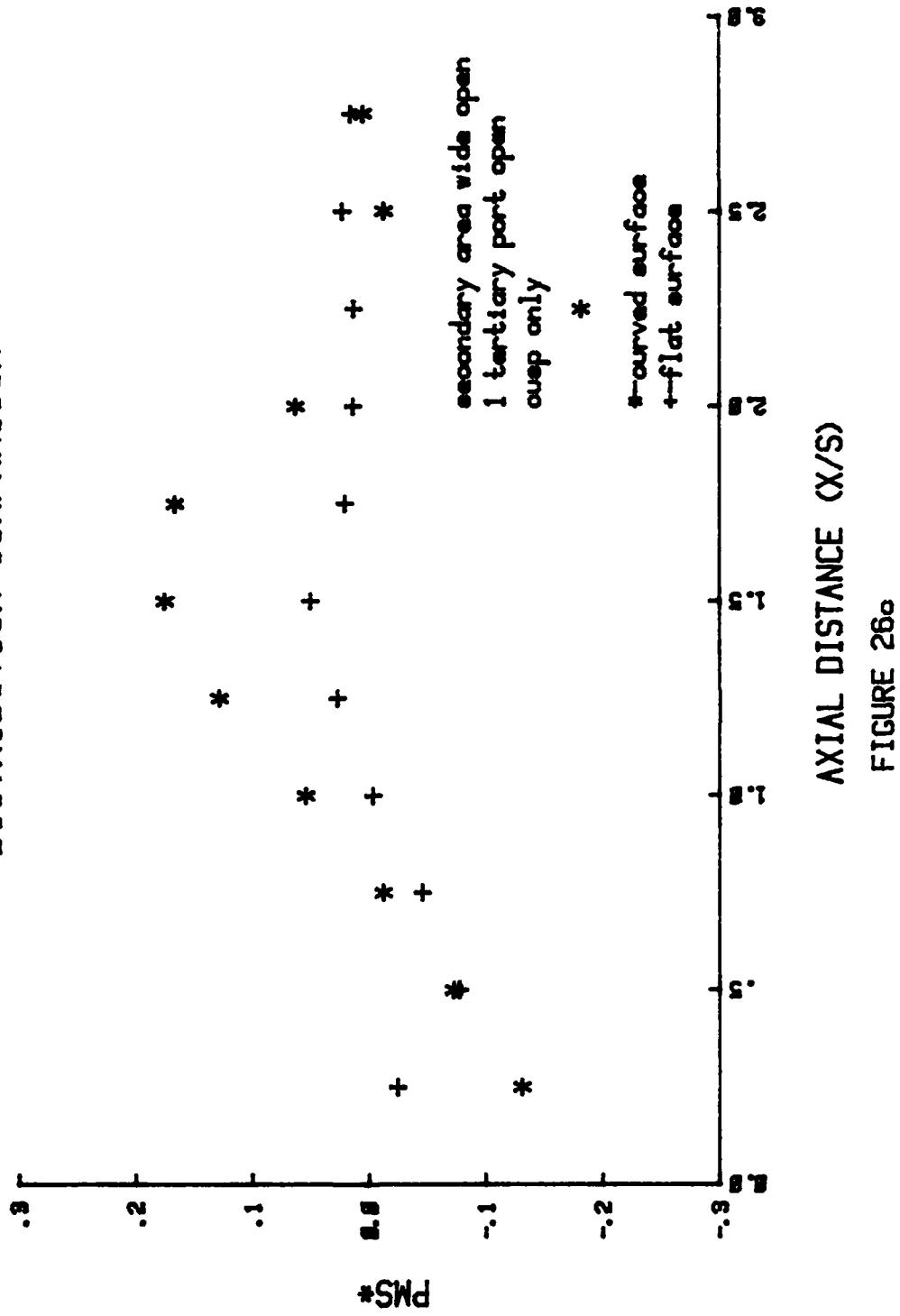


FIGURE 26a

AXIAL PRESSURE DISTRIBUTION COMPARISON

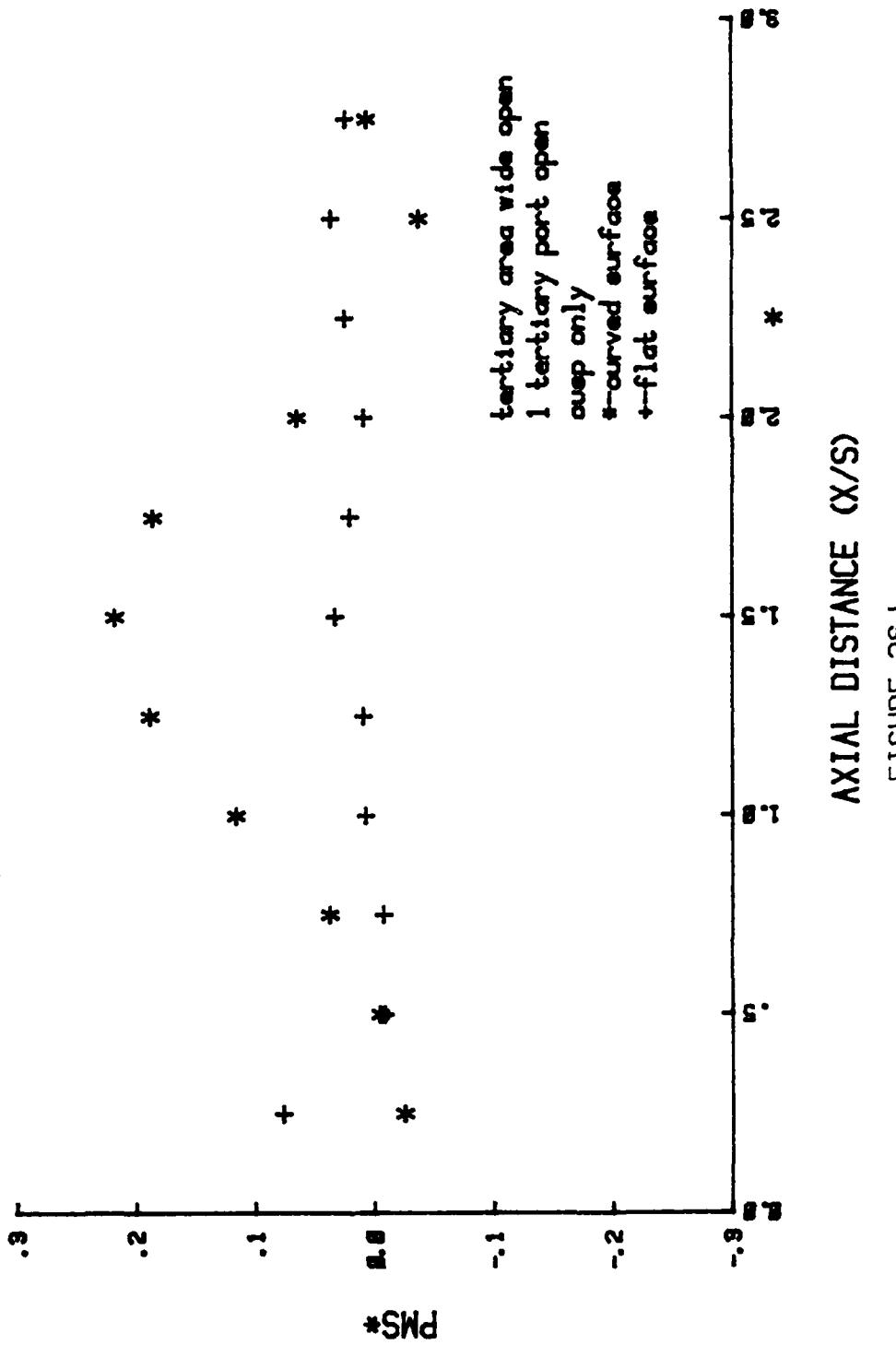


FIGURE 26P

AXIAL PRESSURE DISTRIBUTION COMPARISON

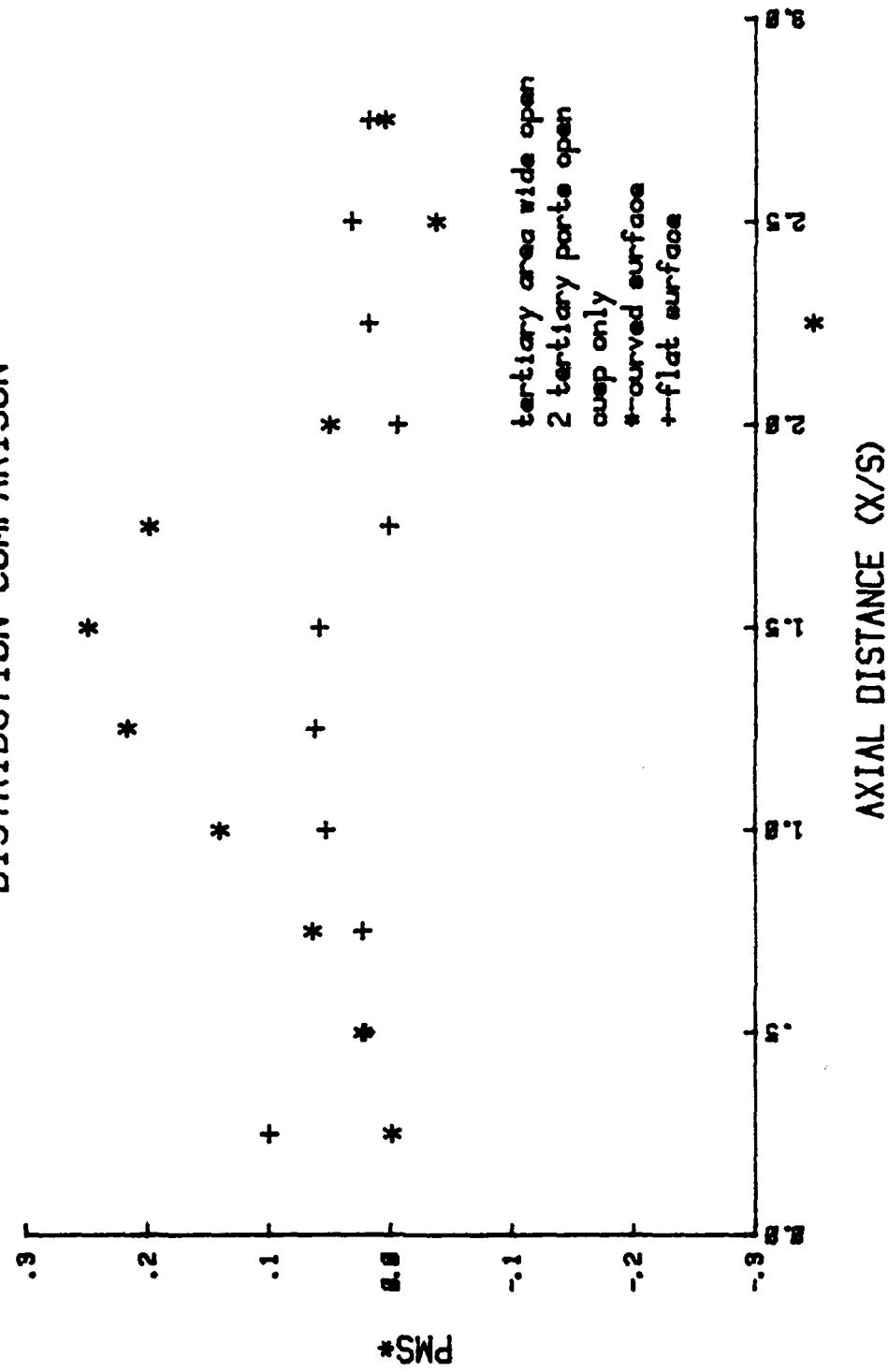


FIGURE 26e
AXIAL DISTANCE (X/S)

AXIAL PRESSURE
DISTRIBUTION COMPARISON

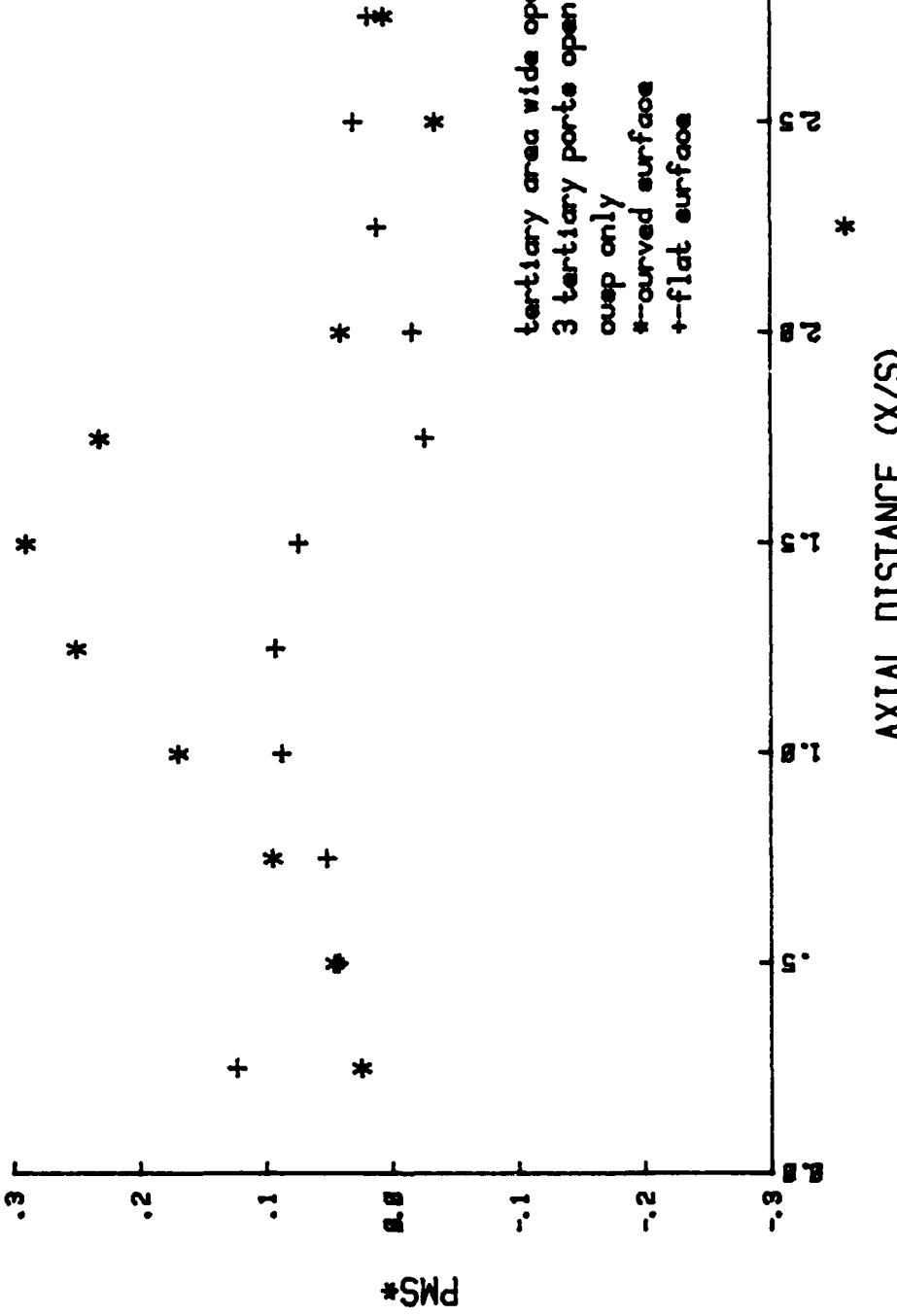


FIGURE 26f

AXIAL PRESSURE DISTRIBUTION COMPARISON

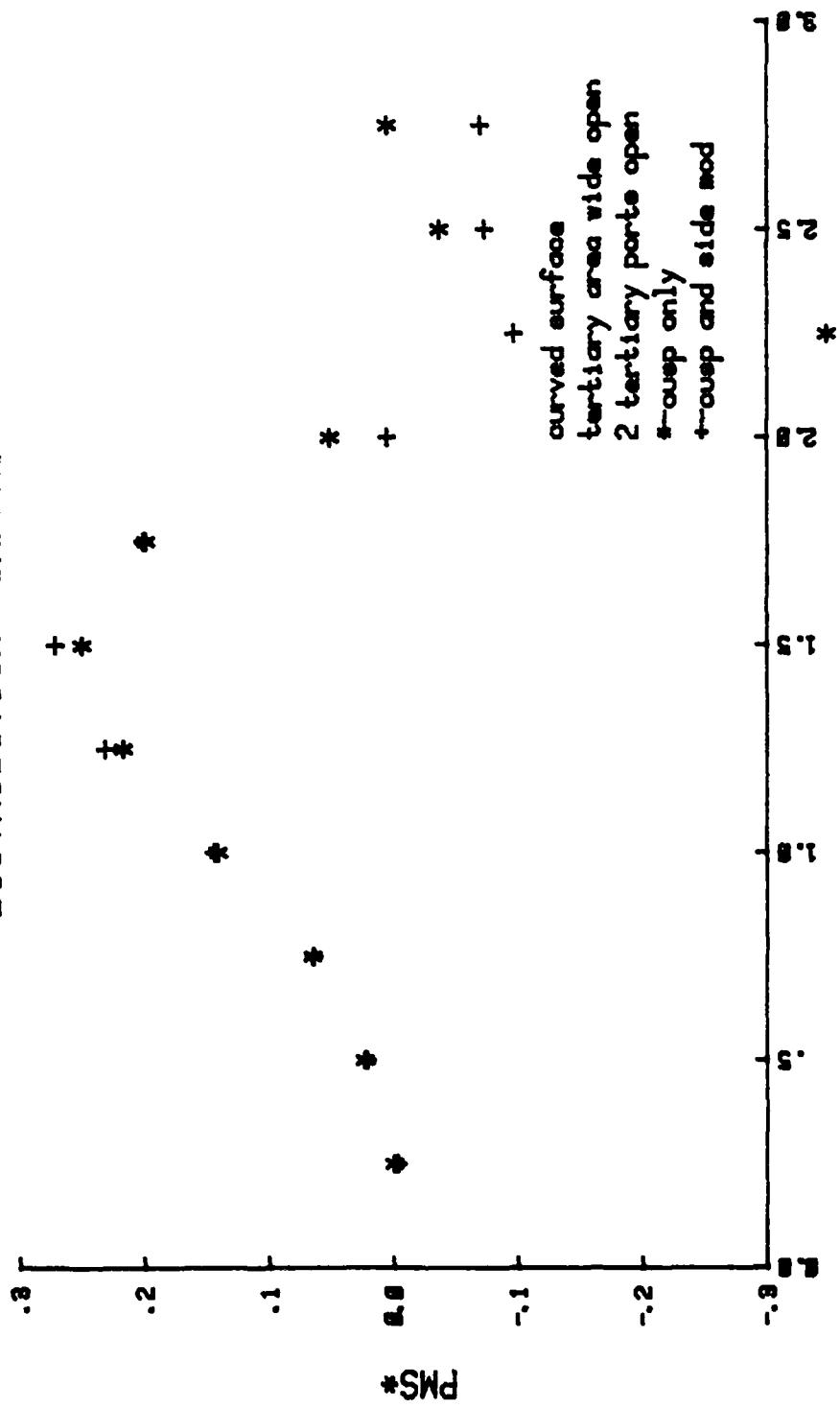
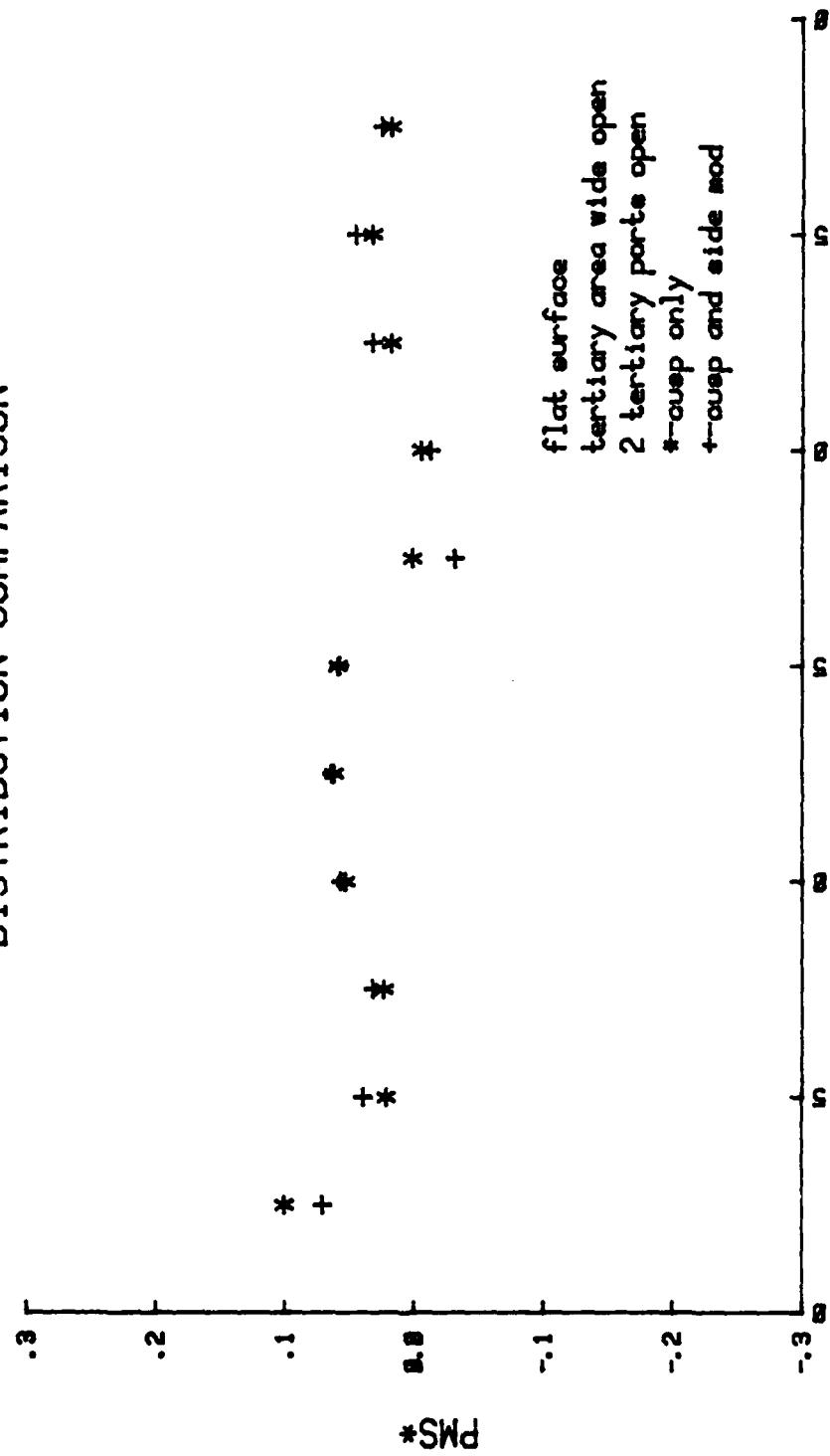


FIGURE 26g
AXIAL DISTANCE (X/S)

AXIAL PRESSURE DISTRIBUTION COMPARISON



AXIAL DISTANCE (X/S)

FIGURE 26h

AXIAL PRESSURE
DISTRIBUTION COMPARISON

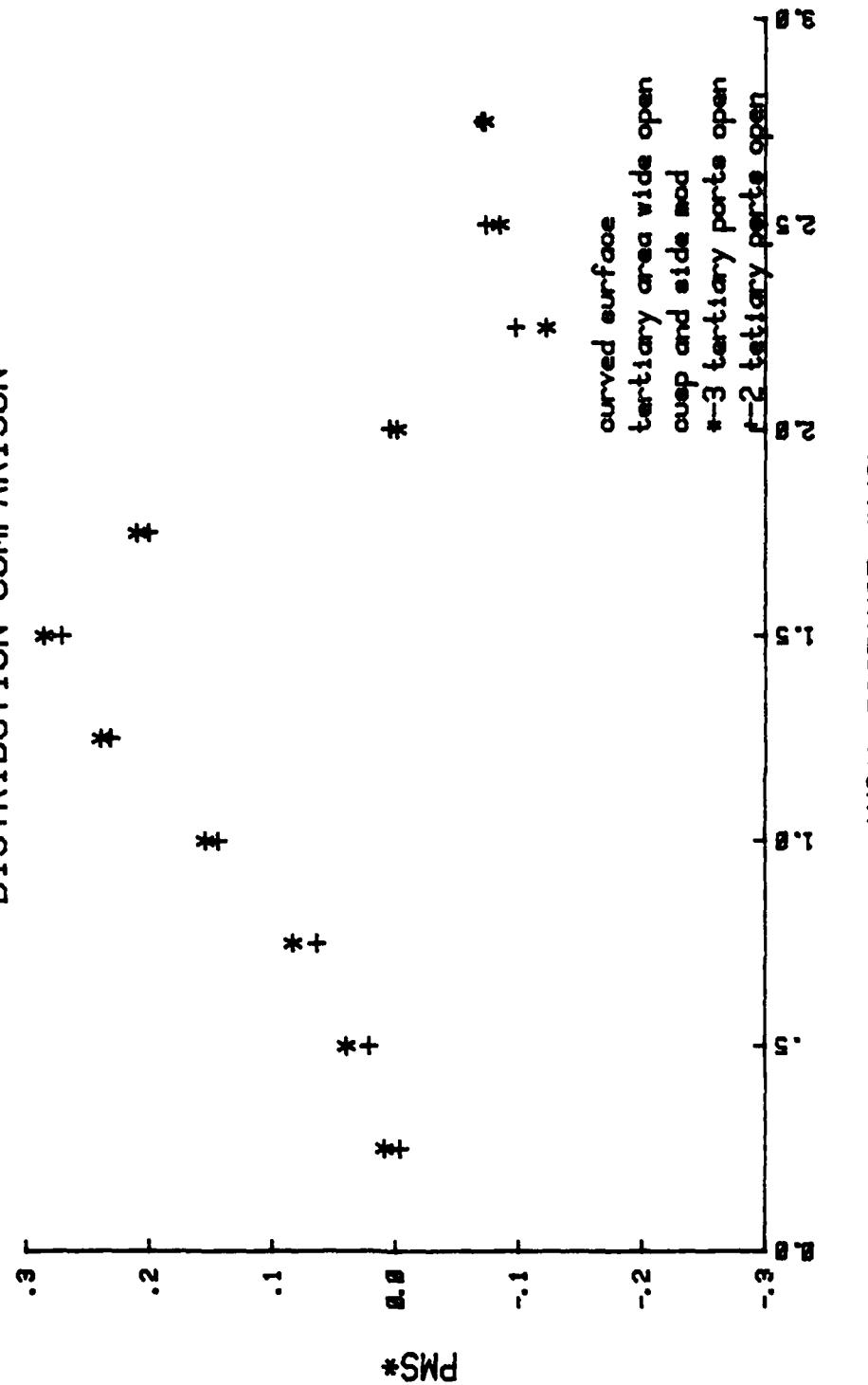


FIGURE 26i

AXIAL PRESSURE
DISTRIBUTION COMPARISON

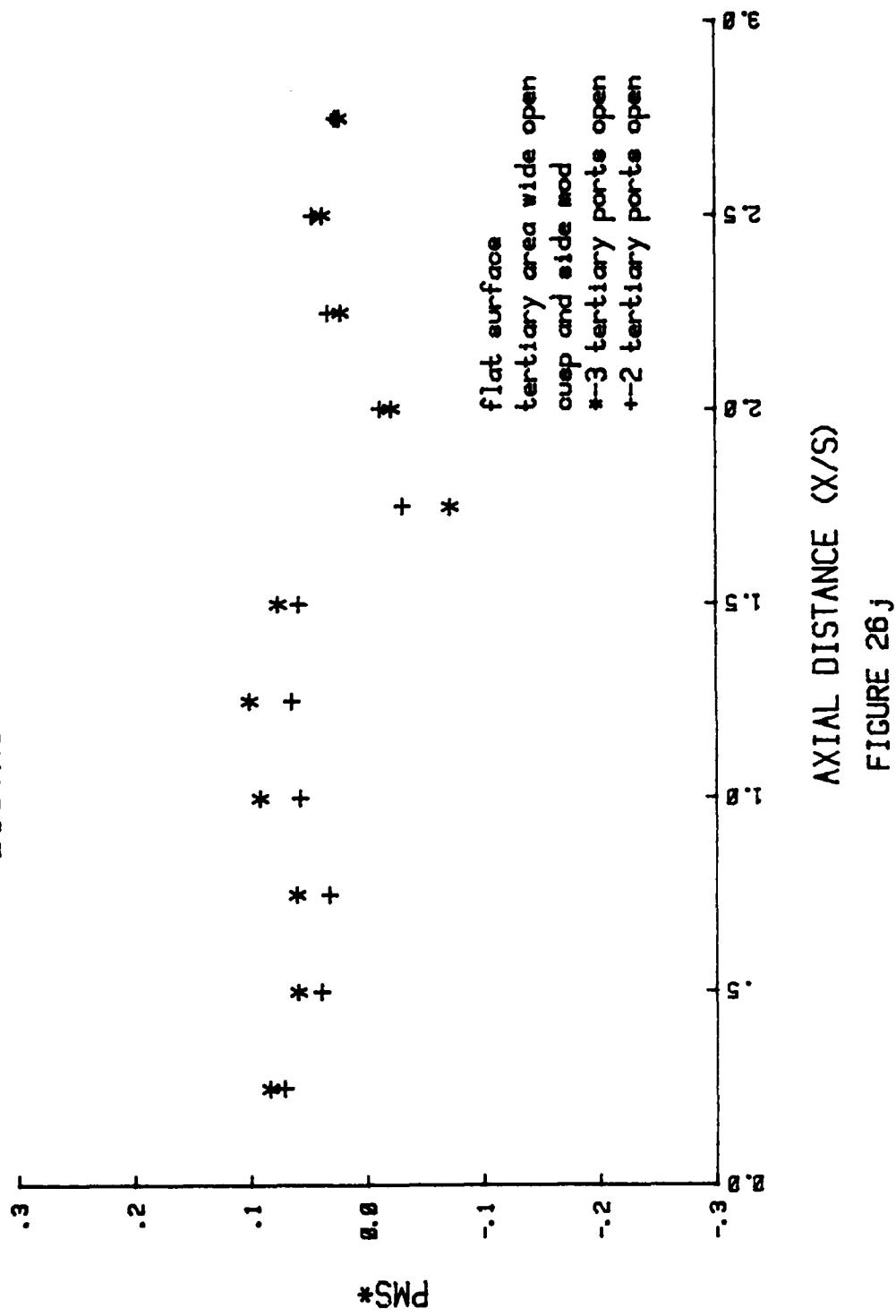


FIGURE 26j

AXIAL PRESSURE
DISTRIBUTION COMPARISON

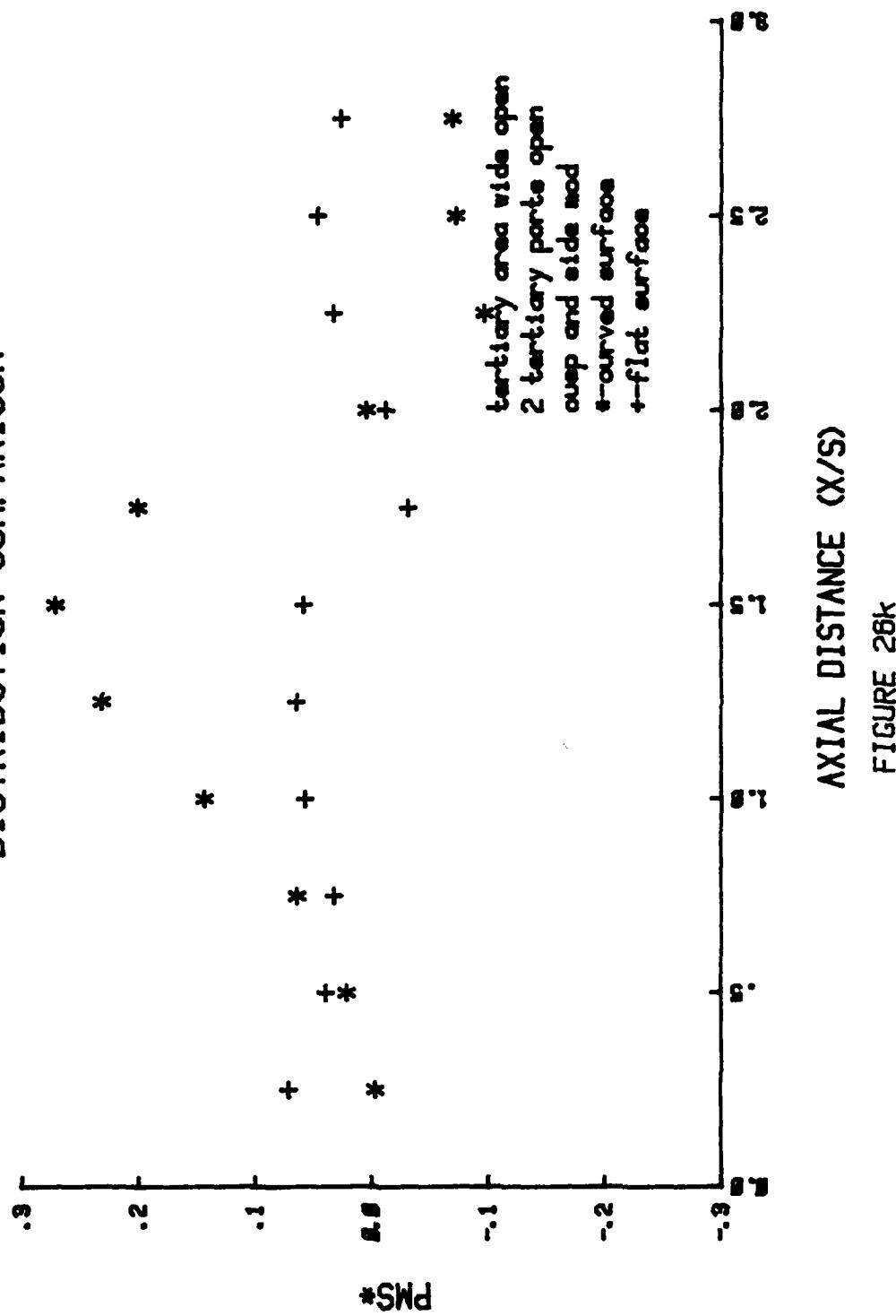
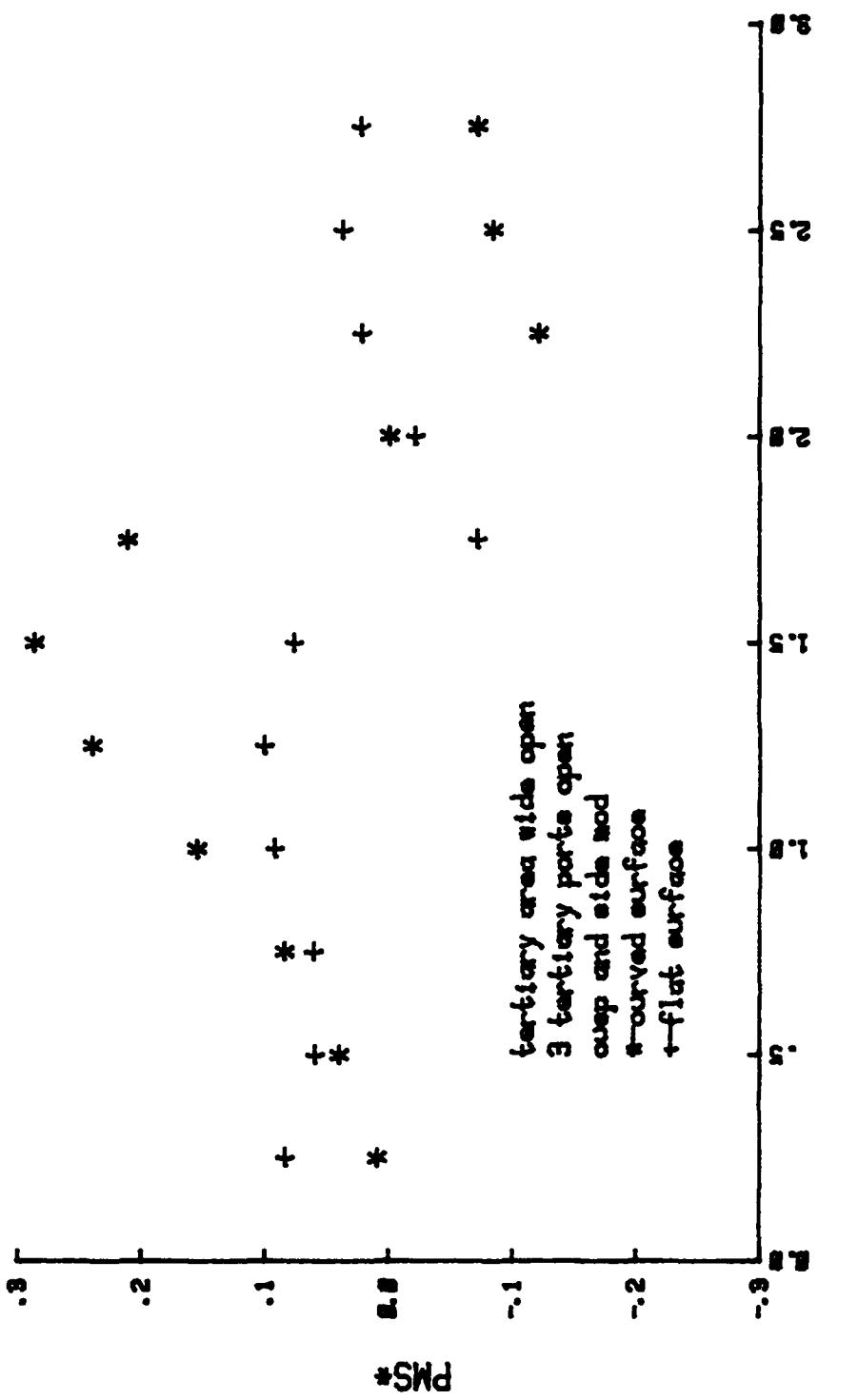


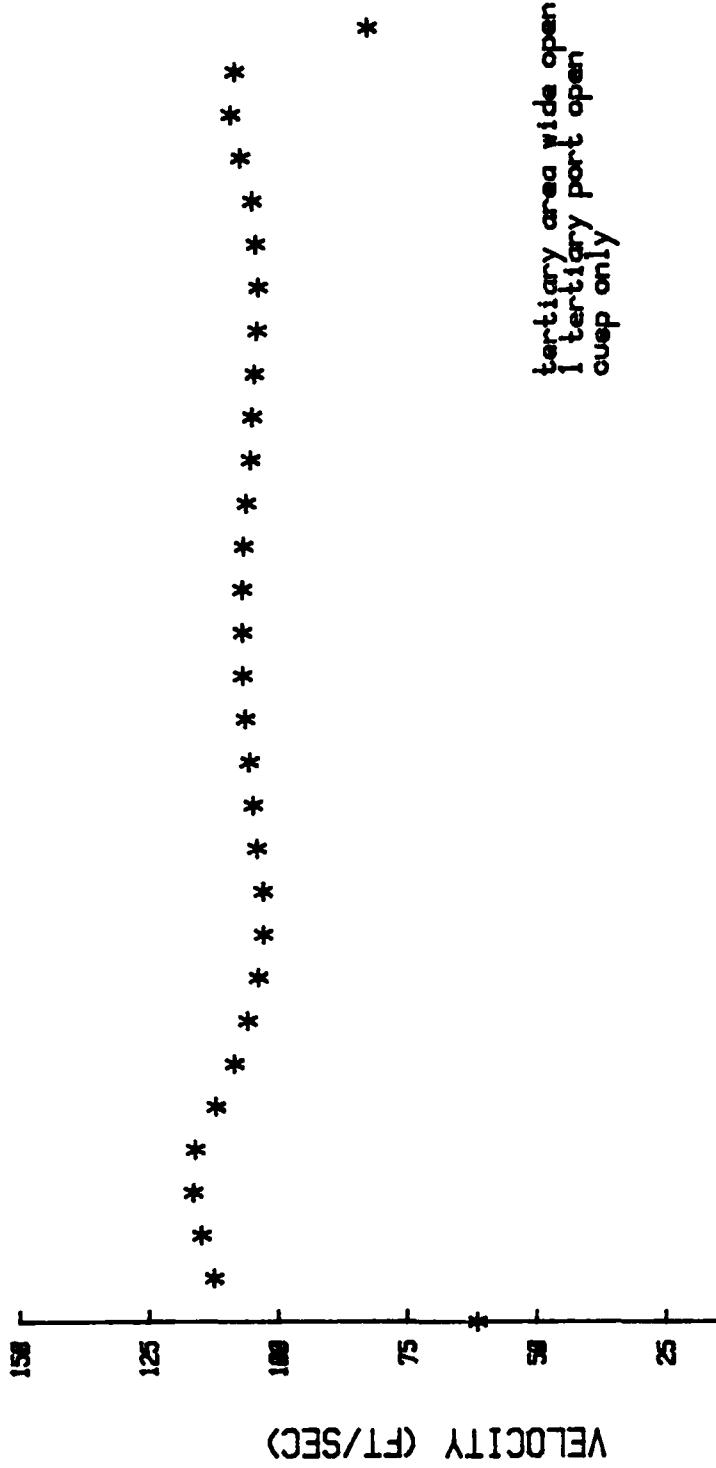
FIGURE 26K

AXIAL PRESSURE
DISTRIBUTION COMPARISON

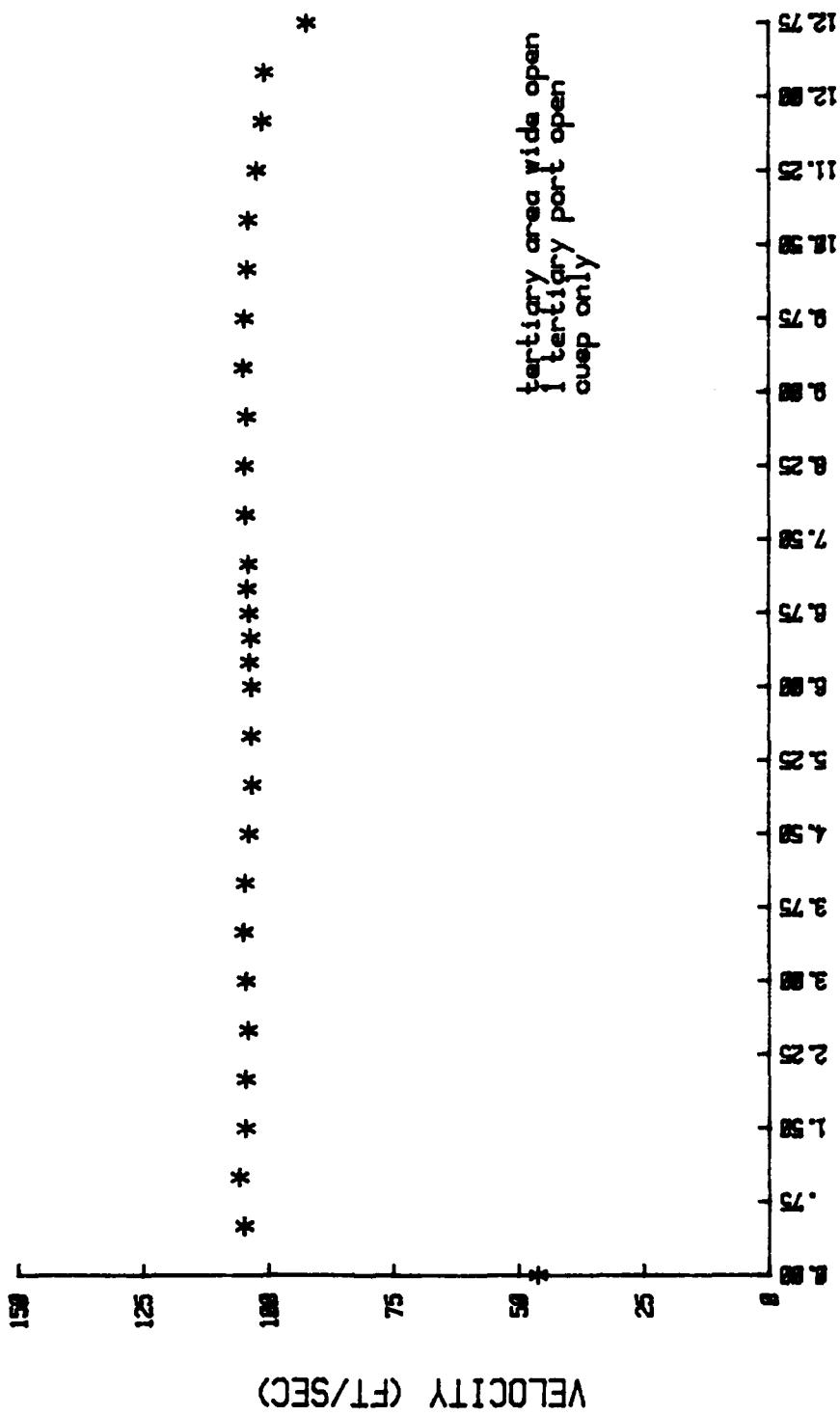


AXIAL DISTANCE (X/S)
FIGURE 261

DIAGONAL VELOCITY TRAVERSE



HORIZONTAL VELOCITY TRAVERSE



HORIZONTAL DISTANCE (IN)
FIGURE 27b

VERTICAL VELOCITY TRAVERSE

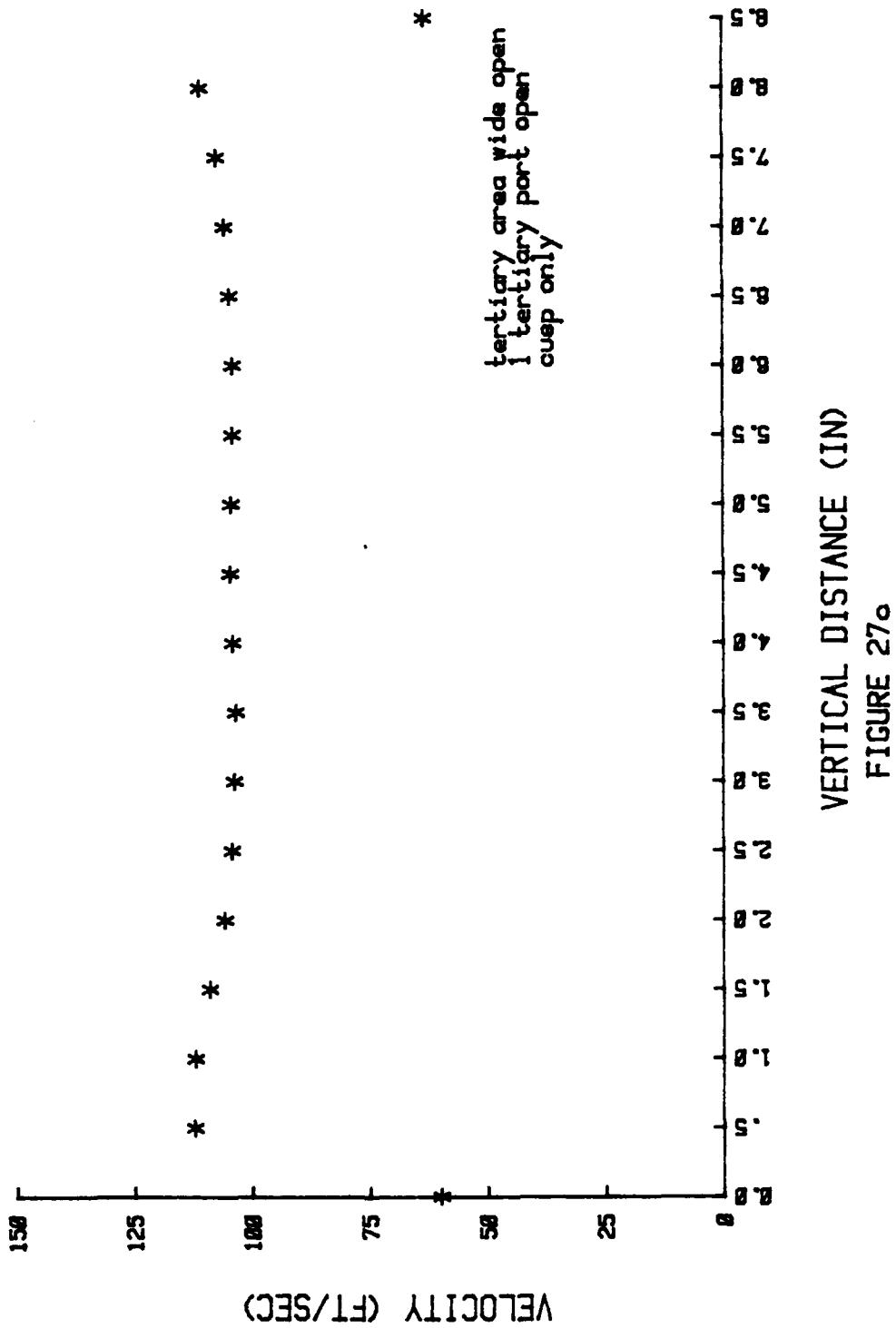
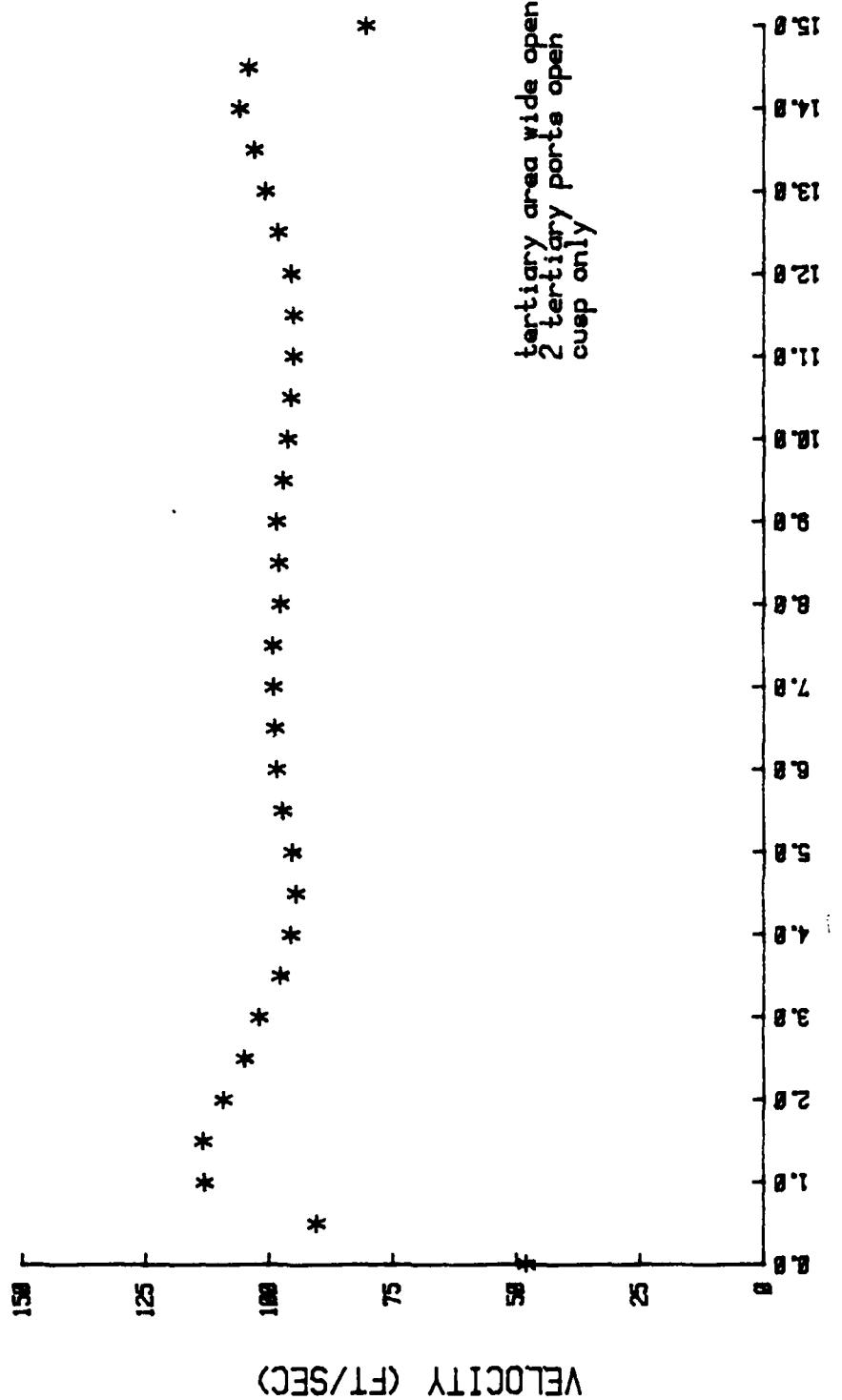


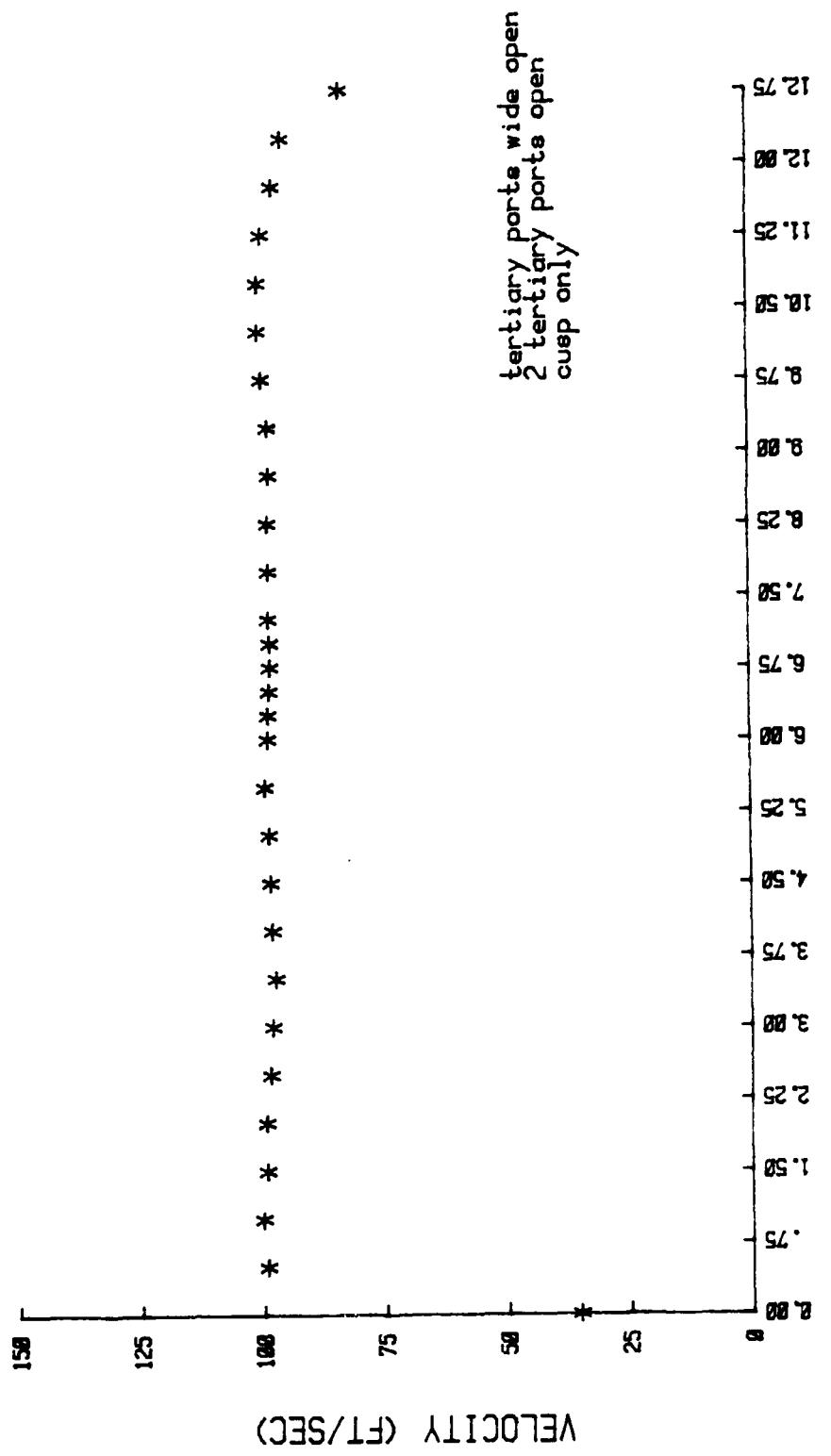
FIGURE 27o

DIAGONAL VELOCITY TRAVERSE



DIAGONAL DISTANCE (IN)
FIGURE 27d

HORIZONTAL VELOCITY TRAVERSE



HORIZONTAL DISTANCE (IN)
FIGURE 27a

VERTICAL VELOCITY TRAVERSE

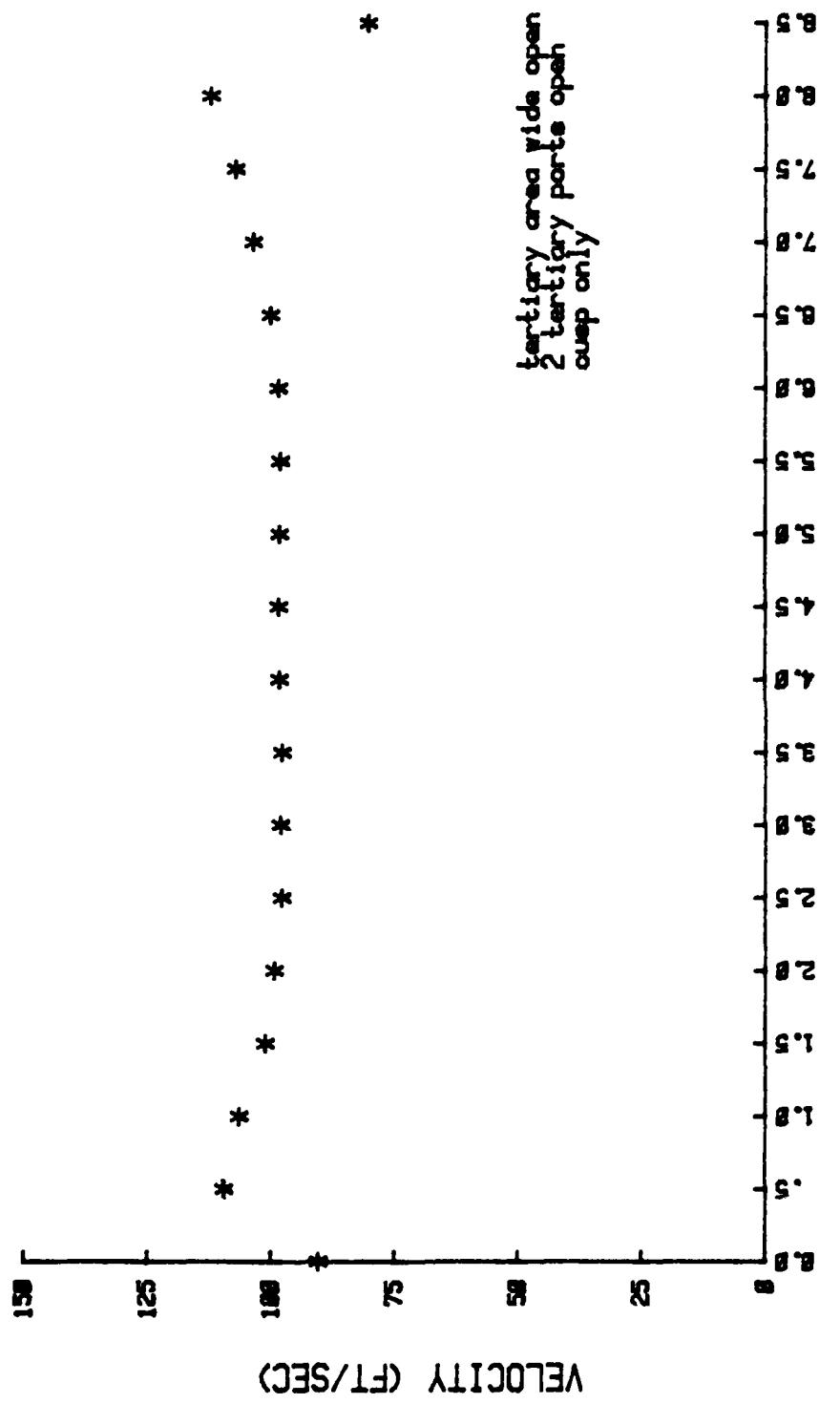
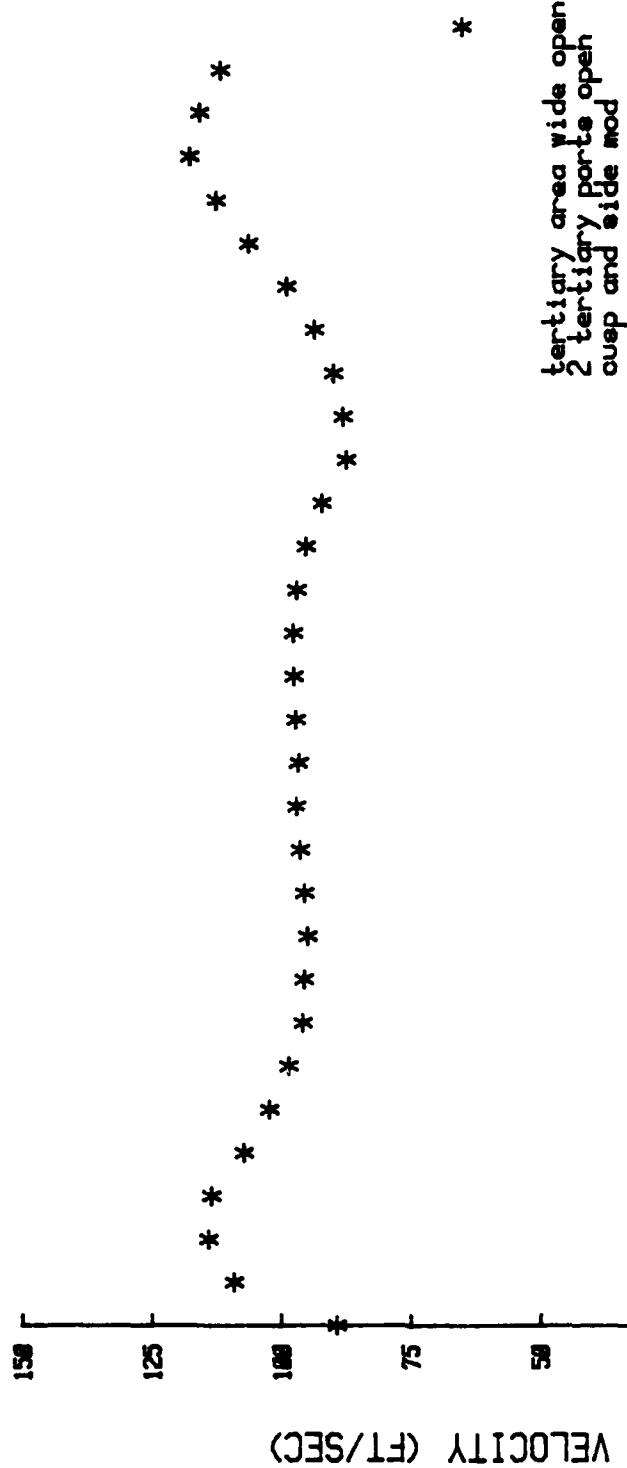


FIGURE 27f

DIAGONAL VELOCITY TRAVERSE



DIAGONAL DISTANCE (IN)
FIGURE 27q

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NAVAL POSTGRADUATE SCHOOL MONTEREY CA
FLOW CHARACTERISTICS OF A MULTIPLE NOZZLE EXHAUST GAS EDUCTOR S-ETC(U)

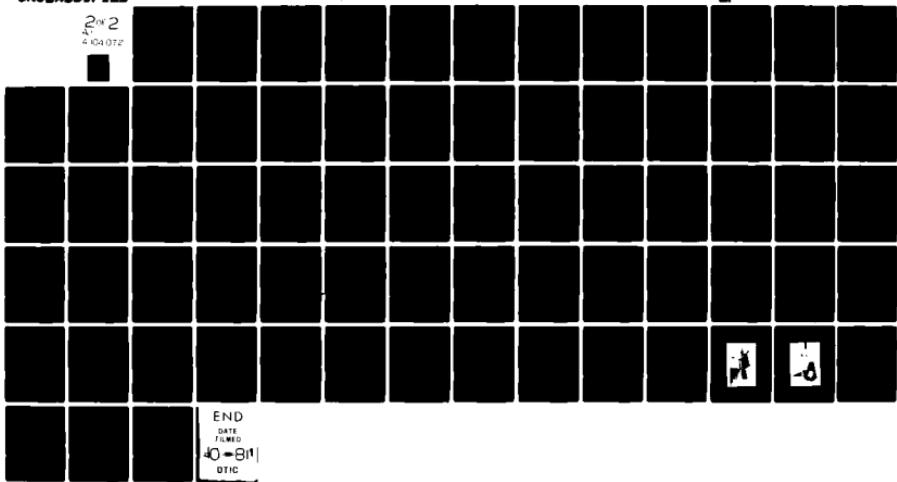
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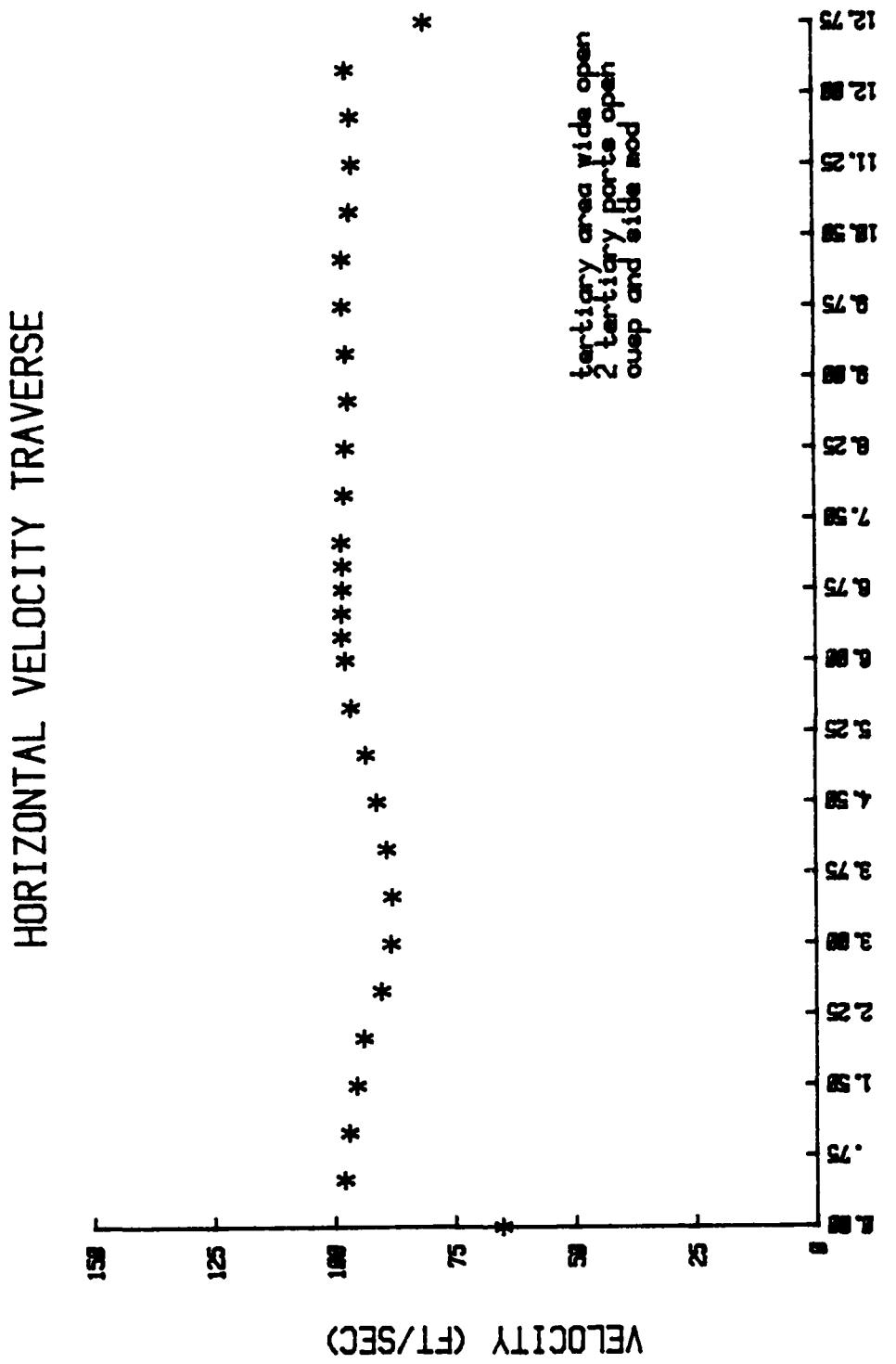
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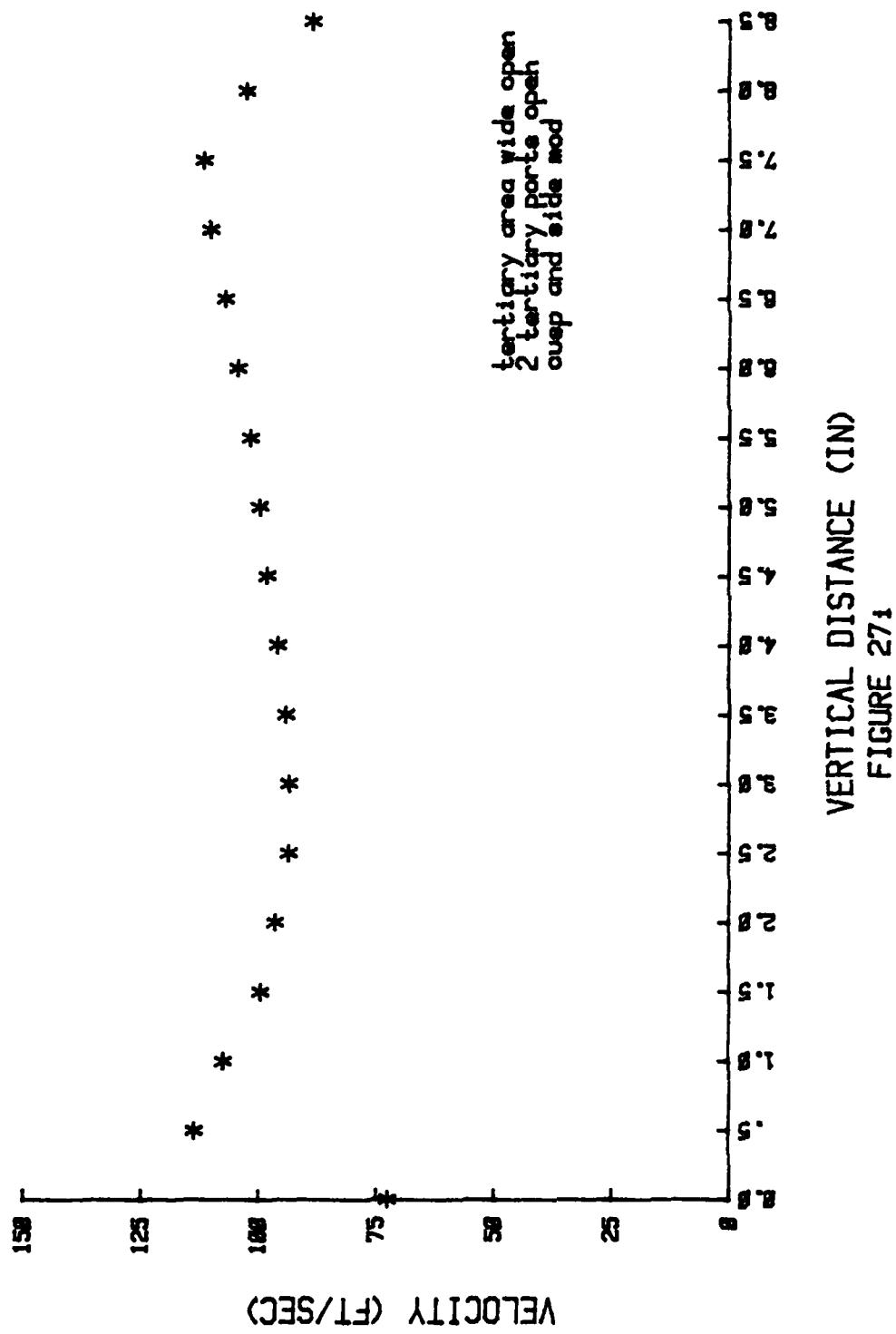
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**HORIZONTAL DISTANCE (IN)
FIGURE 27h**

VERTICAL VELOCITY TRAVERSE



VERTICAL DISTANCE (IN)
FIGURE 271

DATA TAKEN ON 27 JANUARY 1981
DATA TAKEN BY D.L. RYAN
3 TERTIARY PORTS OPEN/CUSP IN

NUMBER OF PRIMARY NOZZLES: 4 UPTAKE AREA: 197.51 SQ IN

PRIMARY NOZZLE AREA: 9.06 SQ IN AREA RATIO (AM/AP): 2.99

MIXING STACK LENGTH: 25.50 IN ORIFICE DIAMETER: 6.968 IN

MIXING STACK DIMENSION: 8.5 IN ORIFICE BETA: 0.497 IN

MIXING STACK L/S: 3.00 AMBIENT PRESSURE: 29.73 IN HG

RUN	H IN OF H ₂ O	POR DFOR	TOR	TUP DEGREES F	TAHB	PU-PA IN OF H ₂ O	PA-PS IN OF H ₂ O	PA-PT IN OF H ₂ O	SECONDARY AREA		TERTIARY AREA SQUARE INCHES
									UP	UH	
1	0.7	22.1	51.3	102.9	55.0	7.70	2.41	0.09	0.00	0.00	*****
2	0.7	21.8	50.0	103.3	55.0	8.20	1.82	0.08	0.00	0.283	*****
3	0.8	22.2	48.9	103.2	55.0	9.00	1.42	0.08	0.00	12.566	*****
4	0.7	21.8	49.9	104.0	55.0	9.65	0.62	0.01	0.01	25.133	*****
5	0.7	21.8	49.8	104.3	55.0	9.50	0.32	0.01	0.01	50.265	*****
6	0.7	21.9	50.8	104.6	55.0	9.70	0.11	0.01	0.01	100.531	*****
7	0.7	21.9	50.4	104.6	55.0	9.90	0.00	0.01	0.01	*****	*****

SECONDARY BOX

RUN	H IN	P ₄	T ₄	P ₄ /T ₄ H ₄ T ₄ .44	HP	HS	UP	UH	UPT MACH		
									LBH/SEC	LBH/SEC	FT/SEC
1	0.0000	0.2280	0.9149	0.2492	0.0000	3.7697	0.0000	214.93	71.91	72.49	0.062
2	0.0779	0.1744	0.9142	0.1907	0.0749	3.7486	0.2921	213.57	76.52	72.03	0.062
3	0.1363	0.1336	0.9144	0.1461	0.1310	3.7863	0.5160	215.49	81.05	72.68	0.062
4	0.2092	0.0787	0.9131	0.0862	0.2010	3.7491	0.7842	213.33	84.98	71.95	0.062
5	0.2613	0.0308	0.9126	0.0337	0.2510	3.7495	0.9798	213.21	86.33	71.91	0.062
6	0.3050	0.0105	0.9121	0.0116	0.2939	3.7545	1.1489	213.49	91.36	72.00	0.062
7	0.4444	0.0060	0.9121	0.0060	****	3.7558	0.6000	213.51	****	72.01	0.062

TABLE 1a

TERTIARY BOX

	W	WIS	PT ₂	TT ₂	PT ₂ /TT ₂	WT _{TT₂}	WT	WT	UE	LBM/SEC
RUN										LBH/SEC
1	37.0000	0.0000	0.9149	0.0000						3.770
2	30.0000	0.0000	0.9142	0.0000						4.041
3	35.0000	0.0000	0.9144	0.0000						4.303
4	34.0000	0.0005	0.9131	0.0005						4.533
5	33.0000	0.0005	0.9126	0.0005						4.729
6	33.0000	0.0005	0.9121	0.0005						4.903
7	33.0000	0.0005	0.9121	0.0005						

MIXING STACK PRESSURE DATA RUN										
7 CURVED SIDE DISTRIBUTION										
X/S:	0.25	0.50	0.75	1.00	1.25	1.50	1.75			
PMS(CINH20):	0.250	0.470	0.930	1.750	2.590	3.010	2.400			
PMS1:	0.024	0.045	0.094	0.168	0.248	0.289	0.230			
X/S:	2.00	2.25	2.50	2.75						
PMS(CINH20):	0.410	-3.750	-0.370	0.050						
PMS1:	0.039	-0.359	-0.035	0.005						

MIXING STACK PRESSURE DATA RUN										
7 FLAT SIDE DISTRIBUTION										
X/S:	0.25	0.50	0.75	1.00	1.25	1.50	1.75			
PMS(CINH20):	1.280	0.449	0.530	0.900	0.950	0.760	-0.280			
PMS1:	0.123	0.042	0.051	0.086	0.091	0.073	-0.027			
X/S:	2.00	2.25	2.50	2.75						
PMS(CINH20):	-0.160	0.110	0.300	0.100						
PMS1:	-0.017	0.011	0.029	0.017						

TABLE Ib

DATA TAKEN ON 28 JANUARY 1981
 DATA TAKEN BY D. L. RYAN
 2 TERTIARY PORTS OPEN/CUSP IN

NUMBER OF PRIMARY NOZZLES: 4

UPTAKE AREA: 167.51 SQ IN

PRIMARY NOZZLE AREA: 9.06 SQ IN AREA RATIO (AM/AP): 2.99

MIXING STACK LENGTH: 25.59 IN ORIFICE DIAMETER: 6.908 IN

MIXING STACK DIMENSION: 8.5 IN ORIFICE BETA: 0.497 IN

MIXING STACK L/S: 3.00

AMBIENT PRESSURE: 29.92 IN HG

N	POR	DPOR	TOR	TUPT	TAMB	PU-PA	PA-PS	PA-PT	SECONDARY AREA		TERTIARY AREA	
									IN OF H2O	SQUARE INCHES	SQUARE INCHES	
1	0.7	21.8	59.3	191.6	56.0	7.25	2.57	0.00	0.00	0.000	0.000	
2	0.7	21.7	56.8	162.5	56.0	7.75	1.98	0.00	6.283	74.511	74.511	
3	0.7	21.8	59.6	193.7	56.0	6.35	1.54	0.00	12.566	74.511	74.511	
4	0.7	21.8	50.7	104.1	56.0	8.90	0.94	0.00	25.133	74.511	74.511	
5	0.7	21.7	59.7	164.3	56.0	9.30	0.46	0.00	50.265	74.511	74.511	
6	0.7	22.6	56.9	164.5	56.0	9.60	0.14	0.00	100.531	74.511	74.511	
7	0.7	21.9	51.0	164.6	56.0	9.90	0.03	0.00	*****	74.511	74.511	

SECONDARY BOX

RUN	N	W*	P*	T*	P*/T*	W*T*/44	WP	WS	UP	UN	UUPT	UPT MACH
1	0.0000	0.2474	0.9188	0.2693	0.0000	3.7596	0.0000	212.58	71.12	71.69	0.062	
2	0.0214	0.1316	0.9173	0.2039	0.0764	3.7491	0.3653	212.02	76.21	71.51	0.062	
3	0.1433	0.1480	0.9153	0.1617	0.1378	3.7587	0.5365	212.79	80.49	71.76	0.062	
4	0.2239	0.0505	0.9147	0.0989	0.2153	3.7582	0.8415	212.59	85.66	71.70	0.062	
5	0.2528	0.0388	0.9144	0.0424	0.2815	3.7497	1.0973	211.98	89.86	71.47	0.061	
6	0.3441	0.0134	0.9149	0.0147	0.3308	3.7746	1.2990	213.25	93.78	71.92	0.062	
7	*****	0.0029	0.9135	0.0032	*****	3.7657	9.3958	212.86	*****	71.77	0.062	

TABLE IIIa

TERTIARY BOX

RUN	N HTS PHS TTS PTS/TTW WT/TTW						WM	WT	UE
	LBM/SEC FT/SEC								
1	0.0000	0.9188	0.0000	*****	3.760	*****			
2	0.0000	0.9173	0.0000	*****	4.054	*****			
3	0.0000	0.9153	0.0000	*****	4.297	*****			
4	0.0000	0.9147	0.0000	*****	4.500	*****			
5	0.0000	0.9144	0.0000	*****	4.848	*****			
6	0.0000	0.9140	0.0000	*****	5.074	*****			
7	0.0000	0.9135	0.0000	*****	*****	*****			

MIXING STACK PRESSURE DATA: RUN 7 CURVED SIDE DISTRIBUTION

X/S:	0.25	0.59	0.75	1.00	1.25	1.50	1.75
PHS(CINH20):	-0.010	0.235	0.670	1.460	2.250	2.590	2.660
PHS1:	-0.001	0.023	0.064	0.140	0.216	0.249	0.198
X/S:	2.00	2.25	2.50	2.75			
PHS(CINH20):	0.520	-3.620	-0.390	0.040			
PHS1:	0.050	-0.348	-0.037	0.004			

MIXING STACK PRESSURE DATA: RUN 7 FLAT SIDE DISTRIBUTION

X/S:	0.25	0.50	0.75	1.00	1.25	1.50	1.75
PHS(CINH20):	1.040	0.220	0.240	0.550	0.640	0.610	0.010
PHS1:	0.100	0.021	0.023	0.053	0.061	0.059	0.001
X/S:	2.00	2.25	2.50	2.75			
PHS(CINH20):	-0.960	0.180	0.330	0.180			
PHS1:	-0.006	0.017	0.032	0.017			

TABLE IIIb

VERTICAL VELOCITY TRAVERSE						
POSIT(IN)	0.00	0.50	1.00	1.50	2.00	2.50
P(IN H20)	1.84	2.78	2.55	2.38	2.22	2.15
V(FT/SEC)	89.47	168.38	165.33	166.63	96.28	96.71
POSIT(IN)	3.58	4.69	4.58	3.88	5.58	6.08
P(IN H20)	2.15	2.17	2.18	2.17	2.16	2.18
V(FT/SEC)	96.71	97.16	97.39	97.16	96.94	97.39
POSIT(IN)	7.00	7.58	8.00	8.50		
P(IN H20)	2.41	2.58	2.82	1.45		
V(FT/SEC)	102.48	105.95	110.76	79.43		

HORIZONTAL VELOCITY TRAVERSE						
POSIT(IN)	0.00	0.50	1.00	1.50	2.00	2.50
P(IN H20)	0.28	2.22	2.26	2.22	2.22	2.18
V(FT/SEC)	34.90	98.28	99.16	98.28	98.28	96.94
POSIT(IN)	3.58	4.00	4.58	5.00	5.58	6.08
P(IN H20)	2.13	2.16	2.17	2.18	2.22	2.18
V(FT/SEC)	96.26	96.94	97.16	97.39	98.28	97.61
POSIT(IN)	6.58	6.75	7.00	7.25	7.75	8.25
P(IN H20)	2.17	2.16	2.16	2.17	2.17	2.16
V(FT/SEC)	97.16	96.94	96.94	97.16	97.16	96.94
POSIT(IN)	9.25	9.75	10.25	10.75	11.25	12.25
P(IN H20)	2.17	2.22	2.25	2.25	2.21	2.03
V(FT/SEC)	97.16	98.28	98.94	98.94	98.94	93.98
POSIT(IN)	12.75					
P(IN H20)	1.55					
V(FT/SEC)	82.12					

DIAGONAL VELOCITY TRAVERSE						
POSIT(IN)	0.69	0.58	1.00	1.50	2.00	2.50
P(IN H20)	0.52	1.84	2.88	2.89	2.69	2.48
V(FT/SEC)	47.56	89.47	111.94	112.13	106.18	103.87
POSIT(IN)	3.58	4.66	4.58	5.00	5.58	6.08
P(IN H20)	2.15	2.05	2.01	2.05	2.13	2.18
V(FT/SEC)	96.71	94.67	93.51	94.44	96.26	97.39
POSIT(IN)	7.00	7.50	8.00	8.50	9.00	9.58
P(IN H20)	2.21	2.21	2.15	2.16	2.18	2.08
V(FT/SEC)	98.06	98.06	96.71	96.94	97.39	96.84
POSIT(IN)	10.50	11.00	11.50	12.00	12.50	13.00
P(IN H20)	2.05	2.03	2.03	2.05	2.16	2.38
V(FT/SEC)	94.44	93.98	93.98	94.44	96.94	95.68
POSIT(IN)	14.00	14.60	14.50	15.00		
P(IN H20)	2.52	2.43	1.45			
V(FT/SEC)	104.71	102.82	79.43			

TABLE IIa

DATA TAKEN ON 29 JANUARY 1961
 DATA TAKEN BY D. L. RYAN
 1 TERTIARY FORT OPEN/CUSP IN

NUMBER OF PRIMARY NOZZLES - 4 UPTAKE AREA - 187.51 SQ IN

PRIMARY NOZZLE AREA - 9.06 SQ IN AREA RATIO (AM/AP) - 2.99

MIXING STACK LENGTH - 25.59 IN ORIFICE DIAMETER - 6.968 IN

MIXING STACK DIMENSION - 8.5 IN ORIFICE BETA - 0.497 IN

MIXING STACK L/S - 3.00 AMBIENT PRESSURE - 29.94 IN HG

RUN	H	POR	DPOR	TOR	TUPT	TAMB	PU-PA	PA-PS	PA-PT	SECONDARY AREA		TERTIARY AREA	
										IN OF H ₂ O	SQUARE INCHES	SQUARE INCHES	SQUARE INCHES
1	0.7	21.8	59.7	105.0	65.0	7.10	2.73	0.00	0.00	0.00	0.000	0.000	0.000
2	0.7	21.8	59.2	104.9	65.8	7.70	2.13	0.00	0.00	0.00	6.283	6.283	6.283
3	0.7	21.9	59.1	104.9	65.9	6.15	1.65	0.00	0.00	0.00	25.366	25.366	25.366
4	0.7	21.8	59.3	104.9	65.9	6.75	1.03	0.00	0.00	0.00	50.265	50.265	50.265
5	0.7	22.0	59.1	104.7	65.8	9.45	0.50	0.00	0.00	0.00	100.531	100.531	100.531
6	0.7	21.8	59.4	104.7	65.8	9.75	0.18	0.00	0.00	0.00	11111111	11111111	11111111
7	0.9	21.8	49.8	104.7	65.8	9.85	0.03	0.00	0.00	0.00	11111111	11111111	11111111

SECONDARY BOX

RUN	H	WT	P*	T*	P* ² /T ² WT ^{1.44}	WP	HS	UP	UN	UWPT	UWPT	UPT MACH	FT/SEC	
1	0.0098	0.26442	0.52292	0.2843	0.0060	3.7594	0.0000	213.80	71.53	72.10	0.062	0.062	0.062	
2	0.0835	0.2066	0.9293	0.2223	0.0808	3.7612	0.3141	213.55	76.96	72.02	0.062	0.062	0.062	
3	0.1466	0.1597	0.9293	0.1718	0.1420	3.7702	0.5528	213.80	81.24	72.11	0.062	0.062	0.062	
4	0.2323	0.1095	0.9293	0.1081	0.2249	3.7608	0.8736	212.95	86.59	71.82	0.062	0.062	0.062	
5	0.3221	0.0485	0.9297	0.0521	0.3120	3.7787	1.2173	213.61	92.84	72.64	0.062	0.062	0.062	
6	0.3385	0.0176	0.9297	0.0190	0.3762	3.7605	1.4608	212.41	96.72	71.64	0.062	0.062	0.062	
7	444444	0.0023	0.5297	0.0032	444444	3.7626	9.3186	212.41	444444	71.64				

TABLE IIIa

TERTIARY BOX

RUN	W	WT	PT*	TT*	PT*/TT*	WT/TT^2	UE	HM	WT	UE
								LB/M SEC	LB/M SEC	FT/SEC
1	*****	0.0000	0.9292	0.0000	*****	*****	3.759	*****	*****	*****
2	*****	0.0000	0.9293	0.0000	*****	*****	4.075	*****	*****	*****
3	*****	0.0000	0.9293	0.0000	*****	*****	4.323	*****	*****	*****
4	*****	0.0000	0.9293	0.0000	*****	*****	4.634	*****	*****	*****
5	*****	0.0000	0.9297	0.0000	*****	*****	4.996	*****	*****	*****
6	*****	0.0000	0.9297	0.0000	*****	*****	5.221	*****	*****	*****
7	*****	0.0000	0.9297	0.0000	*****	*****	*****	*****	*****	*****

MIXING STACK PRESSURE DATA: RUN 7 CURVED SIDE DISTRIBUTION

X/S:	0.25	0.50	0.75	1.00	1.25	1.50	1.75
PMS(CINH20):	-0.260	-0.050	0.380	1.180	1.910	2.210	1.680
PMS: -0.025	-0.005	0.037	0.116	0.187	0.217	0.164	
X/S:	2.00	2.25	2.50	2.75			
PMS(CINH20):	0.650	-3.420	-0.390	0.050			
PMS: 0.064	-0.335	-0.038	0.005				

MIXING STACK PRESSURE DATA: RUN 7 FLAT SIDE DISTRIBUTION

X/S:	0.25	0.50	0.75	1.00	1.25	1.50	1.75
PMS(CINH20):	0.780	-0.690	-0.080	0.670	0.090	0.330	0.290
PMS: 0.076	-0.009	-0.008	0.007	0.009	0.032	0.020	
X/S:	2.00	2.25	2.50	2.75			
PMS(CINH20):	0.080	0.240	0.360	0.230			
PMS: 0.008	0.024	0.035	0.023				

TABLE IIIb

VERTICAL VELOCITY TRAVERSE						
POSIT(IN)	0.00	0.50	1.00	1.50	2.00	2.50
P(IN H20)	0.81	2.84	2.83	2.68	2.52	2.45
V(FT/SEC)	59.86	112.68	111.89	108.68	105.58	104.10
POSIT(IN)	3.50	4.00	4.50	5.00	5.50	6.00
P(IN H20)	2.41	2.44	2.46	2.45	2.44	2.43
V(FT/SEC)	103.25	103.89	104.32	104.16	103.89	104.32
POSIT(IN)	7.00	7.50	8.00	8.50		
P(IN H20)	2.51	2.59	2.76	2.98		
V(FT/SEC)	105.37	107.04	110.49	113.10		

HORIZONTAL VELOCITY TRAVERSE						
POSIT(IN)	0.00	0.50	1.00	1.50	2.00	2.50
P(IN H20)	0.48	2.48	2.52	2.46	2.44	2.46
V(FT/SEC)	46.08	104.74	105.58	104.32	103.89	104.32
POSIT(IN)	3.50	4.00	4.50	5.00	5.50	6.00
P(IN H20)	2.48	2.47	2.43	2.40	2.41	2.43
V(FT/SEC)	104.74	104.53	103.68	103.04	103.25	103.68
POSIT(IN)	6.50	6.75	7.00	7.25	7.75	8.25
P(IN H20)	2.42	2.43	2.45	2.44	2.46	2.47
V(FT/SEC)	103.46	103.68	104.10	103.89	104.32	104.53
POSIT(IN)	9.25	9.75	10.25	10.75	11.25	11.75
P(IN H20)	2.48	2.47	2.44	2.43	2.35	2.30
V(FT/SEC)	104.74	104.53	103.89	103.68	101.96	100.87
POSIT(IN)	12.75					
P(IN H20)	1.91					
V(FT/SEC)	91.92					

DIAGONAL VELOCITY TRAVERSE						
POSIT(IN)	0.00	0.50	1.00	1.50	2.00	2.50
P(IN H20)	0.85	2.85	2.97	3.05	3.04	3.00
V(FT/SEC)	61.32	112.28	114.62	116.15	115.96	111.89
POSIT(IN)	3.50	4.00	4.50	5.00	5.50	6.00
P(IN H20)	2.52	2.42	2.38	2.38	2.43	2.46
V(FT/SEC)	105.58	103.46	102.61	102.61	103.68	104.32
POSIT(IN)	7.09	7.50	8.00	8.50	9.00	9.50
P(IN H20)	2.54	2.56	2.56	2.56	2.55	2.48
V(FT/SEC)	106.00	106.42	106.42	106.42	105.53	104.74
POSIT(IN)	10.50	11.00	11.50	12.00	12.50	13.00
P(IN H20)	2.46	2.44	2.42	2.41	2.43	2.37
V(FT/SEC)	104.32	103.89	103.46	103.25	103.68	104.32
POSIT(IN)	14.00	14.50	15.00			
P(IN H20)	2.66	2.62	1.52			
V(FT/SEC)	106.47	107.66	82.60			

TABLE IIIa

DATA TAKEN ON 1 FEBRUARY 1981
 DATA TAKEN BY O. L. RYAN
 1 TERTIARY PORT OPEN/CUSP IN
 NUMBER OF PRIMARY NOZZLES: 4 UPTAKE AREA: 107.51 SQ IN
 PRIMARY NOZZLE AREA: 9.06 SQ IN AREA RATIO (AN/AP): 2.99
 MIXING STACK LENGTH: 25.50 IN ORIFICE DIAMETER: 6.908 IN
 MIXING STACK DIMENSION: 8.5 IN ORIFICE BETA: 0.497 IN
 MIXING STACK L/S: 3.00 AMBIENT PRESSURE: 30.34 IN HG

N	POR	DPOR	TOR	TUPT	TAMB	PU-PA	PA-PS	PA-PT	SECONDARY AREA		TERTIARY AREA	
									IN OF H ₂ O	SQUARE INCHES	SQUARE INCHES	SQUARE INCHES
1	0.7	21.9	43.5	93.9	54.0	6.90	2.76	0.50	0.000	0.000	0.000	0.000
2	0.7	21.9	43.4	95.3	54.0	7.20	2.76	0.11	0.000	0.000	0.283	0.283
3	0.7	21.9	43.2	96.0	54.0	7.20	2.77	0.03	0.000	0.000	1.2566	1.2566
4	0.7	21.9	43.4	96.8	54.0	7.20	2.77	0.01	0.000	0.000	25.133	25.133
5	0.7	21.9	42.9	97.0	54.0	7.20	2.77	0.01	0.000	0.000	37.699	37.699
6	0.7	21.9	43.3	97.0	54.0	7.20	2.80	0.01	0.000	0.000	100.531	100.531
7	0.7	21.9	43.4	96.8	54.0	7.20	2.78	0.00	0.000	0.000	0.000	0.000

SECONDARY BOX

RUN	N	W*	P4	T*	P*/T* W*T* .44	NP	WS	UP	UN	UOPT	UPT MACH	LBM/SEC LBM/SEC FT/SEC FT/SEC
1	0.0000	0.2670	0.9279	0.2878	0.0000	3.8203	0.0000	210.18	70.32	70.88	0.061	0.061
2	0.0000	0.2655	0.9256	0.2870	0.0000	3.8207	0.0000	210.73	70.50	71.97	0.062	0.062
3	0.0000	0.2658	0.9244	0.2875	0.0000	3.8214	0.0000	211.04	70.61	71.18	0.062	0.062
4	0.0000	0.2651	0.9231	0.2872	0.0000	3.8207	0.0000	211.31	70.70	71.26	0.062	0.062
5	0.0000	0.2647	0.9228	0.2868	0.0000	3.8226	0.0000	211.49	70.76	71.33	0.062	0.062
6	0.0000	0.2677	0.9228	0.2901	0.0000	3.8211	0.0000	211.42	70.73	71.30	0.062	0.062
7	0.0000	0.2661	0.9231	0.2882	0.0000	3.8207	0.0000	211.31	70.70	71.27	0.062	0.062

TABLE IVa

TERTIARY BOX

RUN	H	W ₁₂	P ₁₂	T ₁₂	P ₁₂ /T ₁₂	W ₁₂ T ₁₂	44	W ₁₂	W ₁₂	UE	LBM/SEC	LBM/SEC	FT/SEC
1	0.0000	0.0484	0.9279	0.0521		0.0000		3.829	0.000	70.32			
2	0.0190	0.0106	0.9256	0.0114		0.0184		3.821	0.073	71.84			
3	0.0198	0.0029	0.9244	0.0051		0.0192		3.821	0.076	72.01			
4	0.0229	0.0010	0.9231	0.0010		0.0221		3.821	0.088	72.32			
5	0.0344	0.0010	0.9228	0.0010		0.0332		3.823	0.131	73.19			
6	0.0648	0.0005	0.9228	0.0005		0.0626		3.821	0.248	75.32			
7	333333	0.0000	0.9231	0.0000		*****		3.821	*****	*****			

MIXING STACK PRESSURE DATA: RUN 7 CURVED SIDE DISTRIBUTION

X/S: 0.25	0.50	0.75	1.00	1.25	1.50	1.75
PMS(CINH20): -1.370	-0.760	-0.120	0.560	1.330	1.820	1.730
PMS(FUG): -0.131	-0.073	-0.012	0.054	0.127	0.174	0.166
X/S: 2.00	2.25	2.50	2.75			
PMS(CINH20): 0.659	-1.900	-0.140	0.940			
PMS: 0.062	-0.132	-0.013	0.004			

MIXING STACK PRESSURE DATA: RUN 7 FLAT SIDE DISTRIBUTION

X/S: 0.25	0.50	0.75	1.00	1.25	1.50	1.75
PMS(CINH20): -0.260	-0.810	-0.430	-0.040	0.280	0.520	0.210
PMS: -0.025	-0.078	-0.046	-0.004	0.027	0.056	0.020
X/S: 2.00	2.25	2.50	2.75			
PMS(CINH20): 0.140	0.130	0.230	0.150			
PMS: 0.013	0.012	0.022	0.014			

VERTICAL VELOCITY TRAVERSE					
POSITION(IND):	0.00	0.50	1.00	1.50	2.00
P(IN H20):	0.81	1.49	1.46	1.39	1.32
V(FT/SEC):	58.84	79.80	78.99	77.97	75.11
POSITION(IND):	3.50	4.00	4.50	5.00	5.50
P(IN H20):	1.35	1.36	1.36	1.35	1.33
V(FT/SEC):	75.96	76.24	76.24	75.96	75.39
POSITION(IND):	7.00	7.50	8.00	8.50	9.00
P(IN H20):	1.30	1.35	1.48	1.49	1.50
V(FT/SEC):	74.54	75.96	79.53	45.76	

HORIZONTAL VELOCITY TRAVERSE					
POSITION(IND):	0.00	0.50	1.00	1.50	2.00
P(IN H20):	0.61	1.32	1.26	1.23	1.22
V(FT/SEC):	51.06	75.11	73.38	72.59	72.21
POSITION(IND):	3.50	4.00	4.50	5.00	5.50
P(IN H20):	1.30	1.32	1.34	1.34	1.35
V(FT/SEC):	74.54	75.11	75.68	75.68	75.68
POSITION(IND):	6.50	6.75	7.00	7.25	7.75
P(IN H20):	1.35	1.34	1.36	1.36	1.38
V(FT/SEC):	75.96	75.68	76.24	76.24	76.80
POSITION(IND):	9.25	9.75	10.25	10.75	11.75
P(IN H20):	1.42	1.43	1.41	1.38	1.35
V(FT/SEC):	77.96	78.19	77.63	76.86	75.96
POSITION(IND):	12.75				
P(IN H20):	1.16				
V(FT/SEC):	68.56				

DIAGONAL VELOCITY TRAVERSE					
POSITION(IND):	0.00	0.50	1.00	1.50	2.00
P(IN H20):	0.33	1.83	1.83	1.72	1.62
V(FT/SEC):	37.55	88.44	83.44	85.74	83.21
POSITION(IND):	3.50	4.00	4.50	5.00	5.50
P(IN H20):	1.33	1.30	1.33	1.35	1.35
V(FT/SEC):	75.39	74.54	75.39	75.96	75.96
POSITION(IND):	7.00	7.50	8.00	8.50	9.00
P(IN H20):	1.37	1.36	1.35	1.35	1.36
V(FT/SEC):	76.52	76.24	75.96	75.96	76.24
POSITION(IND):	10.50	11.00	11.50	12.00	12.50
P(IN H20):	1.33	1.38	1.29	1.25	1.22
V(FT/SEC):	75.39	74.54	74.25	73.69	72.21
POSITION(IND):	14.00	14.50	15.00	14.4	9.90
P(IN H20):	1.35	1.44	1.44	1.44	62.02
V(FT/SEC):	75.96	78.45			

TABLE IVc

DATA TAKEN ON 1 FEBRUARY 1981
 DATA TAKEN BY D. L. RYAN
 1 TERTIARY PORT OPEN/CUSP IN

NUMBER OF PRIMARY NOZZLES: 4 UPTAKE AREA: 107.51 SQ IN

PRIMARY NOZZLE AREA: 9.06 SQ IN AREA RATIO (A_M/A_P): 2.99

MIXING STACK LENGTH: 25.50 IN ORIFICE DIAMETER: 6.908 IN

MIXING STACK DIMENSION: 0.5 IN ORIFICE BETA: 0.497 IN

MIXING STACK L/S: 3.66 AMBIENT PRESSURE: 30.33 IN HG

RUN	H IN OF H ₂ O	POR	OPOR	TOR	TUPT	TMB	PU-PA	PA-PS	PA-PT	SECONDARY AREA		TERTIARY AREA
										DEGREES	F	
										IN OF H ₂ O	SQUARE INCHES	SQUARE INCHES
1	0.7	21.9	42.9	97.4	58.0	9.25	0.57	0.58	0.58	37.699	0.600	
2	0.7	21.9	43.4	97.3	58.0	9.25	0.53	0.14	0.14	37.699	6.283	
3	0.7	21.9	43.7	97.4	58.0	9.39	0.51	0.05	0.05	37.699	12.566	
4	0.7	21.9	43.3	97.4	58.0	9.39	0.51	0.02	0.02	37.699	25.133	
5	0.7	21.9	42.8	97.4	58.0	9.35	0.48	0.01	0.01	37.699	37.699	
6	0.7	21.9	43.2	97.5	58.0	9.39	0.48	0.00	0.00	37.699	100.531	
7	0.7	21.9	42.6	97.4	58.0	9.36	0.47	0.00	0.00	37.699	*****	

SECONDARY BOX

RUN	H IN	W ₁ IN	P ₁ IN	T ₁ IN	P ₂ /T ₂ W ₂ T ₂ IN	W ₂ IN	W ₃ IN	U _P IN	U _S IN	U _M IN	U _{UP} IN	U _{PT} IN	U _H IN
1	0.2584	0.0554	0.9293	0.0596	0.2502	3.6232	0.9881	210.47	87.31	78.98	0.061		
2	0.2493	0.0516	0.9294	0.0555	0.2414	3.6213	0.9528	210.31	86.65	70.93	0.061		
3	0.2446	0.0497	0.9293	0.0534	0.2369	3.6202	0.9346	210.23	86.33	70.92	0.061		
4	0.2446	0.0496	0.9289	0.0534	0.2367	3.6217	0.9346	210.44	86.38	70.97	0.061		
5	0.2371	0.0467	0.9293	0.0502	0.2296	3.6236	0.9067	210.45	85.91	70.98	0.061		
6	0.2372	0.0467	0.9291	0.0502	0.2297	3.6221	0.9067	210.49	85.89	70.96	0.061		
7	0.2343	0.0456	0.9293	0.0491	0.2276	3.6267	0.9972	210.61	85.89	71.03	0.061		

TABLE Va

TERTIARY BOX

RUN	TERTIARY BOX						LBM/SEC		
	W	WT	PT*	TT*	PT*/TT*	WT*TT^4.4	W	WT	UE
1	0.0008	0.0564	0.9293	0.0607	0.0000	4.811	0.000	87.31	
2	0.0214	0.0136	0.9234	0.0147	0.0287	4.774	0.082	86.13	
3	0.0255	0.0049	0.9293	0.0052	0.0247	4.755	0.098	86.19	
4	0.0323	0.0019	0.9289	0.0021	0.0313	4.756	0.123	86.62	
5	0.0342	0.0010	0.9293	0.0016	0.0331	4.730	0.131	86.28	
6	0.0000	0.0000	0.9291	0.0000	0.0000	4.729	0.000	85.09	
7	222222	0.0000	0.9293	0.0000	222222	4.724	222222	222222	

MIXING STACK PRESSURE DATA: RUN 7 CURVED SIDE DISTRIBUTION

X/S:	0.25	0.50	0.75	1.00	1.25	1.50	1.75
PMS(CINH20)	-0.520	-0.230	0.200	0.930	1.760	2.090	1.920
PMS4:	-0.050	-0.022	0.019	0.090	0.171	0.263	0.186
X/S:	2.00	2.25	2.50	2.75			
PMS(CINH20)	0.840	-3.050	-0.260	0.060			
PMS4:	0.082	-0.296	-0.025	0.006			

MIXING STACK PRESSURE DATA: RUN 7 FLAT SIDE DISTRIBUTION

X/S:	0.25	0.50	0.75	1.00	1.25	1.50	1.75
PMS(CINH20)	0.780	-0.220	-0.240	0.030	0.230	0.400	0.110
PMS4:	0.076	-0.021	-0.023	0.003	0.022	0.039	0.011
X/S:	2.00	2.25	2.50	2.75			
PMS(CINH20)	0.150	0.300	0.450	0.270			
PMS4:	0.015	0.029	0.044	0.026			

DATA TAKEN ON 2 FEBRUARY 1981
DATA TAKEN BY D. L. RYAN
1 TERTIARY PORT OPEN/CUSP IN

NUMBER OF PRIMARY NOZZLES: 1 UPTAKE AREA: 1.07 .51 .59 .19

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2	0.7	22.0	52.0	104.5	66.0	9.55	0.01	0
3	0.7	22.1	51.5	104.6	66.0	9.50	0.01	0
4	0.7	22.1	50.6	104.8	66.0	9.90	0.01	0
5	0.7	21.8	52.5	105.2	66.0	9.60	0.01	0
6	0.7	21.8	59.2	105.0	66.0	9.60	0.01	0
7	0.7	21.8	51.7	105.1	66.0	9.80	0.00	0

SECONDARY BOX

TABLE VIa

TERTIARY BOX							LBIN/SEC			
RUN	W	WTS	PTs	TTs	PTs/TTs	WTTs	.44	HM	Wt	UE
1	0.0000	0.0608	0.9323	0.0652	0.0600	*****	0.060	*****	0.060	*****
2	0.0205	0.9127	0.9318	0.0136	0.0199	*****	0.078	*****	0.078	*****
3	0.0227	0.9039	0.9316	0.0042	0.0220	*****	0.086	*****	0.086	*****
4	0.0278	0.0015	0.9313	0.0016	0.0270	*****	0.106	*****	0.106	*****
5	0.0324	0.0095	0.9306	0.0095	0.0314	*****	0.122	*****	0.122	*****
6	0.0000	0.0000	0.9309	0.0000	0.0000	*****	0.000	*****	0.000	*****
7	*****	0.0000	0.9309	0.0000	*****	*****	*****	*****	*****	*****

MIXING STACK PRESSURE DATA, RUN 7 CURVED SIDE DISTRIBUTION

X/S:	0.25	0.39	0.75	1.00	1.25	1.50	1.75
PHS<INH20>:	-0.240	-0.330	0.400	1.080	1.250	2.230	1.790
PHSt:	-0.024	-0.003	0.039	0.106	0.192	0.219	0.176
X/S:	2.00	2.25	2.50	2.75			
PHS<INH20>:	0.570	-3.540	-6.390	0.050			
PHSt:	0.056	-0.348	-0.038	0.005			

MIXING STACK PRESSURE DATA, RUN 7 FLAT SIDE DISTRIBUTION

X/S:	0.25	0.50	0.75	1.00	1.25	1.50	1.75
PHS<INH20>:	1.090	0.040	0.060	0.110	0.200	-0.540	0.160
PHSt:	0.107	0.004	0.006	0.011	0.020	-0.053	0.016
X/S:	2.00	2.25	2.50	2.75			
PHS<INH20>:	0.210	0.260	0.350	0.260			
PHSt:	0.021	0.026	0.034	0.026			

DATA TAKEN ON 2 FEBRUARY 1981
 DATA TAKEN BY D.L.RYAN
 2 TERTIARY PORTS OPEN/CUSP IN

NUMBER OF PRIMARY NOZZLES: 4 UPTAKE AREA: 107.51 SQ IN

PRIMARY NOZZLE AREA: 9.06 SQ IN AREA RATIO (AH/AP): 2.99

MIXING STACK LENGTH: 25.50 IN ORIFICE DIAMETER: 6.988 IN

MIXING STACK DIMENSION: 8.5 IN ORIFICE BETA: 0.497 IN

MIXING STACK L/S: 3.99 AMBIENT PRESSURE: 30.26 IN HG

RUN	H	POR	DPOR	TOR	TUPT	TAMB	PU-PA	PA-PS	PA-PT	SECONDARY AREA		TERTIARY AREA
										IN OF H2O	SQUARE INCHES	
1	9.7	21.9	43.5	97.3	58.0	7.20	2.78	1.00	0.000	0.000	0.000	0.000
2	9.7	21.9	42.6	96.9	58.0	7.35	2.63	0.26	0.000	0.000	0.000	6.283
3	9.7	21.9	42.5	96.6	58.6	7.48	2.60	0.19	0.000	0.000	0.000	12.566
4	9.7	21.9	43.1	96.7	58.0	7.49	2.57	0.03	0.000	0.000	0.000	25.133
5	9.7	21.9	42.6	96.9	58.0	7.46	2.57	0.01	0.000	0.000	0.000	58.265
6	9.7	21.9	42.9	96.9	58.0	7.43	2.56	0.00	0.000	0.000	0.000	100.531
7	9.7	21.9	42.9	96.8	58.0	7.40	2.56	0.00	0.000	0.000	0.000	111.111

SECONDARY BOX

RUN	H	H1	Px	Tx	P#-PT# W#T#	.44	WP	HS.	UP	UH	UUPT	UPT MACH
1	0.0909	0.2677	0.9294	0.2880	0.0000	3.8152	0.0000	211.76	70.85	71.42	0.062	
2	0.0500	0.2534	0.9301	0.2724	0.0000	3.8137	0.0000	211.72	70.84	71.41	0.062	
3	0.0600	0.2507	0.9306	0.2694	0.0000	3.8191	0.0000	211.62	70.86	71.37	0.062	
4	0.0000	0.2481	0.9304	0.2666	0.0000	3.8169	0.0000	211.51	70.76	71.33	0.062	
5	0.0000	0.2477	0.9301	0.2663	0.0000	3.8187	0.0000	211.69	70.83	71.39	0.062	
6	0.0000	0.2468	0.9301	0.2654	0.0000	3.8175	0.0000	211.62	70.80	71.37	0.062	
7	0.0000	0.2469	0.9303	0.2654	0.0000	3.8175	0.0000	211.59	70.79	71.36	0.062	

TABLE VIIa

TERTIARY BOX									
	WT%	PT%	TT%	PTx/TT%	WTx/TT%	44	WH	WT	UE
RUN							LBH/SEC	LBH/SEC	FT/SEC
1	0.0000	0.0563	0.9294	0.1036	0.0000		3.615	0.000	70.85
2	0.0291	0.0250	0.9301	0.0269	0.0282		3.619	0.111	72.90
3	0.0361	0.0096	0.9306	0.0194	0.0349		3.619	0.139	73.35
4	0.0395	0.0022	0.9304	0.0031	0.0383		3.617	0.151	73.56
5	0.0323	0.0005	0.9301	0.0005	0.0313		3.619	0.123	73.11
6	0.0000	0.0000	0.9301	0.0000	0.0000		3.616	0.000	70.80
7	0.0000	0.0000	0.9303	0.0000	*****		3.616	*****	*****

TABLE VIIIb

DATA TAKEN ON 2 FEBRUARY 1981
 DATA TAKEN BY D.L.RYAN
 2 TERTIARY PORTS OPEN/CUSP IN

NUMBER OF PRIMARY NOZZLES: 4 UPTAKE AREA: 107.51 SQ IN
 PRIMARY NOZZLE AREA: 9.06 SQ IN AREA RATIO (AM/AP): 2.99
 MIXING STACK LENGTH: 25.50 IN ORIFICE DIAMETER: 6.966 IN
 MIXING STACK DIMENSION: 9.5 IN ORIFICE BETA: 0.497 IN
 MIXING STACK L/S: 3.00 AMBIENT PRESSURE: 30.24 IN HG

RUN	H POR IN OF H2O	DPOR DEGREES F	TOR TUPT	TAMB	PU-PA IN OF H2O	PA-PS IN OF H2O	PA-PT IN OF H2O	SECONDARY AREA		TERTIARY AREA	
								SQUARE INCHES	SQUARE INCHES	SQUARE INCHES	SQUARE INCHES
1	0.7	22.1	43.0	96.9	57.0	9.06	1.04	1.06	37.699	0.900	6.283
2	0.7	22.1	43.4	97.5	57.0	9.10	0.90	0.27	37.699	12.566	12.566
3	0.7	22.1	43.7	97.2	57.0	9.10	0.90	0.08	37.699	25.133	25.133
4	0.7	22.0	42.8	96.9	57.0	9.10	0.89	0.04	37.699	59.265	59.265
5	0.7	22.0	43.1	96.8	57.0	9.10	0.89	0.01	37.699	160.531	160.531
6	0.7	22.0	43.5	97.0	57.0	9.10	0.89	0.01	37.699	160.531	160.531
7	0.7	22.0	43.6	97.3	57.0	9.10	0.89	0.00	37.699	160.531	160.531

RUN	SECONDARY BOX				UPTAKE BOX				TERTIARY BOX			
	H	W	P+	T+	P+	T+	W+	U+	W+	U+	UPT	MACH
1	0.3473	0.0999	0.9283	0.1077	0.3367	3.8333	1.3335	211.65	93.71	71.45	0.062	0.062
2	0.3237	0.0864	0.9273	0.0932	0.3132	3.8318	1.2405	211.92	92.14	71.47	0.062	0.062
3	0.3233	0.0866	0.9278	0.0933	0.3133	3.8306	1.2405	211.75	92.08	71.41	0.062	0.062
4	0.3225	0.0859	0.9283	0.0926	0.3121	3.8254	1.2336	211.34	91.83	71.27	0.062	0.062
5	0.3226	0.0860	0.9285	0.0927	0.3122	3.8242	1.2336	211.23	91.79	71.24	0.062	0.062
6	0.3227	0.0860	0.9281	0.0927	0.3123	3.8227	1.2336	211.23	91.79	71.24	0.062	0.062
7	0.3227	0.0860	0.9276	0.0927	0.3122	3.8223	1.2336	211.32	91.82	71.27	0.062	0.062

TERTIARY BOX

RUN	PT ₁ PT ₂ WT ₁ WT ₂ .44						HM	WT	UE
	W	WT ₂	PT ₁	PT ₂	WT ₁	LBH/SEC			
1	0.0000	0.0961	0.9283	0.1035	0.0000	5.167	0.600	93.71	
2	0.0236	0.0259	0.9273	0.0280	0.0286	5.072	0.113	94.20	
3	0.0322	0.0077	0.9278	0.0083	0.0311	5.071	0.123	94.32	
4	0.0456	0.0039	0.9283	0.0042	0.0441	5.059	0.174	94.99	
5	0.0456	0.0019	0.9285	0.0016	0.0441	5.058	0.174	94.96	
6	0.0645	0.0005	0.9281	0.0005	0.0624	5.056	0.247	96.27	
7	xxxxxx	0.0000	0.9276	0.0000	xxxxxx	5.056	xxxxxx	xxxxxx	

TABLE VIIIb

DATA TAKEN ON 2 FEBRUARY 1961
 DATA TAKEN BY D. L. RYAN
 2 TERTIARY PORTS OPEN/CUSP IN

NUMBER OF PRIMARY NOZZLES: 4 UPTAKE AREA: 107.51 SQ IN

PRIMARY NOZZLE AREA: 9.06 SQ IN AREA RATIO (AM/AP): 2.99

MIXING STACK LENGTH: 25.50 IN ORIFICE DIAMETER: 6.908 IN

MIXING STACK DIMENSION: 8.5 IN ORIFICE BETA: 0.497 IN

MIXING STACK L/S: 3.00 AMBIENT PRESSURE: 30.24 IN HG

RUN	H FOR IN OF H2O	DPOR	TOR	T UPT	TAMB	PU-PA	PA-PS	PA-PT	SECONDARY AREA		TERTIARY AREA	
									DEGREES F	IN OF H2O	SQUARE INCHES	SQUARE INCHES
1	0.7	21.8	43.5	94.7	55.0	9.85	0.01	1.20	*****	*****	0.000	0.000
2	0.7	21.8	43.6	96.2	55.0	9.89	0.01	0.25	*****	*****	6.283	6.283
3	0.7	21.8	43.9	96.8	55.0	9.85	0.01	0.11	*****	*****	12.566	12.566
4	0.7	21.9	43.3	96.6	55.0	9.85	0.01	0.93	*****	*****	25.133	25.133
5	0.7	21.9	43.5	96.9	55.0	9.85	0.01	0.01	*****	*****	50.265	50.265
6	0.7	21.8	43.8	97.0	55.0	9.85	0.01	0.00	*****	*****	100.531	100.531
7	0.7	21.9	43.7	97.2	55.0	9.85	0.00	0.00	*****	*****	*****	*****

SECONDARY BOX

P.U.H	H	H ₂	P ₂	T ₂	P ₄ -T ₄	H ₄	H ₅	U _P	U _M	U _{PT}	U _{FT} MACH	LBM/SEC	
												LBMIN/SEC	FT/SEC
1	0.3372	0.0942	0.9298	0.1013	0.3265	3.8165	1.2868	210.89	92.62	71.13	0.061		
2	0.3069	0.0777	0.9298	0.0836	0.2964	3.8184	1.1686	210.91	98.60	71.13	0.062		
3	0.3603	0.0748	0.9299	0.0905	0.2909	3.8191	1.1465	210.84	99.29	71.11	0.061		
4	0.2964	0.0739	0.9303	0.0795	0.2972	3.8169	1.1315	210.69	89.89	71.06	0.061		
5	0.2961	0.0728	0.9303	0.0783	0.2868	3.8216	1.1315	210.96	83.98	71.15	0.062		
6	0.2958	0.0727	0.9303	0.0781	0.2865	3.8252	1.1315	211.15	90.95	71.21	0.062		
7	0.2938	0.0717	0.9304	0.0771	0.2846	3.8256	1.1239	211.13	89.91	71.20	0.062		

TABLE IXa

TERTIARY BOX										
RUN	W	WT%	PT%	TT%	PT1/TT%	WT1/TT%	.44	HM	WT	UE
								LBM/SEC	LBM/SEC	FT/SEC
1	0.0000	0.1389	0.9298	0.1494	0.9999	0.9999	5.103	0.000	92.62	
2	0.0361	0.0388	0.9298	0.0418	0.0349	0.0349	4.987	0.138	93.11	
3	0.0470	0.0165	0.9299	0.0178	0.0456	0.0456	4.965	0.180	93.46	
4	0.0510	0.0049	0.9303	0.0052	0.0494	0.0494	4.948	0.195	93.43	
5	0.0456	0.0016	0.9303	0.0016	0.0442	0.0442	4.953	0.174	93.15	
6	0.0000	0.0000	0.9303	0.0000	0.0000	0.0000	4.957	0.000	90.85	
7	*****	0.0000	0.9304	0.0000	*****	*****	4.956	*****	*****	

DATA TAKEN ON 2 FEBRUARY 1961
 DATA TAKEN BY D.L.RYAN
 3 TERTIARY PORTS OPEN/CUSP IN

NUMBER OF PRIMARY NOZZLES: 4 UPTAKE AREA: 107.51 SQ IN

PRIMARY NOZZLE AREA: 9.06 SQ IN AREA RATIO (AM/AP): 2.99

MIXING STACK LENGTH: 25.56 IN ORIFICE DIAMETER: 6.908 IN

MIXING STACK DIMENSION: 6.5 IN ORIFICE BETA: 0.497 IN

MIXING STACK L/S: 3.00 AMBIENT PRESSURE: 30.24 IN HG

N	POF	DPOR	TOR	TUPT	TAMB	PU-PA	PA-PS	PA-PT	SECONDARY AREA	TERTIARY AREA	SQUARE INCHES
1	0.7	21.9	42.3	96.7	58.0	7.25	2.77	1.34	0.000	0.000	0.000
2	0.7	21.9	41.9	96.5	58.0	7.45	2.59	0.36	0.000	0.000	6.283
3	0.7	21.9	42.8	96.6	59.0	7.59	2.44	0.16	0.000	0.000	12.566
4	0.7	21.9	43.0	96.7	58.9	7.50	2.40	0.05	0.000	0.000	25.133
5	0.7	21.9	43.2	96.8	58.0	7.55	2.40	0.01	0.000	0.000	50.265
6	0.7	21.9	42.6	96.7	58.0	7.59	2.49	0.00	0.000	0.000	100.531
7	0.7	21.9	42.9	96.8	58.0	7.59	2.38	0.00	0.000	0.000	*****

SECONDARY BOX

RUN	W*	P*	T*	P*/T*	W11-T11	44	WP	HS	UP	UM	UWPT	UPT MACH
1	0.0000	0.2667	0.9304	0.2866	0.0000	3.8186	0.0000	211.65	70.88	71.45	0.062	
2	0.0000	0.2419	0.9308	0.2569	0.0000	3.6201	0.0000	211.72	70.84	71.41	0.062	
3	0.0000	0.2356	0.9306	0.2532	0.0000	3.8167	0.0000	211.54	70.77	71.34	0.062	
4	0.0000	0.2318	0.9304	0.2491	0.0000	3.8159	0.0000	211.52	70.77	71.34	0.062	
5	0.9999	0.2319	0.9303	0.2492	0.0000	3.8151	0.0000	211.51	70.76	71.33	0.062	
6	0.0000	0.2316	0.9304	0.2489	0.0000	3.8174	0.0000	211.66	70.79	71.36	0.062	
7	0.9999	0.2298	0.9303	0.2476	0.0000	3.9163	0.0000	211.56	70.79	71.35	0.062	

TABLE Xa

TABLE Xb

TERTIARY BOX									
	N	WT*	PT*	TT*	PT*/TT*	WT/TT~.44	HM	WT	UE
RUN							LB/M/SEC	LB/M/SEC	FT/SEC
1	0.0000	0.1290	0.9304	0.1387		0.0000	3.819	0.0000	70.88
2	0.0342	0.9147	0.9393	0.9373		0.0331	3.820	0.131	73.26
3	0.0456	0.6155	0.9306	0.9166		0.0442	3.817	0.174	74.09
4	0.0510	0.6048	0.9304	0.9052		0.0494	3.816	0.195	74.39
5	0.0323	0.6005	0.9303	0.9005		0.0313	3.815	0.123	73.05
6	0.0000	0.6000	0.9304	0.6000		0.0000	3.817	0.000	70.79
7	*****	0.6000	0.9393	0.6000		*****	3.816	*****	*****

DATA TAKEN ON 2 FEBRUARY 1961
 DATA TAKEN BY D.L.RYAN
 3 TERTIARY PORTS OPEN/CUSP IN
 NUMBER OF PRIMARY NOZZLES: 4 UPTAKE AREA: 107.51 SQ IN
 PRIMARY NOZZLE AREA: 9.96 SQ IN AREA RATIO (AM/AP): 2.99
 SURFACE DIAMETER: 6.908 IN
 MIXING STACK LENGTH: 25.50 IN
 MIXING STACK DIMENSION: 8.5 IN ORIFICE BETA: 0.497 IN
 MIXING STACK L/S: 3.00 AMBIENT PRESSURE: 30.25 IN HG

RUN	H	POR	OPOR	TOR	TUPI	TAMB	PU-PA	PA-PS	PA-PT	SECONDARY AREA		TERTIARY AREA	
										IN OF H2O	SQUARE INCHES	SQUARE INCHES	
1	0.7	21.9	43.0	97.1	58.0	9.00	0.97	1.43		37.699	0.000		
2	0.7	21.9	42.5	97.1	58.0	9.15	0.80	0.40		37.699	6.283		
3	0.7	21.9	42.6	97.0	58.0	9.15	0.77	0.17		37.699	1.256		
4	0.7	21.9	42.9	96.8	58.0	9.20	0.75	0.05		37.699	25.133		
5	0.7	22.8	43.9	96.9	58.0	9.20	0.75	0.01		37.699	50.265		
6	0.7	22.8	43.9	96.8	58.0	9.15	0.75	0.00		37.699	100.531		
7	0.7	22.8	42.9	96.7	58.0	9.20	0.74	0.00		37.699	111111		

RUN	SECONDARY BOX						TERTIARY BOX						
	H	W*	P*	T*	P* \times T*	W*T*	44	UP	WS	UP	UM	UUFT	UPT MACH
								LBM/SEC	LBM/SEC	FT/SEC	FT/SEC	FT/SEC	FT/SEC
1	0.3372	0.0942	0.9298	0.1013	0.3265	3.8165	1.2868	210.89	92.62	71.13	0.061		
2	0.3060	0.0777	0.9238	0.0836	0.2364	3.8124	1.1636	210.51	90.60	71.13	0.062		
3	0.3093	0.0748	0.9239	0.0805	0.2966	3.8131	1.1465	210.84	90.20	71.11	0.061		
4	0.2964	0.0730	0.9303	0.0785	0.2872	3.8169	1.1315	210.69	69.89	71.06	0.061		
5	0.2961	0.0728	0.9303	0.0783	0.2863	3.8218	1.1315	210.96	89.98	71.15	0.062		
6	0.2958	0.0727	0.9303	0.0781	0.2865	3.8252	1.1315	211.15	90.95	71.21	0.062		
7	0.2938	0.0717	0.9304	0.0771	0.2846	3.8256	1.1239	211.13	89.91	71.20	0.062		

TABLE XIa

TABLE XIb

TERTIARY BOX							
RUN	N	WT	PTz	TTz	PTz/TTz	WT	UE
					LBM/SEC	LBM/SEC	FT/SEC
1	0.0000	0.1389	0.9298	0.1494	0.0000	5.103	0.000
2	0.0361	0.0388	0.9298	0.0418	0.0349	4.987	0.138
3	0.0479	0.0165	0.9293	0.0178	0.0456	4.965	0.180
4	0.0516	0.0049	0.9303	0.0052	0.0494	4.948	0.195
5	0.0436	0.0010	0.9303	0.0010	0.0442	4.953	0.174
6	0.0000	0.0000	0.9303	0.0000	0.0000	4.957	0.000
7	*****	0.0000	0.9304	0.0000	*****	4.950	*****

0018 TAKEN ON 2 FEBRUARY 1981

DATA TAKEN BY D.L.RYAN
3 TERRITORY PORTS OPEN/CLOSED
21 JUNE 1961

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NUMBER 40 PAPER, HEEH, 1873. 31 DEC 1941

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WIXINC SITACK LEMCHI: 23 . 36 IN ORIFICE DIAMETER: 6 . 966 IN

MIXING STACK DIMENSION: 6.5 IN ORIFICE BETA: 0.497 IN

MIXING STACK 1/8" 3.00 INCHES DIAMETER PRESSURE 3.0 25 INCHES

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RUN	IN OF H ₂ O	DEGREES F	IN OF H ₂ O
1	1.0	100	0.0
2	1.0	100	0.0
3	1.0	100	0.0
4	1.0	100	0.0
5	1.0	100	0.0
6	1.0	100	0.0
7	1.0	100	0.0
8	1.0	100	0.0
9	1.0	100	0.0
10	1.0	100	0.0
11	1.0	100	0.0
12	1.0	100	0.0
13	1.0	100	0.0
14	1.0	100	0.0
15	1.0	100	0.0
16	1.0	100	0.0
17	1.0	100	0.0
18	1.0	100	0.0
19	1.0	100	0.0
20	1.0	100	0.0
21	1.0	100	0.0
22	1.0	100	0.0
23	1.0	100	0.0
24	1.0	100	0.0
25	1.0	100	0.0
26	1.0	100	0.0
27	1.0	100	0.0
28	1.0	100	0.0
29	1.0	100	0.0
30	1.0	100	0.0
31	1.0	100	0.0
32	1.0	100	0.0
33	1.0	100	0.0
34	1.0	100	0.0
35	1.0	100	0.0
36	1.0	100	0.0
37	1.0	100	0.0
38	1.0	100	0.0
39	1.0	100	0.0
40	1.0	100	0.0
41	1.0	100	0.0
42	1.0	100	0.0
43	1.0	100	0.0
44	1.0	100	0.0
45	1.0	100	0.0
46	1.0	100	0.0
47	1.0	100	0.0
48	1.0	100	0.0
49	1.0	100	0.0
50	1.0	100	0.0
51	1.0	100	0.0
52	1.0	100	0.0
53	1.0	100	0.0
54	1.0	100	0.0
55	1.0	100	0.0
56	1.0	100	0.0
57	1.0	100	0.0
58	1.0	100	0.0
59	1.0	100	0.0
60	1.0	100	0.0
61	1.0	100	0.0
62	1.0	100	0.0
63	1.0	100	0.0
64	1.0	100	0.0
65	1.0	100	0.0
66	1.0	100	0.0
67	1.0	100	0.0
68	1.0	100	0.0
69	1.0	100	0.0
70	1.0	100	0.0
71	1.0	100	0.0
72	1.0	100	0.0
73	1.0	100	0.0
74	1.0	100	0.0
75	1.0	100	0.0
76	1.0	100	0.0
77	1.0	100	0.0
78	1.0	100	0.0
79	1.0	100	0.0
80	1.0	100	0.0
81	1.0	100	0.0
82	1.0	100	0.0
83	1.0	100	0.0
84	1.0	100	0.0
85	1.0	100	0.0
86	1.0	100	0.0
87	1.0	100	0.0
88	1.0	100	0.0
89	1.0	100	0.0
90	1.0	100	0.0
91	1.0	100	0.0
92	1.0	100	0.0
93	1.0	100	0.0
94	1.0	100	0.0
95	1.0	100	0.0
96	1.0	100	0.0
97	1.0	100	0.0
98	1.0	100	0.0
99	1.0	100	0.0
100	1.0	100	0.0

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SECONDARY BOX

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3	44444444	0.90000	0.93001	0.90000	44444444	3.8173	0.0000	216.36	74.31	70.95	0.051
4	44444444	0.90000	0.9303	0.90000	44444444	3.8181	0.0000	216.37	74.34	70.95	0.051
5	44444444	0.90000	0.93008	0.90000	44444444	3.8177	0.0000	216.33	74.31	70.94	0.051
6	44444444	0.90000	0.9308	0.90000	44444444	3.8196	0.0000	216.34	74.37	70.94	0.051
7	44444444	0.90000	0.9306	0.90000	44444444	3.8023	0.0000	216.30	74.33	70.95	0.051

TABLE XI

TABLE XIIIb

TERTIARY BOX									
RUN	W	WT%	PT%	TT%	PT% / TT%	WT% / TT% ~ .44	WM	WT	UE
1	0.0000	0.1531	0.9299	0.1668	0.0000	0.0000	0.0000	0.000	xxxxxx
2	0.0337	0.0342	0.9364	0.0367	0.0327	0.0327	0.129	0.129	xxxxxx
3	0.0434	0.0142	0.9301	0.0152	0.0421	0.0421	0.166	0.166	xxxxxx
4	0.0484	0.0044	0.9303	0.0047	0.0469	0.0469	0.185	0.185	xxxxxx
5	0.0323	0.0095	0.9366	0.0005	0.0313	0.0313	0.123	0.123	xxxxxx
6	0.0000	0.0000	0.9368	0.0000	0.0000	0.0000	0.000	0.000	xxxxxx
7	0.0000	0.0000	0.9368	0.0000	0.0000	0.0000	0.000	0.000	xxxxxx

DATA TAKEN ON 9 MARCH 1981
 DATA TAKEN BY D. L. RYAN
 2 TERTIARY PORTS OPEN/SIDE MOD
 NUMBER OF PRIMARY NOZZLES: 4 UP TAKE AREA: 107.31 SQ IN
 PRIMARY NOZZLE AREA: 9.86 SQ IN AREA RATIO (AM/AP): 2.99
 MIXING STACK LENGTH: 25.50 IN ORIFICE DIAMETER: 6.906 IN
 MIXING STACK DIMENSION: 8.5 IN ORIFICE BETA: 0.497 IN
 MIXING STACK L/S: 33.99 AMBIENT PRESSURE: 30.23 IN HG

IN	POF	TOR	TUPT	TAMB	PU-PA	PA-PS	PA-PT	SECONDARY AREA	TERTIARY AREA
RUN	IN OF H ₂ O	DEGREES F		IN OF H ₂ O				SQUARE INCHES	SQUARE INCHES
1	9.7	22.0	58.9	113.2	83.0	7.80	2.17	9.99	6.293
2	9.7	22.0	57.6	112.6	83.0	9.10	0.86	0.00	25.133
3	9.7	22.0	59.6	113.0	83.0	9.79	0.15	0.00	100.531

SECONDARY BOX

IN	HS	PS	T ₂	P ₂ /T ₂ HS ^{1/2}	WP	HS	UP	UH	UOPT	UPT MACH
PSH										
							LBM/SEC	LBM/SEC	FT/SEC	FT/SEC
1	0.0632	0.2130	0.9473	0.2249	0.0812	3.7652	0.3132	214.64	77.51	72.46
2	0.2692	0.0050	0.9483	0.0896	0.2644	3.7693	0.7987	214.18	85.85	72.23
3	0.3502	0.0149	0.9476	0.0157	0.3420	3.7623	1.3175	213.55	95.15	72.02

TABLE XIIIa

TERTIARY BOX

	N	WT%	PT%	T%	PT%/T%	WT%	WT	UE	
RUN									LBM/SEC LBM/SEC FT/SEC
1	*****	0.0000	0.9473	0.0000	*****	4.973	*****	*****	
2	*****	0.0000	0.9483	0.0000	*****	4.559	*****	*****	
3	*****	0.0000	0.9476	0.0000	*****	5.089	*****	*****	

MIXING STACK PRESSURE DATA: RUN 3 CURVED SIDE DISTRIBUTION

X/S:	0.25	0.50	0.75	1.00	1.25	1.50	1.75
PMS(CINH20):	-0.040	0.210	0.640	1.440	2.320	2.720	2.910
PMS%:	-0.004	0.021	0.064	0.143	0.231	0.270	0.290

X/S:	2.00	2.25	2.50	2.75			
PMS(CINH20):	0.040	-0.980	-0.740	-0.710			
PMS%:	0.004	-0.997	-0.074	-0.071			

MIXING STACK PRESSURE DATA: RUN 3 FLAT SIDE DISTRIBUTION

X/S:	0.25	0.50	0.75	1.00	1.25	1.50	1.75
PMS(CINH20):	0.710	0.330	0.320	0.570	0.640	0.580	0.320
PMS%:	0.071	0.039	0.032	0.057	0.064	0.058	0.032

X/S:	2.00	2.25	2.50	2.75			
PMS(CINH20):	-0.130	0.320	0.450	0.250			
PMS%:	-0.013	0.032	0.045	0.025			

TABLE XIIIb

VERTICAL VELOCITY TRAVERSE					
POSITION(IN)	0.98	0.50	1.00	1.50	2.00
P(IN H20)	1.16	2.85	2.54	2.19	2.05
V(FT/SEC)	72.50	113.64	107.28	99.62	96.38
POSITION(IN)	3.50	4.00	4.50	5.00	5.50
P(IN H20)	1.96	2.83	2.12	2.19	2.28
V(FT/SEC)	94.24	95.91	98.01	99.62	101.64
POSITION(IN)	7.00	7.50	8.00	8.50	9.00
P(IN H20)	2.67	2.74	2.31	1.72	1.72
V(FT/SEC)	110.00	111.43	102.31	88.28	88.28

HORIZONTAL VELOCITY TRAVERSE					
POSITION(IN)	0.98	0.50	1.00	1.50	2.00
P(IN H20)	0.94	2.12	2.03	2.01	1.95
V(FT/SEC)	65.27	98.01	97.08	95.44	94.68
POSITION(IN)	3.50	4.00	4.50	5.00	5.50
P(IN H20)	1.71	1.75	1.83	1.92	2.05
V(FT/SEC)	68.63	69.65	91.66	93.28	96.38
POSITION(IN)	6.59	6.75	7.00	7.25	7.75
P(IN H20)	2.13	2.12	2.12	2.13	2.18
V(FT/SEC)	98.24	98.01	98.01	98.24	97.55
POSITION(IN)	9.25	9.75	10.25	10.75	11.25
P(IN H20)	2.68	2.11	2.11	2.04	2.02
V(FT/SEC)	97.00	97.78	97.78	96.15	95.67
POSITION(IN)	12.75	12.75	12.75	12.75	12.75
P(IN H20)	1.43	1.43	1.43	1.43	1.43
V(FT/SEC)	86.50	86.50	86.50	86.50	86.50

DIAGONAL VELOCITY TRAVERSE					
POSITION(IN)	0.00	0.50	1.00	1.50	2.00
P(IN H20)	1.76	2.64	2.83	2.65	2.55
V(FT/SEC)	89.30	109.33	114.24	113.64	107.43
POSITION(IN)	3.50	4.00	4.50	5.00	5.50
P(IN H20)	2.04	2.03	2.00	2.03	2.06
V(FT/SEC)	96.15	95.91	95.20	95.91	96.62
POSITION(IN)	7.00	7.50	8.00	8.50	9.00
P(IN H20)	2.19	2.12	2.12	2.09	2.02
V(FT/SEC)	97.55	93.01	96.01	97.32	95.67
POSITION(IN)	10.50	11.00	11.50	12.00	12.50
P(IN H20)	1.73	1.80	1.95	2.18	2.51
V(FT/SEC)	88.54	90.31	94.00	99.39	106.65
POSITION(IN)	14.00	14.50	15.00	113.04	110.14
P(IN H20)	2.98	2.78	2.95	2.95	2.95
V(FT/SEC)	116.21	112.24	65.61	65.61	65.61

TABLE XIII

DATA TAKEN ON 9 MARCH 1981
 DATA TAKEN BY D.L.RYAN
 2 TERTIARY PORTS OPEN/SIDE HOD
 NUMBER OF PRIMARY NOZZLES: 4 UPTAKE AREA: 187.51 SQ IN
 PRIMARY NOZZLE AREA: 9.06 SQ IN AREA RATIO (AM/AP): 2.99
 MIXING STACK LENGTH: 25.50 IN ORIFICE DIAMETER: 6.968 IN
 MIXING STACK DIMENSION: 6.5 IN ORIFICE BETA: 0.497 IN
 MIXING STACK L/S: 3.00 AMBIENT PRESSURE: 30.29 IN HG

RUN	N	POR	DPOR	TOR	TUPT	TAMB	PU-PA	PA-PS	PR-PT	SECONDARY AREA		TERTIARY AREA
										IN OF H2O	SQUARE INCHES	
1	0.7	22.0	47.8	100.2	57.0	7.46	2.59	0.31	-	0.000	6.283	
2	0.7	22.0	48.0	101.2	57.0	7.45	2.56	0.34	-	0.000	25.133	
3	0.7	22.0	47.8	101.4	57.0	7.45	2.54	0.31	-	0.000	100.531	

SECONDARY BOX

RUN	N	W ₁	P ₄	T ₂	P ₂ /T ₂ WATT: .44	WP	HS	UP	UM	UUPT	UPT MACH	FT/SEC
1	0.0000	0.2476	0.9228	0.2683	0.0000	3.6096	0.0000	212.24	71.01	71.58	0.062	
2	0.0000	0.2449	0.9212	0.2648	0.0000	3.8088	0.0000	212.56	71.12	71.69	0.062	
3	0.0000	0.2419	0.9209	0.2626	0.0000	3.8696	0.0000	212.67	71.15	71.72	0.062	

TABLE XIVa

TABLE XIVb

TERTIARY BOX									
	H	WT%	PT%	TT%	PT% / TT%	WT% / TT% ^a	WH	WT	UE
RUN	LB/H/SEC LB/H/SEC FT/SEC								
1	0.0319	0.0296	0.9226	0.9321	0.0306	0.0306	3.616	6.121	73.27
2	0.0458	0.0038	0.9212	0.9041	0.0442	0.0442	3.809	6.174	74.37
3	0.0648	0.0005	0.9269	0.9995	0.0625	0.0625	3.616	6.247	75.76

DATA TAKEN ON 9 MARCH 1981
 DATA TAKEN BY D.L.RYAN
 2 TERTIARY PORTS OPEN-SIDE MOD

NUMBER OF PRIMARY NOZZLES - 4 UPTAKE AREA - 107.51 SQ IN

PRIMARY NOZZLE AREA - 9.06 SQ IN AREA RATIO (AM/AP) - 2.99

MIXING STACK LENGTH - 25.50 IN ORIFICE DIAMETER - 6.900 IN

MIXING STACK DIMENSION - 8.5 IN ORIFICE BETA - 0.497 IN

MIXING STACK L/S - 3.00 AMBIENT PRESSURE - 30.30 IN HG

RUN	H	POR	DPOR	TOR	TUPT	TAMB	PU-PA	PA-PS	PA-PT	SECONDARY AREA		TERTIARY AREA
										IN OF H2O	SQUARE INCHES	
1	0.7	22.0	18.4	102.2	58.0	9.48	9.57	9.28	37.699	6.283		
2	0.7	22.0	18.8	102.2	58.0	9.48	9.37	9.84	37.699	25.133		
3	0.7	22.0	19.0	102.2	58.0	9.46	9.37	9.91	37.699	100.531		

SECONDARY BOX

RUN	H	W ₁	P ₂	T ₂	P ₂ /T ₂ H211.44	WP	HS	UP	UM	UUPT	UPT MACH
1	0.2593	0.0548	0.9213	0.0595	0.2591	3.8068	0.9872	211.79	87.76	71.43	0.061
2	0.2594	0.0548	0.9213	0.0595	0.2592	3.8065	0.9872	211.71	87.73	71.48	0.061
3	0.2594	0.0549	0.9213	0.0596	0.2592	3.8057	0.9872	211.67	87.72	71.39	0.061

TABLE Xva

TABLE XVIb

TERTIARY BOX									
RUN	H	HTS	PTS	TT*	PTS/TT*	HTTTT~.44	HM	HT	UE
							LBH/SEC	LBH/SEC	FT/SEC
1	0.0303	0.0269	0.9213	0.0292	0.0292	4.795	0.115	69.87	
2	0.0458	0.0338	0.9213	0.0642	0.0442	4.794	0.174	98.92	
3	0.0648	0.0095	0.9213	0.0095	0.0625	4.793	0.247	92.23	

DATA TAKEN ON 9 MARCH 1981
 DATA TAKEN BY D. L. RYAN
 3 TERTIARY PORTS OPEN/SIDE HOD
 NUMBER OF PRIMARY NOZZLES: 4 UPTAKE AREA: 107.51 SQ IN
 PRIMARY NOZZLE AREA: 9.06 SQ IN AREA RATIO (AM/AP): 2.99
 MIXING STACK LENGTH: 25.50 IN ORIFICE DIAMETER: 6.908 IN
 MIXING STACK DIMENSION: 8.5 IN ORIFICE BETA: 0.497 IN
 MIXING STACK L/S: 3.00 AMBIENT PRESSURE: 30.32 IN HG

RUN	IN OF H2O	DEGREES F	TAMB	PU-PA	PA-PS	PA-PT	SECONDARY AREA		TERTIARY AREA	
							IN OF H2O	SQUARE INCHES	SQUARE INCHES	SQUARE INCHES
1	0.7	21.9	53.4	107.0	68.0	8.10	1.86	0.00	6.283	44444444
2	0.7	22.0	53.2	107.2	68.0	9.20	0.81	0.00	25.133	22222222
3	0.7	22.0	52.6	106.6	68.0	9.09	0.13	0.00	160.531	24444444
4	0.7	21.9	53.4	107.0	68.0	9.03	0.03	0.00	44444444	44444444

SECONDARY BOX

RUN	H*	P*	T*	P*/T*	H*T~.44	WP	WS	UP	UN	UWPT	UPT MACH	LBH/SEC		LBH/SEC		FT/SEC		FT/SEC	
												LBH/SEC	LBH/SEC	FT/SEC	FT/SEC	FT/SEC	FT/SEC		
1	0.0779	0.1897	0.5312	0.1546	0.6755	3.7820	0.2945	212.67	76.29	71.72	0.061								
2	0.2659	0.9786	0.3308	0.0845	0.1987	3.7913	0.7774	212.73	84.73	71.74	0.061								
3	0.3284	0.0127	0.9318	0.0136	0.3183	3.7936	1.2457	212.28	92.74	71.59	0.061								
4	333333	0.00029	0.9312	0.0032	xxxxxx	3.7820	9.3562	211.73	444444	71.41	0.061								

TERTIARY BOX

	N	WT%	PT%	TT%	PT% / TT%	WT / TT%	.44	WT	WT	UE
RUN								LB/M SEC	LB/M SEC	FT / SEC
1		0.0000	0.9312	0.0000				*****	4.076	*****
2		0.0000	0.9306	0.0000				*****	4.569	*****
3		0.0000	0.9318	0.0000				*****	5.039	*****
4		0.0000	0.9312	0.0000				*****	*****	*****

MIXING STACK PRESSURE DATA: RUN 4 CURVED SIDE DISTRIBUTION

X/S:	0.25	0.50	0.75	1.00	1.25	1.50	1.75
PMS(CINH20):	0.030	0.400	0.850	1.570	2.440	2.910	2.140
PMS%:	0.009	0.083	0.154	0.239	0.285	0.210	
X/S:	0.009	0.039	0.154	0.239	0.285	0.210	
PMS(CINH20):	0.020	0.250	0.500	0.750	1.000	1.250	1.500
PMS%:	-0.020	-0.122	-0.085	-0.073	-0.073	-0.073	

MIXING STACK PRESSURE DATA: RUN 4 FLAT SIDE DISTRIBUTION

X/S:	0.25	0.50	0.75	1.00	1.25	1.50	1.75
PMS(CINH20):	0.850	0.690	0.610	0.930	1.020	0.770	0.740
PMS%:	0.883	0.559	0.060	0.991	0.100	0.375	0.973
X/S:	2.00	2.25	2.50	2.75			
PMS(CINH20):	-0.230	0.210	0.370	0.220			
PMS%:	-0.023	0.021	0.036	0.022			

TABLE XVIIb

VERTICAL VELOCITY TRAVERSE					
POSIT(IN):	0.00	0.50	1.00	1.50	2.00
P(IN H20):	1.57	2.77	2.35	2.07	1.92
V(FT/SEC):	83.05	116.31	101.61	95.36	91.84
POSIT(IN):	3.50	4.00	4.50	5.00	5.50
P(IN H20):	1.88	1.92	2.03	2.16	2.31
V(FT/SEC):	98.86	91.84	94.44	97.41	100.74
POSIT(IN):	7.00	7.50	8.00	8.50	9.00
P(IN H20):	2.86	2.93	2.69	2.95	2.98
V(FT/SEC):	112.09	113.45	108.71	64.69	64.69

HORIZONTAL VELOCITY TRAVERSE					
POSIT(IN):	0.00	0.50	1.00	1.50	2.00
P(IN H20):	1.05	1.85	1.80	1.86	1.85
V(FT/SEC):	67.92	98.15	88.92	90.39	90.15
POSIT(IN):	3.50	4.00	4.50	5.00	5.50
P(IN H20):	1.82	1.86	1.91	1.97	2.00
V(FT/SEC):	89.42	90.39	91.60	93.73	94.44
POSIT(IN):	6.50	6.75	7.00	7.25	7.75
P(IN H20):	2.01	1.98	2.00	1.99	1.98
V(FT/SEC):	93.97	93.26	93.73	93.58	93.26
POSIT(IN):	9.25	9.75	10.25	10.75	11.25
P(IN H20):	2.01	2.93	2.07	2.04	2.01
V(FT/SEC):	93.97	94.44	95.36	94.67	93.97
POSIT(IN):	12.75				
P(IN H20):	1.04				
V(FT/SEC):	67.59				

DIAGONAL VELOCITY TRAVERSE					
POSIT(IN):	0.00	0.50	1.00	1.50	2.00
P(IN H20):	1.30	2.43	2.74	2.78	2.63
V(FT/SEC):	75.57	103.32	109.71	110.51	107.49
POSIT(IN):	3.50	4.00	4.50	5.00	5.50
P(IN H20):	2.16	2.12	2.11	2.19	2.09
V(FT/SEC):	97.41	96.51	96.28	96.05	95.82
POSIT(IN):	7.00	7.50	8.00	8.50	9.00
P(IN H20):	1.93	1.94	1.94	1.99	2.00
V(FT/SEC):	92.08	92.32	92.32	93.50	93.73
POSIT(IN):	10.50	11.00	11.50	12.00	12.50
P(IN H20):	1.87	1.87	1.95	2.10	2.28
V(FT/SEC):	90.64	90.64	92.56	96.95	100.00
POSIT(IN):	14.00	14.50	15.00	16.00	16.67
P(IN H20):	2.47	2.30	1.98		
V(FT/SEC):	104.17	100.52	93.26		

TABLE XVIc

DATA TAKEN ON 9 MARCH 1981
DATA TAKEN BY D. L. RYAN

3 TERTIARY PORTS OPEN/SIDE MOD

NUMBER OF PRIMARY NOZZLES: 4 UPTAKE AREA: 107.51 SQ IN

PRIMARY NOZZLE AREA: 9.06 SQ IN AREA RATIO (CH/AP): 2.99

MIXING STACK LENGTH: 25.50 IN ORIFICE DIAMETER: 6.908 IN

MIXING STACK DIMENSION: 8.5 IN ORIFICE BETA: 0.497 IN

MIXING STACK L/S: 3.00 AMBIENT PRESSURE: 30.28 IN HG

RUN	H ₀	P ₀	D _{P0}	TOR	T _{UPT}	T _{AMB}	P _{U-PA}	P _{A-PS}	P _{A-PT}	SECONDARY AREA		TERTIARY AREA	
										IN OF H ₂ O	SQUARE INCHES	SQUARE INCHES	SQUARE INCHES
1	0.7	22.0	49.6	99.0	56.0	7.50	2.46	0.43	0.088	6.283			
2	0.7	22.0	49.4	100.8	56.0	7.60	2.35	0.64	0.088	25.133			
3	0.7	22.0	49.2	101.4	56.0	7.65	2.35	0.61	0.088	100.531			

SECONDARY BOX

RUN	H ₀	H ₁	P ₁	T ₁	P _{2/TT} H ₂ T ₂ .44	W _P	W _S	U _P	U _H	UUPT		UPT MACH	
										LBIN/SEC	LBIN/SEC	FT/SEC	FT/SEC
1	0.0000	0.2367	0.9238	0.2564	0.0000	3.8022	0.0000	211.38	70.72	71.29	0.062		
2	0.0000	0.2247	0.9201	0.2442	0.0000	3.8038	0.0000	212.04	70.94	71.51	0.062		
3	0.0000	0.2241	0.9191	0.2436	0.0000	3.8037	0.0000	212.31	71.03	71.60	0.062		

TABLE XVIIa

TERTIARY BOX

	N	WT%	PT%	TT%	PT% / TT% WT*TT^-.44	WH	WT	UE
RUN						LBH/SEC	LBH/SEC	FT/SEC
1	0.0376	0.0414	0.9236	0.9448	0.0363	3.802	0.143	73.38
2	0.9459	0.0038	0.9201	0.0042	0.0443	3.803	0.175	74.20
3	0.9918	0.0016	0.9191	0.0018	0.0885	3.804	0.349	77.55

TABLE XVIIIb

DATA TAKEN ON 9 MARCH 1981
 DATA TAKEN BY D. L. RYAN
 3 TERTIARY PORTS OPEN/SIDE MOD
 NUMBER OF PRIMARY NOZZLES: 4
 PRIMARY NOZZLE AREA: 9.96 SQ IN AREA RATIO (AM/AP): 2.99
 MIXING STACK LENGTH: 25.50 IN ORIFICE DIAMETER: 6.908 IN
 MIXING STACK DIMENSION: 6.5 IN ORIFICE BETA: 0.497 IN
 MIXING STACK L/S: 3.00 AMBIENT PRESSURE: 30.28 IN HG

RUN	H	P ₀	DPOR	TOR	T _{PT}	T _{AMB}	F	PU-PA	PA-PS	PA-PT	SECONDARY AREA		TERTIARY AREA
											IN OF H ₂ O	SQUARE INCHES	
1	6.7	22.0	48.6	102.0	58.0	9.58	0.53	6.40	-	-	37.699	6.283	
2	6.7	22.0	48.4	101.6	58.0	9.58	0.48	6.97	-	-	37.699	25.133	
3	6.7	22.0	48.4	101.8	58.0	9.58	0.46	6.81	-	-	37.699	100.531	

RUN	SECONDARY BOX						TERTIARY BOX					
	H	H ₂	P ₂	T ₂	P ₂ /T ₂	H ₂ T ₂	NP	HS	UP	UH	UUPT	UPT MACH
1	6.2500	0.0510	0.92117	0.0554	0.2412	3.8069	0.9517	211.72	87.14	71.41	0.061	
2	0.2379	0.0463	0.9223	0.0502	0.2296	3.8067	0.9057	211.59	86.31	71.36	0.061	
3	0.2329	0.0443	0.9220	0.0481	0.2247	3.8067	0.8866	211.65	86.96	71.38	0.061	

TABLE XVIIIa

TABLE XVIIIb

TERTIARY BOX							
RUN	N	WT _S	PT _S	TT _S	PT _S /TT _S	WT _T TT ⁻¹ ~.44	WT
						LBH/SEC	LBH/SEC
1	0.0362	0.0385	0.9217	0.0418	0.0349	4.758	6.138
2	0.0606	0.0067	0.9223	0.0073	0.0585	4.712	6.231
3	0.0916	0.0010	0.9226	0.0010	0.0684	4.693	6.349

APPENDIX: A

ONE-DIMENSIONAL ANALYSIS OF A SIMPLE EDUCTOR

This investigation, being an extension of the work of Shaw [Ref. 8], and others Ref. 1, 2 and 7] uses the same one-dimensional analysis of a simple eductor system. Similarity between the basic geometry tested by previous researchers was maintained in order to correlate data. The dimensionless parameters controlling the flow phenomena used previously were also used in this investigation along with the basic means of data analysis and presentation. Dynamic similarity was maintained by using Mach number similarity to establish the model's primary flow rate.

Although the analysis presented here is for an eductor model with only primary and secondary air flows, it should be kept in mind that many of the results presented are for systems with primary, secondary, and tertiary air flows. Systems with tertiary and film or wall cooling air flows have been non-dimensionalized with the same base parameters as the secondary air flow and have been calculated using the same one-dimensional analysis. This allows for easy comparison of the results. Parameters pertaining to the secondary systems are subscripted with an "s", those relating to the tertiary box are subscripted with a "t".

A. MODELING TECHNIQUE

Dynamic similarity between the models tested and the actual prototype was maintained by using the same primary air flow Mach number. For the primary air flow Mach number used (0.064), and based on the average flow properties within the mixing stack and the hydraulic diameter of the mixing stack, the air flow through the eductor system is turbulent ($Re > 10^5$). As a consequence of this, momentum exchange is predominant over shear interaction, and the kinetic and internal energy terms are more influential on the flow than are viscous forces. It can also be shown that the Mach number represents the ratio of kinetic energy of a flow to its internal energy and is, therefore, a more significant parameter than the Reynolds number in describing the primary flow through the uptakes.

B. ONE-DIMENSIONAL ANALYSIS OF A SIMPLE EDUCTOR

The theoretical analysis of an eductor may be approached in two ways. One method attempts to analyze the details of the mixing process of the primary and secondary air streams as it takes place inside the mixing stack. This requires an interpretation of the mixing phenomenon which, when applied to a multiple nozzle system, becomes extremely complex. The other method, which was chosen here, analyzes the overall performance of the eductor system and is not concerned with the actual mixing process. The one-dimensional analysis

based on a single primary nozzle exhausting into a mixing stack, as shown in Figure A-1. To avoid repetition with previous reports, only the main parameters and assumption will be represented here. A complete derivation of analysis used can be found in References [1] and [4]. The one-dimensional flow analysis of the simple eductor system described depends on the simultaneous solution of the continuity, momentum and energy equations coupled with the equation of state, all compatible with specific boundary conditions.

The idealizations made for simplifying the analysis are as follows:

1. The flow is steady state and incompressible.
2. Adiabatic flow exists throughout the eductor with isentropic flow of the secondary stream from the plenum (at section 0) to the throat or entrance of the mixing stack (at section 1) and irreversible adiabatic mixing of the primary and secondary streams occurs in the mixing stack (between sections 1 and 2).
3. The static pressure across the flow at the entrance and exit planes of the mixing-tube (at sections 1 and 2) is uniform.
4. At the mixing-stack entrance (section 1) the primary flow velocity U_p and temperature T_p are uniform across the primary stream, and the secondary flow velocity U_s and temperature T_s are uniform across

the secondary stream, but U_p does not equal U_s , and T_p does not equal T_s .

5. Incomplete mixing of the primary and secondary streams in the mixing stack is accounted for by the use of a non-dimensional momentum correction factor K_m which relates the actual momentum rate to the pseudo-rate based on the bulk-average velocity and density and by the use of a non-dimensional kinetic energy correction factor K_e which relates the actual kinetic energy rate to the pseudo-rate based on the bulk-average velocity and density.
6. Both gas flows behave as perfect gases.
7. Flow potential energy position changes are negligible.
8. Pressure changes P_{s0} to P_{s1} and P_1 to P_a are small relative to the static pressure so that the gas density is essentially dependent upon temperature (and atmospheric pressure).
9. Wall friction in the mixing stack is accounted for with the convectional pipe friction factor term based on the bulk-average flow velocity U_m and the mixing stack wall area A_w .

The following parameter, defined here for clarity, will be used in the following development.

$$\frac{A_p}{A_m}$$

area ratio of primary flow area to mixing stack cross sectional area

$$\frac{A_w}{A_m}$$

area ratio of wall friction area to mixing stack cross sectional area

$$k_p$$

momentum correction factor for primary mixing

$$K_m$$

momentum correction factor for mixed flow

$$f$$

wall friction factor

Based on the continuity equation, the conservation of mass principle for steady flow yields

$$W_m = W_p + W_s + W_t \quad (1)$$

where

$$W_p = \rho_p U_p A_p$$

$$W_s = \rho_s U_s A_s$$

$$W_t = \rho_t U_t A_t$$

$$W_m = \rho_m U_m A_m$$

(1a)

All of the above velocity and density terms, with the exception of ρ_m and U_m , are defined without ambiguity by the virtue of idealizations (3) and (4) above. Combining equations (1) and (1a) above, the bulk average velocity at the exit plane of the mixing stack becomes

$$U_m = \frac{W_s + W_t + W_p}{\rho_m A_m} \quad (1b)$$

where A_m is fixed by the geometric configuration and

$$\rho_m = \frac{P_a}{R T_m} \quad (2)$$

where T_m is calculated as the bulk average temperature from the energy equation (9) below. The momentum equation stems

from Newton's second and third laws of motion and is the conventional force and momentum-rate balance in fluid mechanics.

$$K_p \left(\frac{W_p U_p}{g_c} \right) + \left(\frac{W_s U_s}{g_c} \right) + \left(\frac{W_t U_t}{g_c} \right) + P_1 A_1 = K_m \left(\frac{W_m U_m}{g_c} \right) + P_2 A_2 + F_{fr} \quad (3)$$

Note the introduction of idealizations (3) and (5). To account for a possible non-uniform velocity profiles across the primary nozzle exit, the momentum correction factor K_p is introduced here. It is defined in a manner similar to that of K_m and by idealization (4), supported by work conducted by Moss, it is set equal to unity. K_p is carried through this analysis only to illustrate its effect on the final result. The momentum correction factor for the mixing stack exit is defined by the relation

$$K_m = \frac{1}{W_m U_m} \int_0^{A_m} U_m^2 \rho_2 dA \quad (4)$$

where U_m is evaluated as the bulk-average velocity from equation (1b). The wall skin friction force F_{fr} can be related to the flow stream velocity by

$$F_{fr} = f A_w \left(\frac{U_m^2 \rho_m}{2 g_c} \right) \quad (5)$$

using idealization (9). As a reasonably good approximation for turbulent flow, the friction factor may be calculated from the Reynolds number

$$f = 0.046 (Re_m)^{-0.2}. \quad (6)$$

Applying the conservation of energy principle to the steady flow system in the mixing stack between the entrance and exit planes,

$$\begin{aligned} W_p (h_p + \frac{U_p^2}{2g_c}) + W_s (h_s + \frac{U_s^2}{2g_c}) + W_t (h_t + \frac{U_t^2}{2g_c}) \\ = W_m (h_m + K_e \frac{U_m^2}{2g_c}) \end{aligned} \quad (7)$$

neglecting potential energy of position changes (idealization 7). Note the introduction of the kinetic energy correction factor K_e , which is defined by the relation

$$K_e = \frac{1}{W_m U_m^2} \int_0^{A_m} U_2^3 \rho_2 dA \quad (8)$$

It may be demonstrated that for the purpose of evaluating the mixed mean flow temperature T_m , the kinetic energy terms may be neglected to yield

$$h_m = \frac{W_p}{W_m} h_p + \frac{W_s}{W_m} h_s + \frac{W_t}{W_m} h_t \quad (9)$$

where $T_m = \phi(h_m)$ only, with the idealization (6).

The energy equation for the isentropic flow of the secondary air from the plenum to the entrance of the mixing stack may be shown to reduce to

$$\frac{P_o - P_s}{\rho_s} = \frac{U_s^2}{2g_s} \quad (10)$$

similarly, the energy equation for the tertiary air flow reduces to

$$\frac{P_o - P_t}{\rho_t} = \frac{U_t^2}{2g_c}$$

The foregoing equations may be combined to yield the vacuum produced by the eductor action in either the secondary or tertiary air plenums. For the secondary air plenum, the vacuum produced is

$$P_a - P_{os} = \frac{1}{g_c A_m} (K_p \frac{W_p^2}{A_p \rho_p} + \frac{W_s^2}{A_s \rho_s} (1 - \frac{1}{2} \frac{A_m}{A_s}) - \frac{W_m^2}{A_m \rho_m} (K_m + \frac{f}{2} \frac{A_w}{A_m})) \quad (11)$$

where it is understood that A_p and ρ_s apply to the secondary flow at this same section, and A_m and ρ_m apply to the mixed flow at the exit of the mixing stack system. P_a is atmospheric pressure, and is equal to the pressure at the exit of the mixing stack. A_w is the area of the inside wall of the mixing stack.

For the tertiary air plenum, the vacuum produced is

$$\begin{aligned} P_a - P_{ot} &= \frac{1}{g_c A_m} (K_p \frac{(W_p + W_s)^2}{A_p \rho_p + A_s \rho_s} + \frac{W_t^2}{A_t \rho_t} (1 - \frac{1}{2} \frac{A_m}{A_t})) \\ &= \frac{W_m^2}{A_m \rho_m} (K_m + \frac{f}{2} \frac{A_w}{A_m}) \end{aligned} \quad (11a)$$

where the primary flow now consists of both the primary and secondary air flows.

C. NON-DIMENSIONAL FORM OF THE SIMPLE EDUCTOR EQUATION

In order to provide the criteria of similarity of flows with geometric similarity, the non-dimensional parameters which govern the flow must be determined. The means chosen for determining these parameters is to normalize equations (11) and (11a) with the following dimensionless groupings.

$$\Delta P^* = \frac{P_a - P_{os}}{\frac{\rho_s}{U_p^2} \frac{p}{2g_c}}$$

a pressure coefficient which compares the pumped head $P_a - P_{os}$ for the secondary flow to the driving head U_p^2 of the primary flow

$$\Delta PT^* = \frac{P_a - P_{ot}}{\frac{\rho_t}{U_p^2} \frac{p}{2g_c}}$$

a pressure coefficient which compares the pumped head $P_a - P_{os}$ for the tertiary flow to the driving head U_p^2 of the primary flow

$$W^* = \frac{W_s}{W_p}$$

a flow rate ratio, secondary to primary mass flow rate

$$WT^* = \frac{W_t}{W_p}$$

a flow rate ratio, tertiary to primary mass flow rate

$$T^* = \frac{T_s}{T_p}$$

an absolute temperature ratio, secondary to primary

$$TT^* = \frac{T_t}{T_p}$$

an absolute temperature ratio, tertiary to primary

$$\rho_s^* = \frac{\rho_s}{p}$$

a flow density ratio of the secondary to primary flows. (note that since the fluids are considered perfect gases,

$$\rho_s^* = \frac{T_p}{T_s} = \frac{1}{T_s^*}$$

$$\rho_t^* = \frac{\rho_t}{\rho_p}$$

a flow density ratio of the tertiary or film cooling flow to primary flows. (Note that since the fluids are considered perfect gases,

$$\rho_t^* = \frac{T_p}{T_t} = \frac{1}{T_t^*}$$

$$A_s^* = \frac{A_s}{A_p}$$

an area ratio of secondary flow area to primary flow area

$$A_t^* = \frac{A_t}{A_p}$$

an area ratio of tertiary flow area to primary flow area

With these non-dimensional groupings, equations (11) and (11a) can be rewritten in dimensionless form. Since both equations follow the same format, only the results for the secondary air plenum will be presented here.

$$\begin{aligned} \frac{\Delta P^*}{T^*} = & 2 \frac{A_p}{A_m} \left(\left(K_p - \frac{A_p}{A_m} \beta \right) - W^* \left(K_p + T^* \right) \frac{A_p}{A_m} \beta \right. \\ & \left. + W^{*2} T^* \left(\frac{1}{A^*} \left(K_p - \frac{A_m}{2A^* A_p} \right) - \frac{A_p}{A_m} \beta \right) \right) \end{aligned} \quad (12)$$

where

$$\beta = K_m + \frac{f}{2} \frac{A_w}{A_m}$$

This may be rewritten as

$$\frac{P^*}{T^*} = C_1 + C_2 W^* (T + 1) + C_3 W^{*2} T^* \quad (13)$$

where

$$C_1 = 2 \frac{A_p}{A_m} (K_p - \frac{A_p}{A_m} \beta),$$

$$C_2 = -2 \left(\frac{A_p}{A_n} \right)^2 \beta, \text{ and}$$

$$C_3 = 2 \frac{A_p}{A_m} \left(\frac{1}{A^*} - \frac{A_m}{2A^*A_p} \beta - \frac{A_p}{A_m} \beta \right).$$

As can be seen from equation (13),

$$\Delta P^* = F(W^*, T^*).$$

The additional dimensionless quantities listed below were used to correlate the static pressure distribution down the length of the mixing stack.

$$PMS^* = \frac{\rho_s}{\frac{U_p^2}{2g_c}}$$

a pressure coefficient which compares the pumping head PMS for the secondary flow to the driving head $\frac{U_p^2}{2g_c}$ of the primary flow, where $PMS = \frac{\rho_s}{\frac{U_p^2}{2g_c}}$ static pressure along the mixing stack length

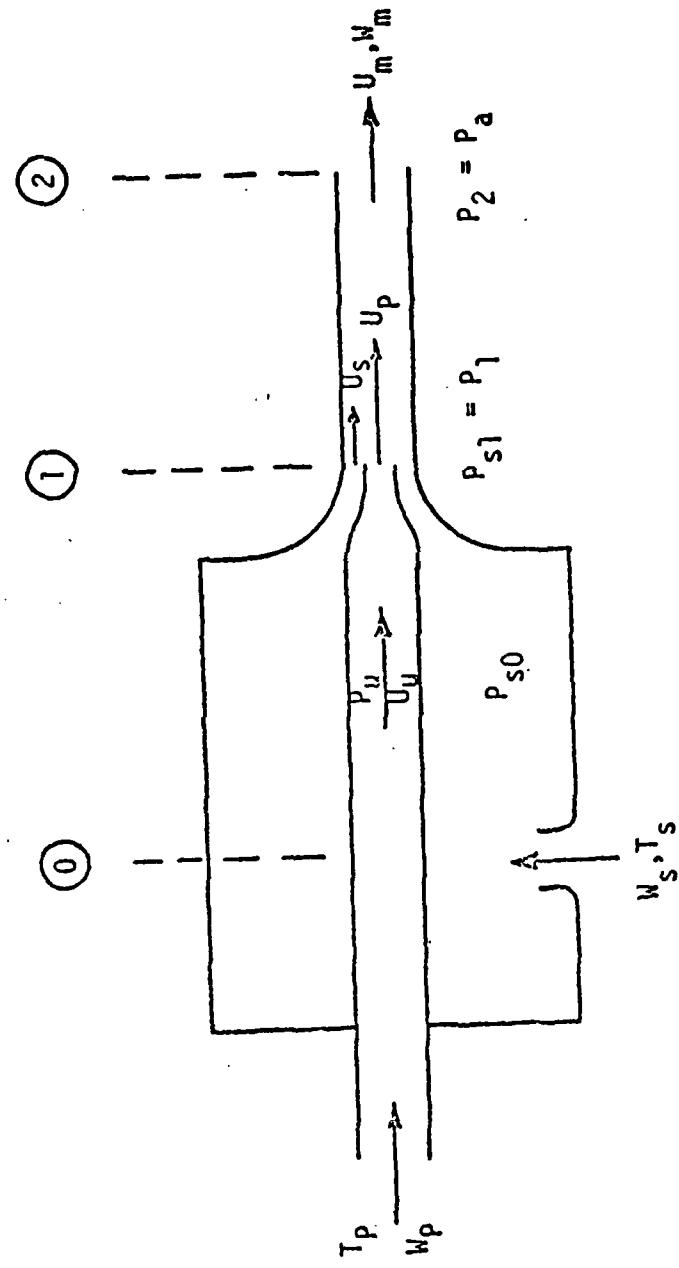


FIGURE A-1. SIMPLE SINGLE NOZZLE EDUCTOR SYSTEM

APPENDIX: B

FORMULA

Presented here are the formulas used to obtain the primary and secondary mass flow rates. According to the ASME Power Test Code (5), the general equation for mass flow rate appearing in equation (a)

$$W(\text{lbm/sec}) = (0.12705) K A Y F_a (\rho \Delta P)^{0.5} \quad (a)$$

may be used with flow nozzles and square edge orifices provided the flow is subsonic. In the above equation, K (dimensionless) represents the flow coefficient for the metering device and is defined as $K = C(1 - \beta^4)^{-0.5}$ where C is the coefficient of discharge and β is the ratio of throat to inlet diameters; A(in^2) is the total cross sectional area of the metering device; Y (dimensionless) is the expansion factor for the flow; F_a (dimensionless) is the area thermal expansion factor; ρ (lbm/ft^3) is the flow mass density; and ΔP (inches H_2O) is the differential pressure across the metering device. Each of these quantities are evaluated, according to the guidelines set forth in Reference [5], for the specific type of flow measuring device used.

Using a square edge orifice for measurement of the primary mass flow rate, the quantities in equation (a) are defined as follows:

1. The flow coefficient K is 0.62 based on a β of 0.502 and a constant coefficient of discharge over the range of flows considered of 0.60.
2. The orifice area is 37.4145 in^2 .
3. Corresponding to the range of pressure ratios encountered across the orifice, the expansion factor γ is 0.98.
4. Since the temperature of the metered air is nearly ambient temperature, thermal expansion factor is essentially 1.0.
5. The primary air mass density ρ_{or} is calculated using the perfect gas relationship with pressure and temperature evaluated upstream of the orifice.

Substituting these values into equation (a) yields

$$W_p \text{ (lbm/sec)} = (2.8882) (\rho_{or} \Delta P_{or})^{0.5} \quad (b)$$

The secondary mass flow rate is measured using long radius flow nozzles for which case the quantities in equation (a) becomes:

1. For a flow nozzle installed in a plenum, β is approximately zero in which case the flow coefficient is approximately equal to the coefficient of discharge. For the range of secondary flows encountered, the flow coefficient becomes 0.98.
2. A is the sum of the throat areas of the flow nozzles in use.

3. Since the pressure ratios across the flow nozzles are very close to unity, the expansion coefficient γ is 1.0.
4. Since the temperature of the metered air is nearly ambient temperature, the thermal expansion factor is essentially 1.0.
5. The secondary air mass density ρ_s is evaluated using the perfect gas relationship at ambient conditions. Substituting these values into equation (a) yields the equation for the secondary mass flow rate measured using long radius flow nozzles.

$$W_s \text{ (lbm/sec)} = (0.13451) A (\rho_s \Delta P_s)^{0.5} \quad (c)$$

APPENDIX: C

DATA ACQUISITION AUTOMATION

The method used in taking data with this and previous theses has been completely manual. Although the procedure has been adequate, it has been cumbersome. An ancillary objective of this thesis has been to assemble a system to automatically collect, reduce and store the data required.

The system developed utilizes the following equipment:

- 1) Hewlett Packard Model 85 Computer
- 2) Hewlett Packard Model 82901M Flexible Disc Drive
- 3) Hewlett Packard Model 7225-A Plotter
- 4) Hewlett Packard Model 3497A Data Acquisition/Control Unit.

These equipments were connected via a Hewlett Packard Model 82937A Interface Bus. Additionally, the HP-85 had installed the Mass Storages, Plotter/Printer, and Matrix ROMS (Read Only Memories). Figures C-I and C-II show a system schematic and the actual equipment used.

Pressure reading were taken with poly-flo tubing from the scanivalve via a manifold arrangement to Validyne Model DP 103 pressure transducers. Two transducers are utilized, one with a pressure range of 0 to 5.5 inches of H_2O (0-0.2 psi), the other with a pressure range of 0 to 35 inches of

H_2O (0-1.2 psi). The DC electrical signal from the transducer is amplified through a Validyne CD10 Carrier Demodulator. The carrier demodulator also provides controls for setting the zero point of the transducer and varying the span of the voltage output from the unit. The DC output from the demodulator is measured by the voltmeter within the HP-3497A data acquisition unit. The 3497A transmits the voltage value to the HP-85 where a subroutine of the acquisition program converts the voltage to a pressure value.

Temperature readings are taken in a similar manner. The thermocouple voltages are read directly by the 3497A data acquisition unit. A different subroutine within the data acquisition program converts the voltage reading to temperature values.

The conversion formula for temperature was derived from reference tables for type T copper-constantan thermocouples. Voltages from the reference table at ten degree increments from 0 to 120 degrees Fahrenheit were input into a program from the HP-85 Math Pac. This program provided the coefficients for a degree Chebyshev polynomial. The following formula results:

$$T = .0608773(V)^3 - 1.289098(V)^2 + 46.58608(V) + 32.00443$$

where T is degrees Fahrenheit and V is millivolts.

Unlike the thermocouple, calibration of the transducers must be accomplished everytime the equipment is energized. This is because the location of the zero point is not always

constant. Calibration of the transducers required use of the building air supply. The compressor in the building is a Worthington Air Compressor set at a nominal operating pressure of 100 psig. Sufficient air pressure was obtained by regulating the supply valve from the building air system and installing a throttling valve prior to a manifold. Both the pressure transducers and a 20 inch water manometer are connected to the manifold. In addition an exhaust valve that can be throttled is also attached to the manifold. With the building air supply completely disconnected and all manifold valves open, the zero point is set. The air supply is then connected to the manifold and the pressure is raised to 5.0 or 20 inches of water depending on the transducer to be calibrated. A vacuum can be obtained by inserting a tee fitting into the poly-flo tubing ahead of the inlet throttling valve. This forms an eductor that takes a suction on the manifold. By varying the building air supply valve and the throttle valves connected to the manifold the desired vacuum can be reached. When calibrating the 0 to 35 inch transducer it is vital that the valve isolating the lower range transducer be secured. An additional safety precaution is to have a vent valve installed between the manifold and the transducer. This safety valve is to be open whenever the 5.5 inch transducer is not in use. After raising the pressure to the high end calibration point the pressure is trapped in the manifold by closing all valves. A transducer reading is

then taken and recorded with the pressure reading from the manometer. The process is then repeated with several values at the low end of the pressure range. These voltages and corresponding pressure values are then inputed into the Chebyshev polynomial program and the coefficients for a 1st degree euqation are obtained. These coefficients are then inserted into the data acquisition program.

It is important to note that the transducer can be set for reading either positive or negative differential pressure. Once a transducer has been calibrated for the desired positive or negative range measuring a pressure of the opposite sign will cause the zero point of the transducer to shift. Recalibration will then be required. Because both positive and negative differential pressures were measured it was found that one transducer had to be calibrated for pressures less than atmospheric and one for pressures greater than atmospheric. A Sample calibration program is included in Table C-I.

Once the calibration process is complete the data acquisition program is run. This program will record the required pressures and temperatures and upon completion store the values under the desired filename on the flexible disc. The data reduction program is then loaded from the disc storage and executed. Upon completion of the data reduction program, plot programs are called which will provide output on the 7225-A Plotter.

The data acquisition system as presently configured is semi-automatic. Temperatures can be automatically taken in sequence but the program must stop and allow the operator to manually step the scanivalve and arrange the valving for the correct transducer. The use of a digital stepping device would further automate the process. Additionally, a transducer dedicated to only determining positive or negative differential pressure and electrically operated manifold valves would complete the process.

Even though the system is not yet completely automatic, the ability to reduce data on site provides more flexibility and ease of conducting the thesis research.

```

10 ! The name of this program      430 MAT PRINT X
is CAL35X0.
20 OPTION BASE 0
30 DISP "This program is design
ed to"
40 DISP "calibrate the 0-35 in
trans-"
50 DISP "ducer. It will give you
values"
60 DISP "to input into the CHEB
Y Pro-"
70 DISP "ram to set coefficien
ts for a"
80 DISP "calibration function. M
ake sure"
90 DISP "that the 5.5 in transd
ucer is"
100 DISP "DISCONNECTED from the
manifold."
110 DIM X(4,1)
120 DISP "With the 35 in transdu
cer dis-"
130 DISP "connected at the manif
old, set"
140 DISP "the zero point on the
carrier"
150 DISP "demodulator."
160 OUTPUT 709 ;"VR5ARAC3SA"
170 DISP "Press CONT when ready.
"
180 PAUSE
190 DISP "Reconnect transducer t
o the"
200 DISP "manifold. When air pres
sure is"
210 DISP "set at 20 in H20 Press
CONT."
220 PAUSE
230 I=20
240 FOR J=4 TO 0 STEP -1
250 X(J,0)=I
260 OUTPUT 709 ;"VR5RI3SA"
270 ENTER 709 ; X(J,1)
280 OUTPUT 709 ;"ARAC3"
290 DISR X(J,0),X(J,1)
300 I=I-5
310 DISP "Decrement IN H20 by 5
IN to ";I
320 IF J=0 THEN 340
330 PAUSE
340 NEXT J
350 DISP " P      V"
360 DISP "inH20 voltage"
370 MAT DISP X;
380 DISP "HARD COPY(1=yes 2=no)"
;
390 INPUT N
400 IF N#1 THEN 440
410 PRINT " P      V"
420 PRINT "inH20 voltage"

```

TABLE C-I. SAMPLE CALIBRATION PROGRAM

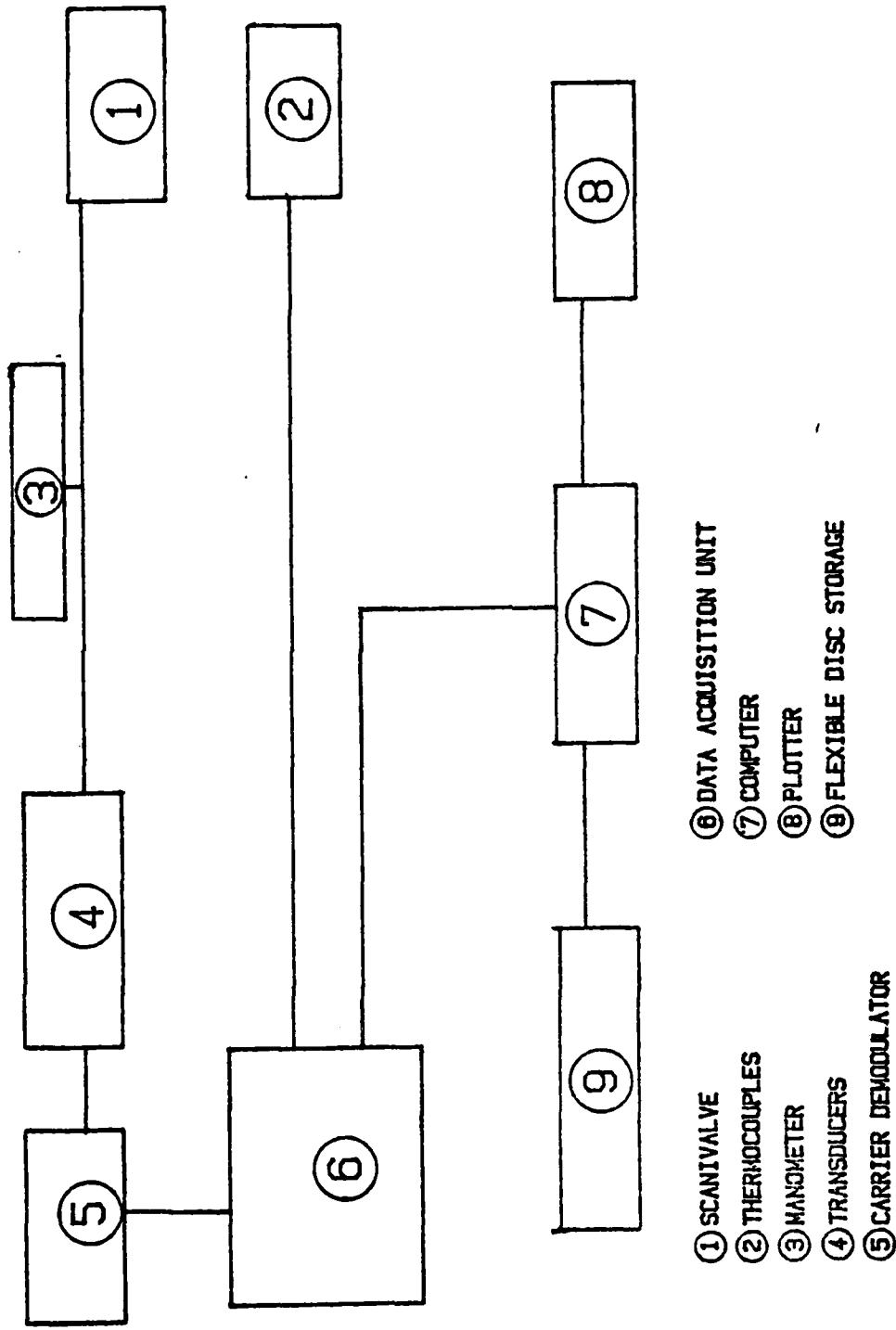


FIGURE C-1. SCHEMATIC OF DATA ACQUISITION SYSTEM

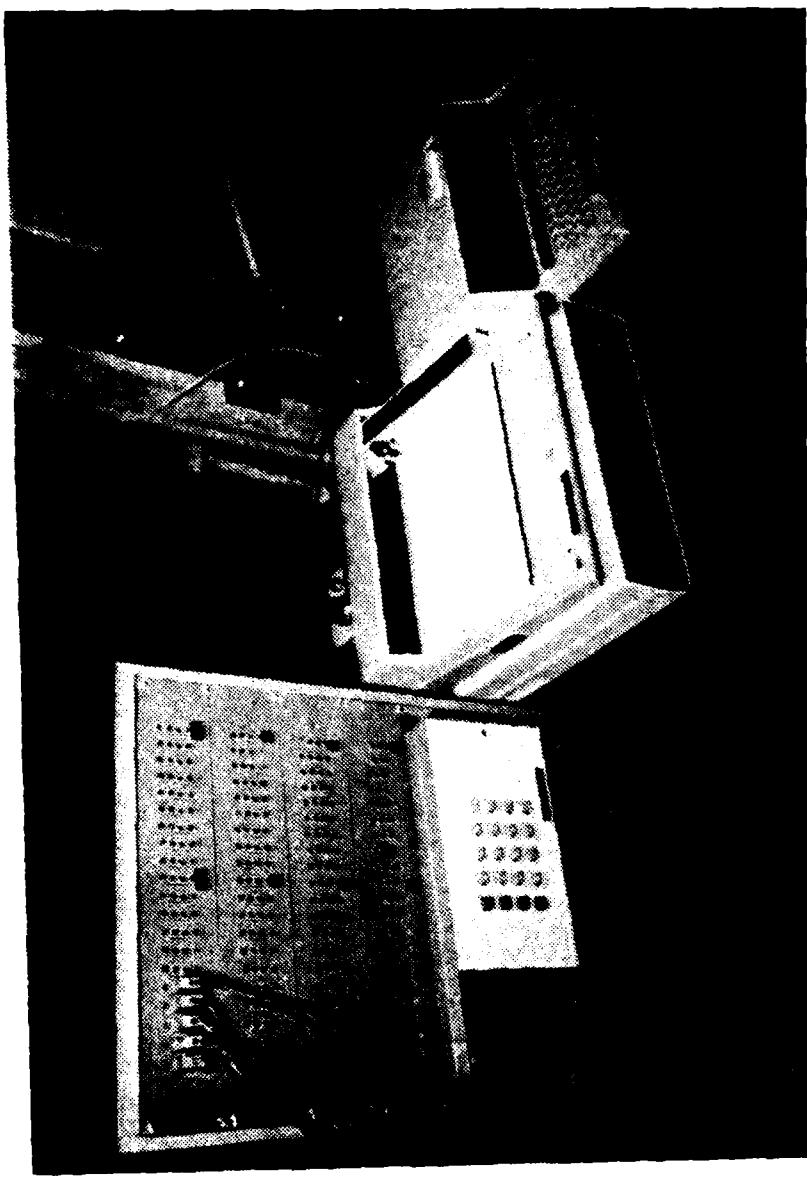


FIGURE C-IIa. ILLUSTRATION OF DATA ACQUISITION SYSTEM



FIGURE C-IIb. ILLUSTRATION OF DATA ACQUISITION SYSTEM

APPENDIX: D

UNCERTAINTY ANALYSIS

The determination of the uncertainties in the experimentally determined pressure coefficients, pumping coefficients, and velocity profiles was made using the method described by Kline and McClintock [6]. The uncertainties obtained by Ellin [1] using the second order equation suggested by Kline and McClintock [6] are all applicable to the experimental work reported herein and are summarized in the following table.

TABLE XIV

UNCERTAINTY IN MEASURED VALUES

T_s	$\pm 1 R$
T_p	$\pm 1 R$
P_a	$\pm 0.01 \text{ psia}$
ΔP	$\pm 0.01 \text{ in. H}_2\text{O}$
P_V	$\pm 0.01 \text{ in. H}_2\text{O}$
P_u	$\pm 0.05 \text{ in. H}_2\text{O}$
$\Delta P_s (+)$	$\pm 0.01 \text{ in. H}_2\text{O}$
$\Delta P_t (**)$	$\pm 0.01 \text{ in. H}_2\text{O}$
P_{or}	$\pm 0.01 \text{ in. H}_2\text{O}$
ΔP_{or}	$\pm 0.20 \text{ in. H}_2\text{O}$

T_{or}	$\pm 1 R$
T_a	$\pm 1 R$
PT (++)	± 0.1 in. H_2O

UNCERTAINTY IN CALCULATED VALUES

$\frac{AP^*}{T^*}$	1.9%
W^*T^* ^{0.44}	1.4%
V/V_{avg}	2.5%
(+)	The pressure differential across the secondary flow nozzles, P_s , is the major source of uncertainty in the pumping coefficient.
(++)	The pressure differential across the tertiary flow nozzles, P_t , is the major source of uncertainty in the pumping coefficient.
(+++)	The measurement of the total pressure for the velocity profile is the major source of uncertainty in the velocity calculation.

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